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Photovoltaic Power Stations (PVPS)

By Mohamed A. Darwish, Hassan K. Abdulrahim & Adel O. Sharif

Qatar Environment and Energy Research Institute (QEERI), Qatar

Abstract- Qatar declared that by 2020 solar energy would produce at least 2% of its total generated electric power (EP). The known solar power plants EP at utility scale level are concentrating solar power (using parabolic trough collectors, linear Fresnel collector, and solar tower), photovoltaic (PV), and integrated solar combined cycle using fossil fuel (natural gas) besides solar collectors.

EP generation by PV is reliable, clean, well proven, and matured technology, with 25 years warranties on solar panels. PV is the direct conversion of solar radiation (sunlight) into direct electric current by semiconductors that exhibit PV effect. The PV can be applied to large scale power plants called photovoltaic power station or solar parks. A solar park is connected to the grid, and thus supplies its bulk produced EP to this grid. Transfer solar energy directly to EP is achieved without using moving parts means very low maintenance and operation requirements. Once a solar park is installed (with relatively high cost compared to conventional power plant such as combined cycle), the operating costs with no fuel supply are extremely low compared to conventional power plants.

This paper presents the technology and economics of the PV power station. It outlines the main components of the PV power plants including the solar PV modules, module mounting and tracking systems, inverters (or converters), and step-up transformers. It reviews the materials of the PV cells, the PV cells degradation, and the existing PV power plant. Utility PV power plants around the world were reviewed.

PV panel are extensively used for small-distributed power generation used in homes and in remote areas. One of the advantages of building solar parks in Qatar (and other GCC) is the coincide of its power output with the high air conditioning electric power demand in hot summer days. The GCC is the Gulf Co-operation countries including Saudi Arabia, United Arab Emirates, Qatar, Oman, and Bahrain. Recent reductions in photovoltaic cells cost are the driving force behind the trend of building more solar parks worldwide.

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Photovoltaic Power Stations (PVPS)

Mohamed A. Darwish ^α, Hassan K. Abdulrahim ^σ & Adel O. Sharif ^ρ

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The System Advisor Model (SAM) software developed by National Renewable Energy Laboratory (NREL) gives the total direct capital cost of the 20 MW PV plant as \$88.0 million (M), and total installed cost as \$ 97.202 M; or \$4.86 M/MW. This is almost half the cost of the CSP using by parabolic trough plants. The LEC, as given by the computer program is \$0.16/kWh. The main disadvantage of the PV power station is non-dispatch ability.

I. INTRODUCTION

Production of Electric Power (EP) by photovoltaic (PV) is clean well proven technology, that are applied in large scale Power Plants (PPs) called PV Power Station (PVPS), or solar parks. The bulk of the generated EP by the PVSPS is supplied to the electric grid output, Fig. 1, [1].

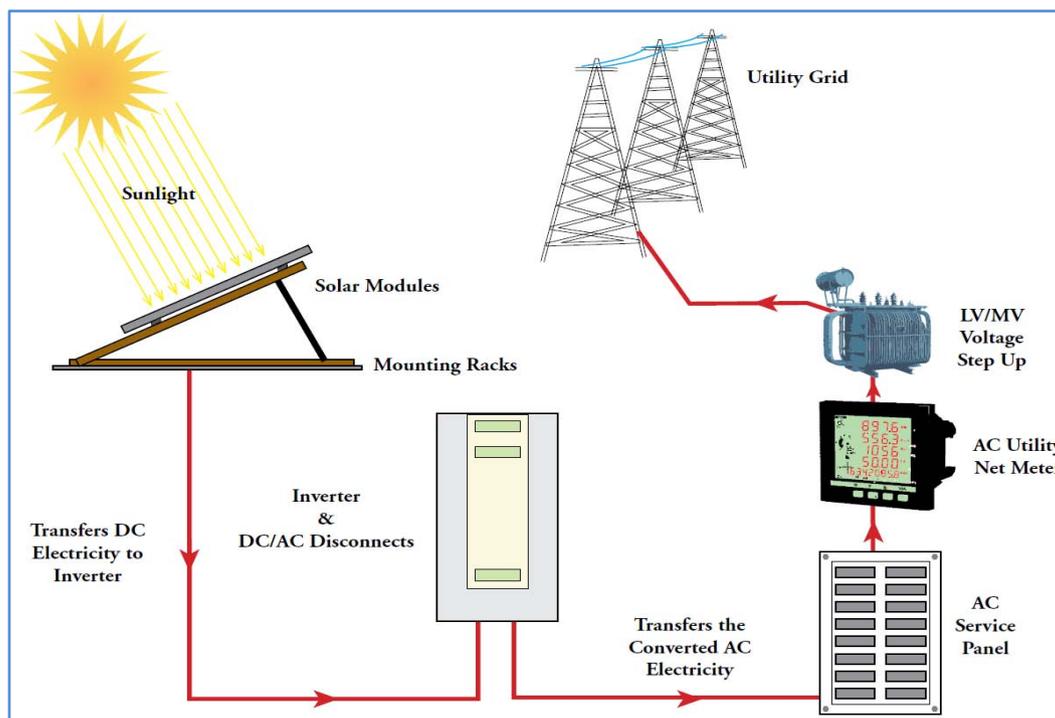


Figure 1 : Overview of Solar PV Power Plant, [1]

The economy of PVPS is improving by time as shown Fig.2a, [2]; and solar cell production is increasing, Fig.2b, [3]. The capacity of the PVPS is on the rise worldwide, Fig. 3a, [3] due to the decrease of PV cells' cost, Fig. 3b, [4]. By the end of 2013, the installed capacity of PVPS reached 136 GW, see Fig. 3a. The PVPS was rated the third in terms of capacity of the renewable energy power plants after hydro and wind in 2011, [3]. This capacity is almost doubled between 2011

to 2013 due to PV cells continuous falling costs and increasing cost of fossil fuel used in conventional power plants. It is estimated that solar module prices used in utility-scale sector (2.5 MW and above) would fall from 1.22 €/W in 2012 to 0.92 €/W in 2022, [5]. Module prices cost, are continuously decreasing as shown in Fig. 3b. A list of the countries having the highest PVPS capacity is given in Table 1, [4].

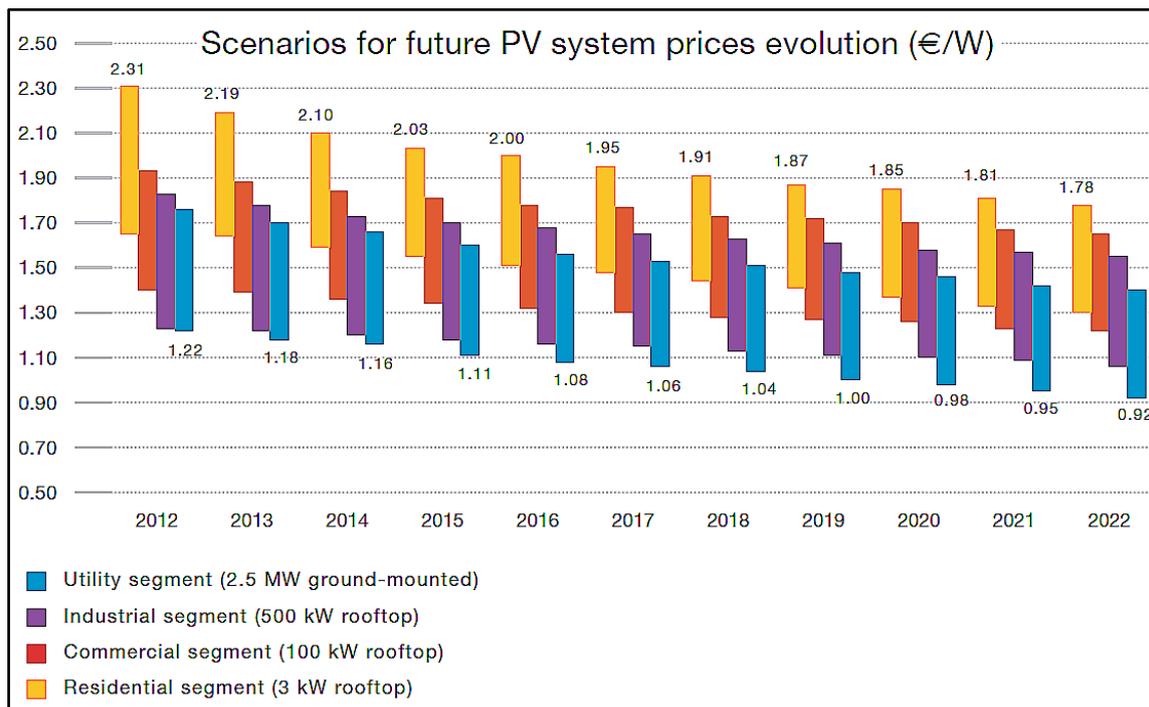


Figure 2a : Future PV Systems Evolution in Euro/W, [2]

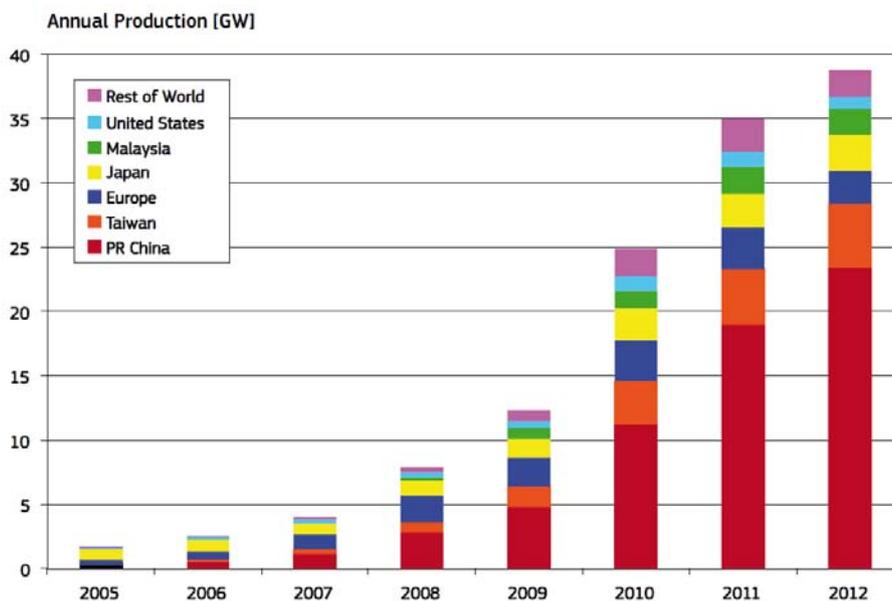


Figure 2b : World PV Cell/Module Production from 2005 to 2012 (data source: Photon International [Pho 2012], PV Activities in Japan [Pva 2013], PV News [Pvn 2013] and own analysis), [3]

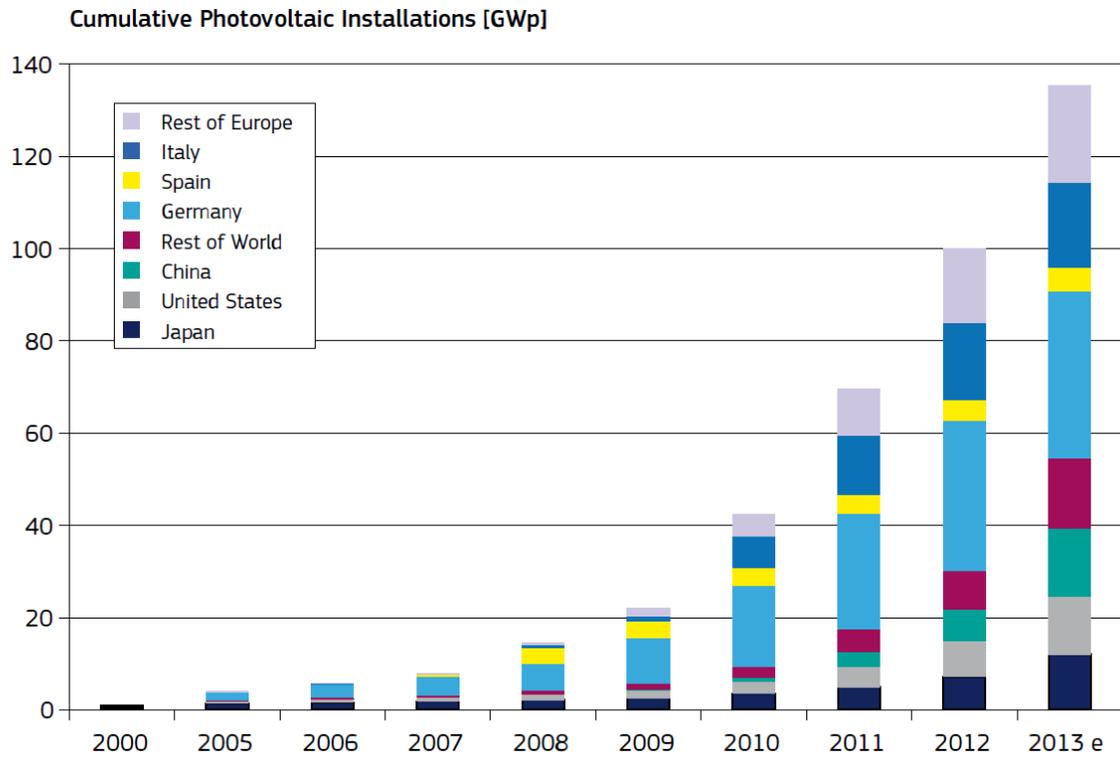


Figure 3a : Cumulative PV Cumulative Photovoltaic Installations [GWp] installations from 2000 to 2013, [3]

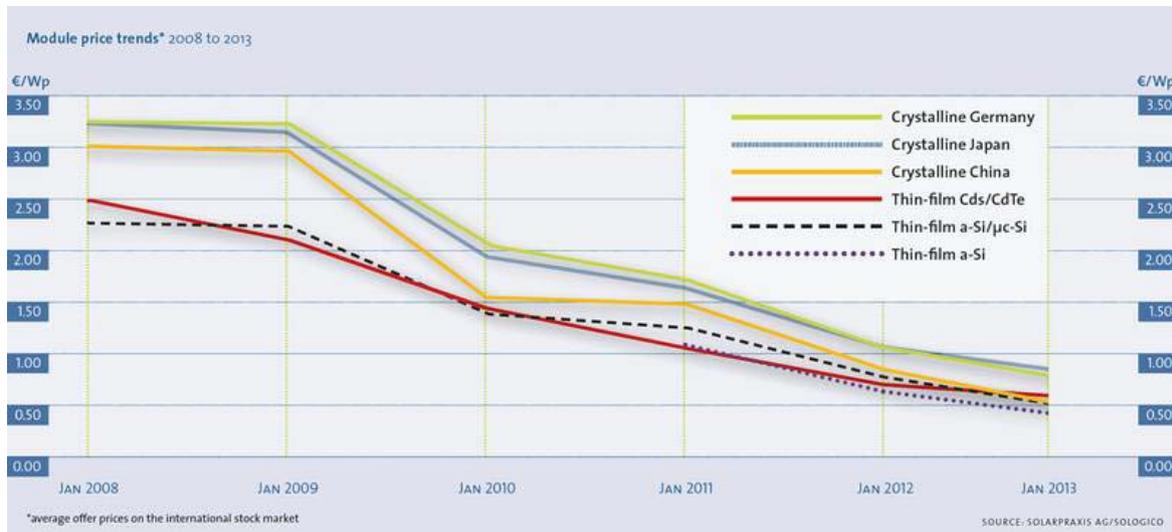


Figure 3b : Module price trends, [4]

Table 1 : Top 15 markets 2012 worldwide, [5]

	COUNTRY	2012 NEWLY CONNECTED CAPACITY (MW)	2012 CUMULATIVE INSTALLED CAPACITY (MW)
1	 Germany	7,604	32,411
2	 China	5,000	8,300
3	 Italy	3,438	16,361
4	 USA	3,346	7,777
5	 Japan	2,000	6,914
6	 France	1,079	4,003
7	 Australia	1,000	2,412
8	 India	980	1,205
9	 United Kingdom	952	1,829
10	 Greece	912	1,536
11	 Bulgaria	767	908
12	 Belgium	599	2,650
13	 Spain	276	5,166
14	 Canada	268	765
15	 Ukraine	182	373
	 Rest of the World	2,692	9,546
	Total	31,095	102,156

In Qatar, the advantages of using PVPS are clear. The primary solar energy (sunlight) is free and abundant, no moving parts and thus the needed maintenance is low, and low operating cost as no fuel is used. No water is required for operation except that needed for cleaning the panels. The decreasing cost of the PV modules lowers the capital cost and drives for installing more PVPS. The main factors hindering the spread of PVPS are still high capital cost, large needed site area, and the fact that the PVPS are not dispatchable plants. The site area of a PVPS having 15% efficiency and fixed tilt modules is about 10,000 m²/MW in tropic regions (23.5 degrees to the North and South of the Equator respectively); and up to 20,000 m²/MW in Northern Europe. One square kilometer site can be used for 50 MW. This area increased about 10% for a single axis tracker, and 20% for a 2-axis tracker to avoid shadow.

The largest cost of PVPS is still that for the modules, (accounts for about 50% of total cost), followed by costs of installation materials, labor, and the inverters. The inverters replacement cost can be significant. The PV modules warranty is generally about 20–25 years long; while, the inverters warranty is typically 10–15 years long. Improvements are rapidly achieved in many subsectors, [6].

Ratings of PVPS are usually given in terms of the solar arrays DC peak capacity in MWP, or nominal maximum AC output in MW or mega volt-amperes (MVA). Solar parks usually have medium capacity (1-20

MW), although there are large capacity operating PVPS in operation, and large plants capacity (up to one GW) are planned. The Agua Caliente solar project is now the largest operating PVPS with 290 MW in Yuma County, Arizona. Figure 5 shows a 3.5 MW plant operating in Saudi Arabia.



Figure 5 : Saudi Arabia: The ground-mounted photovoltaic plant with a peak output of 3.5 MW is located in Riyadh in the grounds of the KAPSARC (King Abdullah Petroleum Studies and Research Center), the largest oil research center in the world, Photo: Phoenix Solar AG, [4]

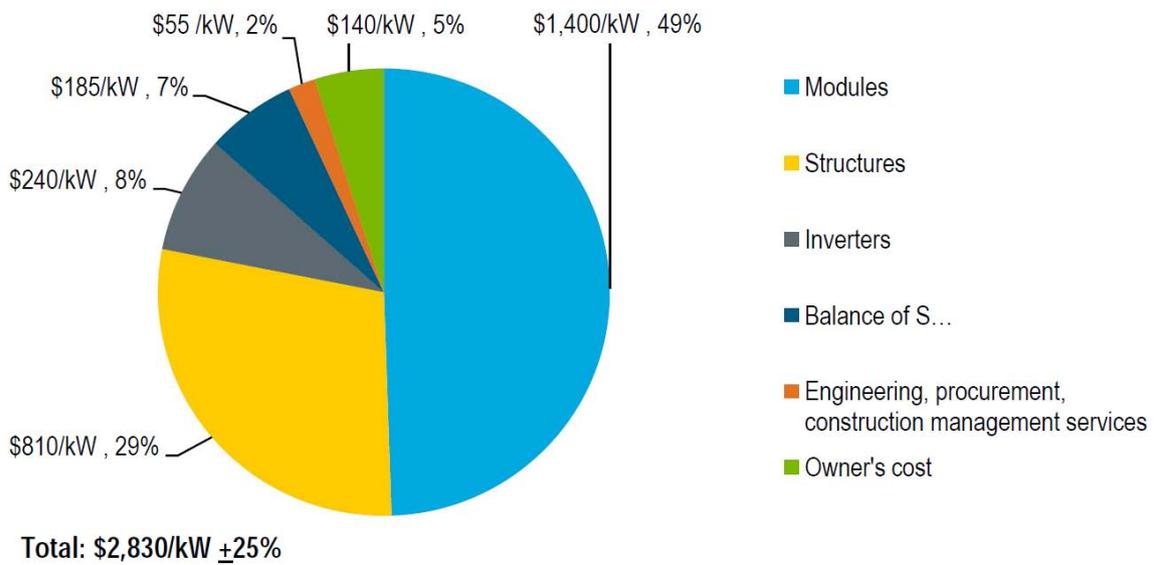


Figure 6a : Utility-scale PV facility by cost component, [7]

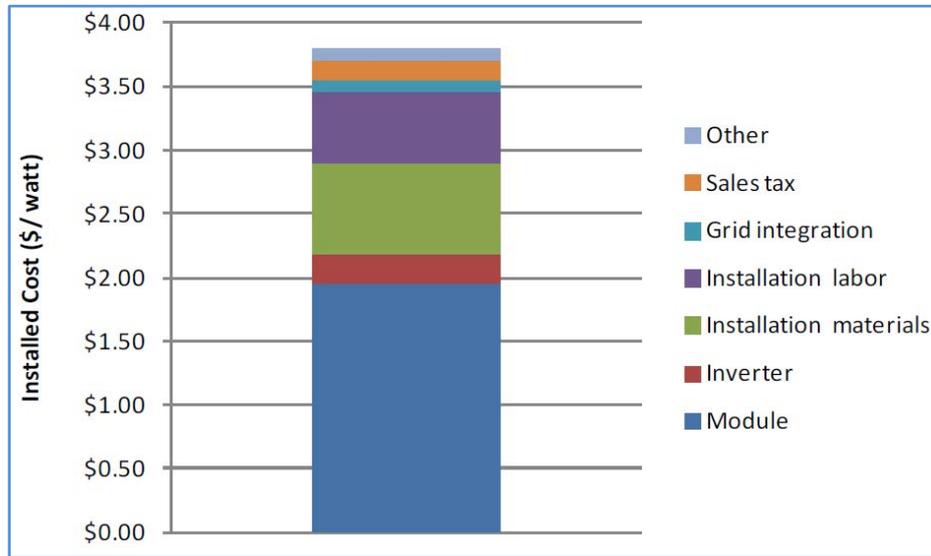


Figure 6b : Utility-scale PV facility by cost component, 2011, [7]

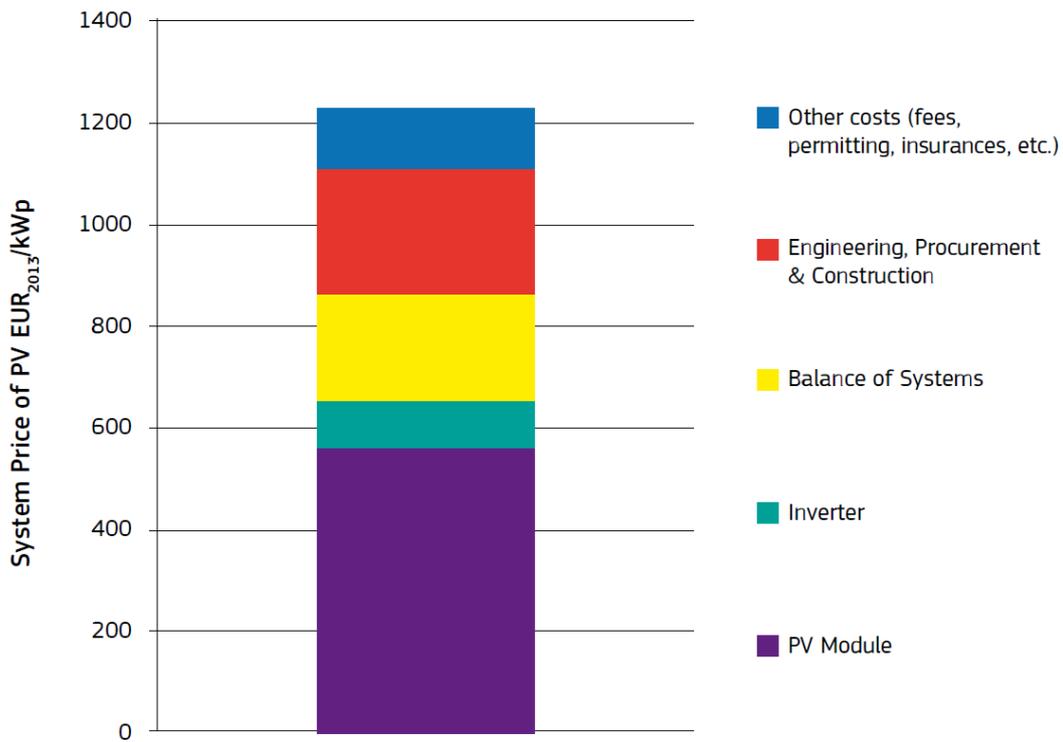


Figure 6c : Utility-scale PV facility by cost component, 2013, [3]

The cost breakdown for a fixed-tilt utility-scale PV system utilizing crystalline-silicon (c-Si) modules is shown in Figs. 6a-6c. Lower efficiency thin-film modules generally cost less but can have higher balance of plant (or non-module) expenses. This includes costs for supporting structures, DC cabling, and inverters.

The PVPS high cost and low load factor in comparison with conventional EP generation plants options are the main obstacles against the widespread

of the PVPS. Factors that can improve the competitiveness of PVPS with other EP generating systems are: (a) cost reductions of solar cell modules, (b) growing concerns about energy security and climate change, and (c) continuous increase of the fossil fuels cost. Solar panel cost per watt have been falling steadily from \$70/W in 1970 to \$4/W in 2011, (this cost does not reflect the total system cost, which will vary widely based on the application.). However, the PVPS cost is still

expensive compared to other power generation systems, [8]. The cost of the Gas Turbine Combined Cycle (GTCC) power plants that are commonly used in Qatar is low, in the range of \$1.5/W compared with \$5/W for the PVPS.

The National Renewable Energy Laboratory (NREL) in US, [9], conducted an analysis showed that the 2010 prices of PV systems in the US (cash purchase, before subsidy and considering reported target installer operating overhead and profit margins) are:

- \$5.71/WP DC – 5 kWp DC residential rooftop
- \$4.59/WP DC – 217 kWp DC commercial rooftop
- \$3.80/WP DC – 187.5 MWP DC fixed-axis utility-scale ground mount
- \$4.40/WP DC – 187.5 MWP DC one-axis utility-scale ground mount

The US showed great growth in solar power plants. Solar parks capable of delivering a total capacity of up to 750 MW are being planned or are already under construction in California, Arizona, New Mexico and Nevada.

In the hot summer in the GCC, the highest demands of EP occur in the afternoon when air conditioning machines in homes and public building are working at their highest capacity and solar power produces its maximum yields.

II. PHOTOVOLTAIC (PV) POWER PLANT SYSTEM COMPONENT

The structure of a PV cell, as shown in Fig. 7a, has two semiconductor materials, the n-type that has extra electrons in a conduction band, and the p-type that has extra holes in a valence band. When photons of greater energy than the semiconductor band gap energy, E_g , see Fig. 7b, are absorbed by the cell, the photons excite the electrons of the composite material into a higher state of energy. This allows the electrons separation from their atoms, drive electrons from the valence band to the conduction band. The movement of electrons is allowed in single direction by the nature of solar cell composition. Due to the electrons separation, positive charges are created (called holes) that flow in direction opposite of the released electrons, and this creates holes-electron pairs flowing in opposite directions across the junction, and act as charge carriers for a direct electric current. This process is called photovoltaic (PV) effect. The generated electron/hole pairs by the energy of the incident photons overcoming the energy band gap of the PV material to make a current flow according to the built-in potential slope, typically with a p-n junction of semiconductor, in the material. The freed electrons carried away by metal electrodes, and power is produced by connecting the electrodes to an external load. So, the operation of solar

cells is based on the binding energy of electrons of a crystal. Two bands, called conduction and valence, can be totally or partially occupied by electrons, Fig. 7b. Therefore, the PV cells consist of layered of semiconductors in contact with metal electrodes and covered by a protective transparent glazing. The semiconductor material used in cells is predominantly silicon because the band gap energy of silicon results in theoretical efficiency very near to the maximum for solar radiation. The maximum efficiency of a PV cell can be increased further if multiple semiconductor layers, or junctions, are stacked. In this case, the band gap of each layer is optimized for a different range of photon energies, thereby taking advantage of a greater range of the solar spectrum and improving the overall cell efficiency. A solar module consists of assembled and connected solar cells, and an array consists of assembled and connected solar modules. The array converts solar energy into a usable amount of direct current (DC) electricity.

a) Main Components

The main components in the PV power systems include:

i. Solar PV modules

As given before, a PV module is combination of PV cells that produce direct electric current (DC) from sunlight with no moving parts.

Typical cells of 3W, 0.5 volts can be connected in series to produce summation of the 0.5 volts and power. When cell are connected in parallel, the output current will be the summation of current produced by the cells, but the voltage would be that of the cell.

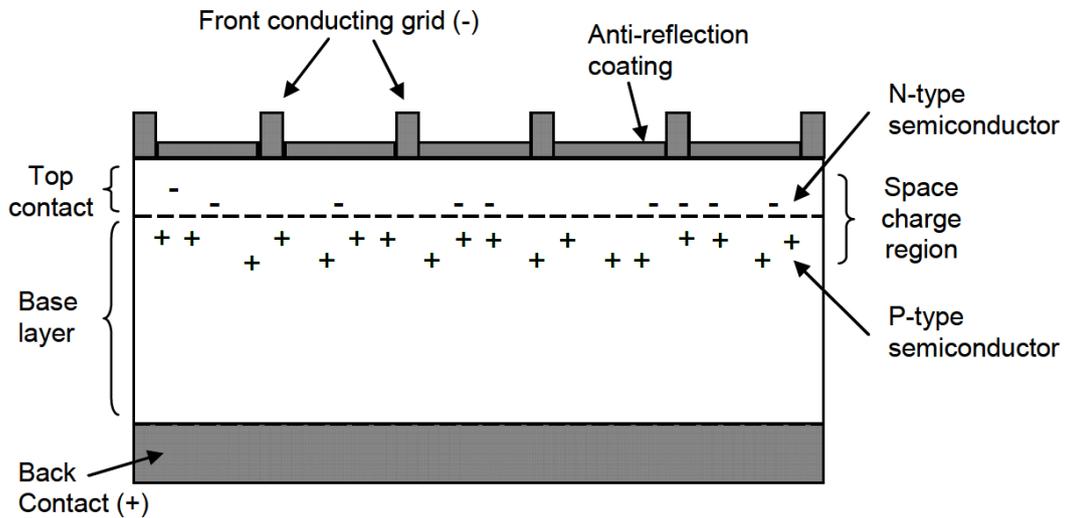


Figure 7a : PV cell structure, [10]

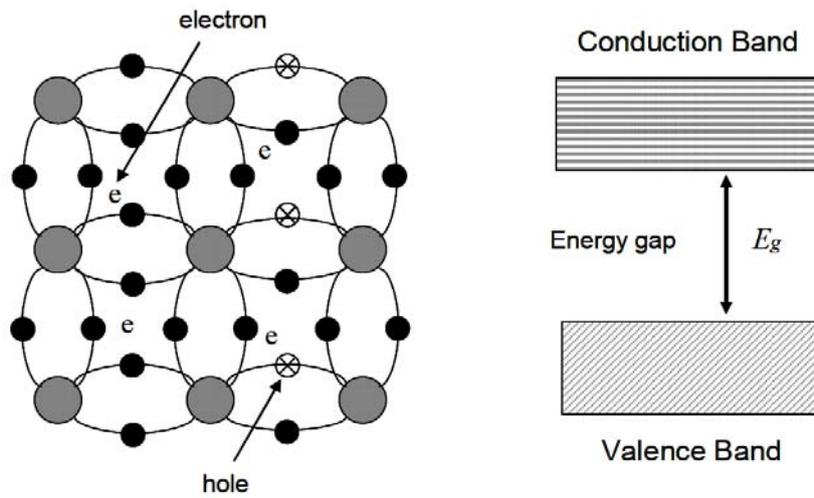


Figure 7b : The cells are electrically connected in series and in parallel to form a module, [10]

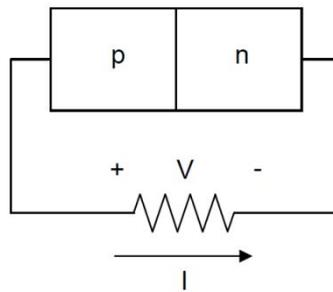


Figure 7c : Operating scheme of photovoltaic cell, [10]

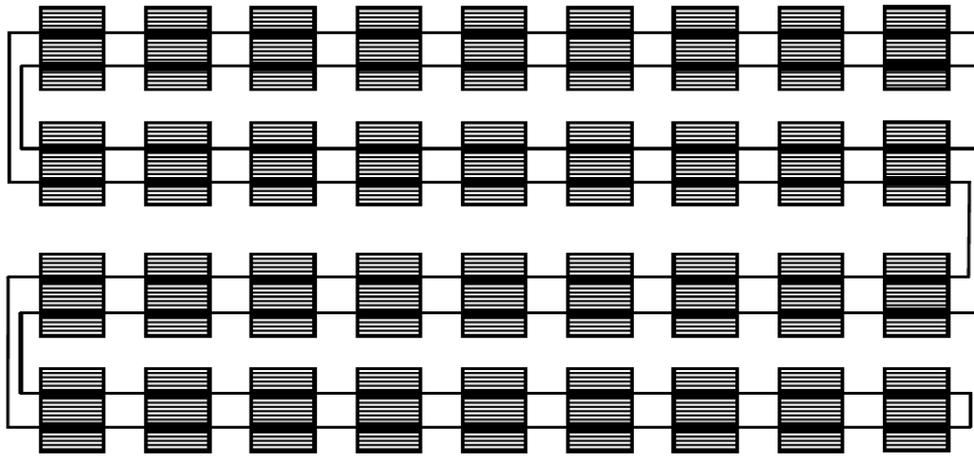


Figure 7d : Sturcture of a PV module consisting of 36 cells connected in series, [10]

When modules are connecting in series, high voltage can be obtained; and when connected in parallel, high current can be obtained.

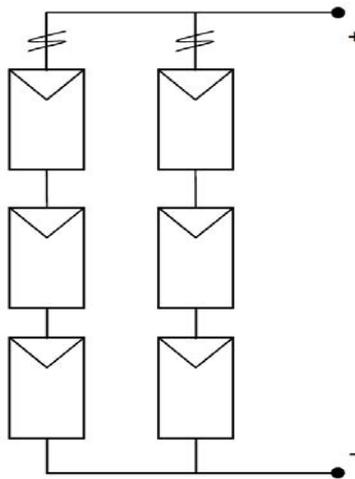


Figure 7e : Modules forming a panel connected in series-parallel with internal by pass diodes and series fuses, [10]

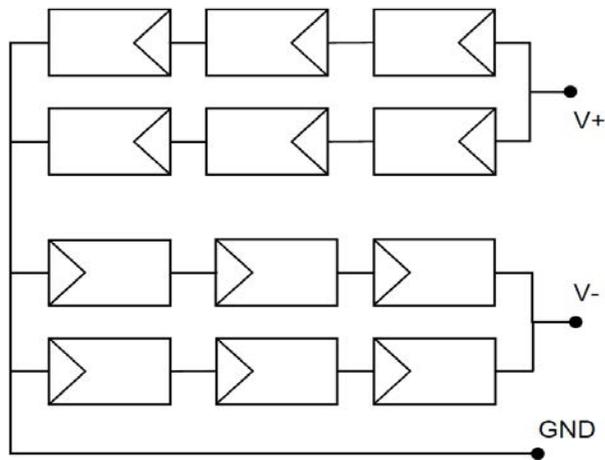


Figure 7f : Modules forming a panel connected in series-parallel with center grounded to provide + and - supplies (fuses and diodes not shown), [10]

Figure 8a shows the current (I)-voltage (V) for a module at specific irradiance. It shows the short circuit current (I_{sc}), open-circuit voltage (V_{oc}) and the maximum power point (I_{mp} ; V_{mp}), at which maximum power is attained. These three points are usually given by the PV

cell manufacturers as shown for a typical PV module (KC200GT).

The I-V curves of modules are affected by the irradiance and temperatures as shown in Fig. 8a and 8b, [11].

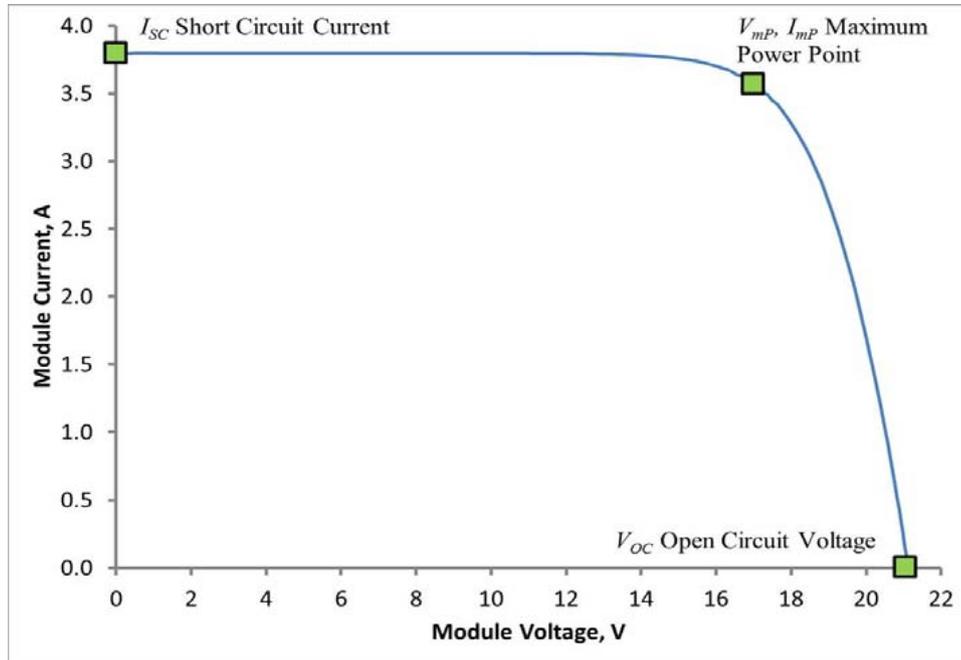


Figure 8a : Photovoltaic module I-V characteristics

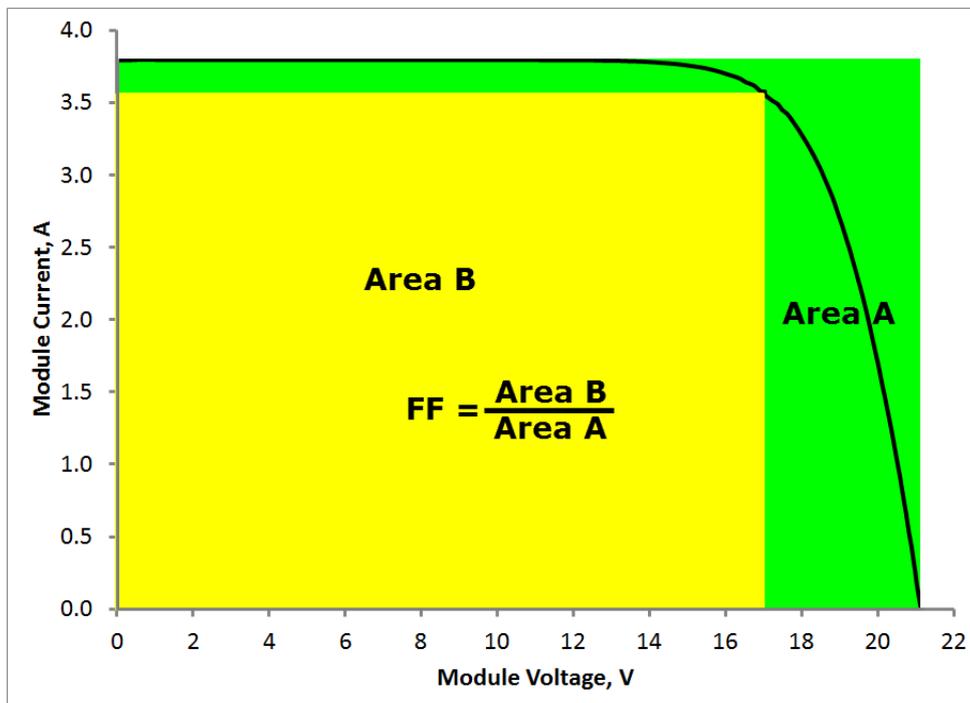


Figure 8b : Photovoltaic module characteristics showing the fill factor, [11]

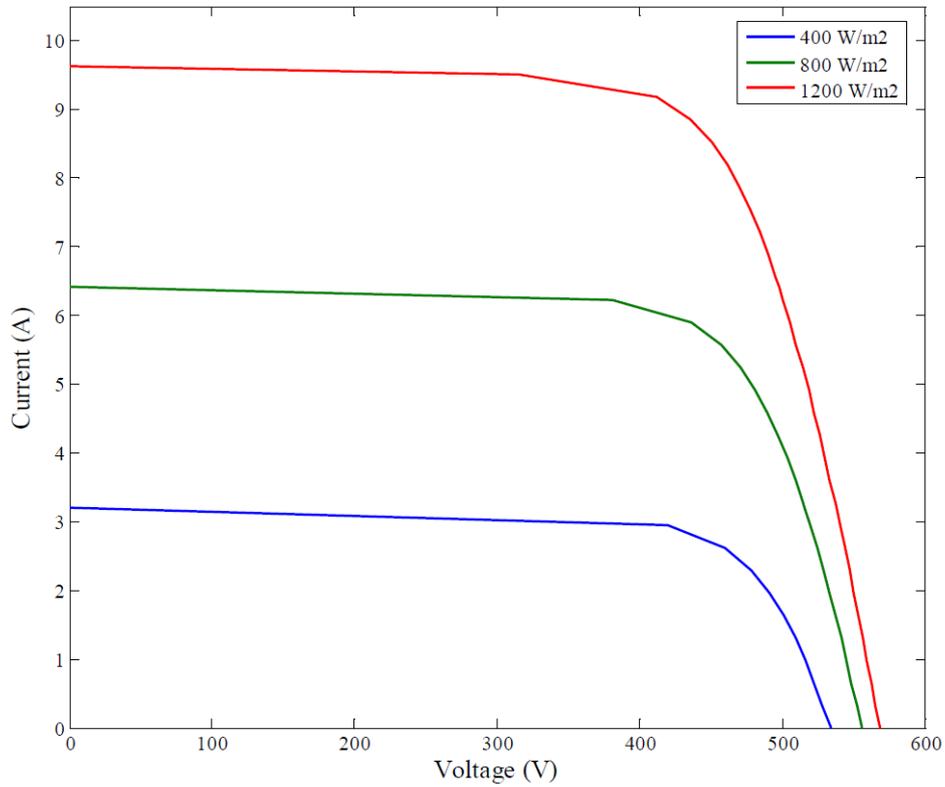


Figure 8c : The effect of irradiance on the I-V characteristics for typical module, [11]

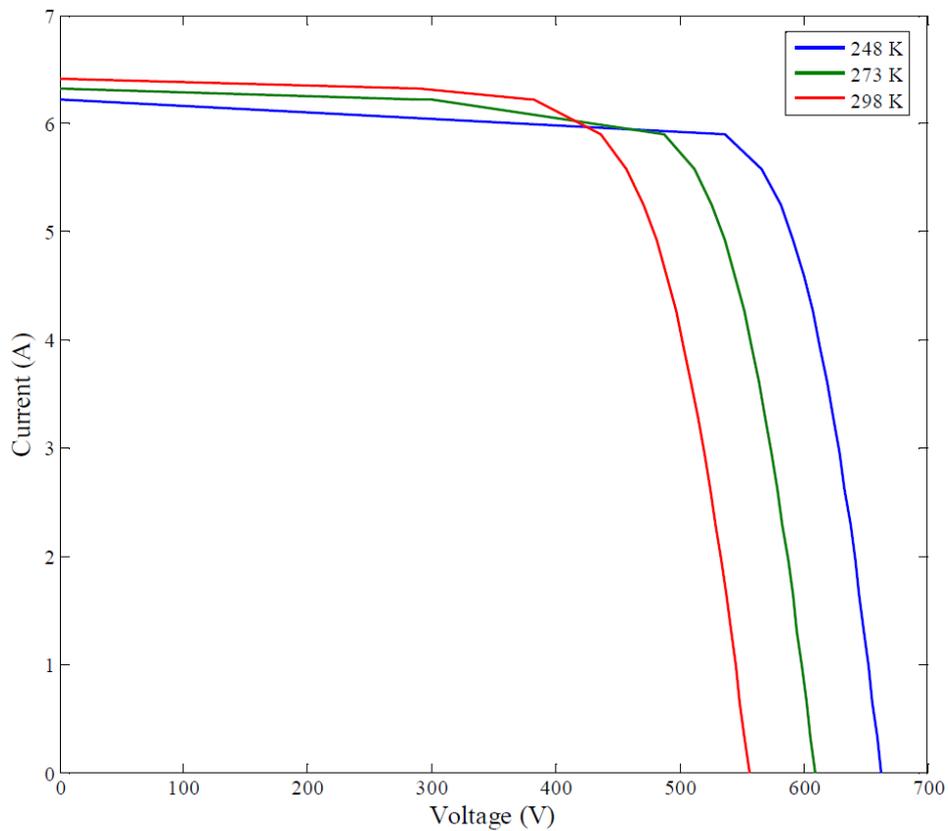


Figure 8d : The effect of temperature on the I-V characteristics for typical module at 800 W/m² irradiance, [11]

Table 2 : Datasheet Parameters for KC200GT, [12]

Irradiance	1000 W/m ²	800 W/m ²
Maximum Power Point Current	7.61A	6.13A
Maximum Power Point Voltage	26.3V	23.2 V
Maximum Power Point	200.14 W	142W
Short Circuit Current	8.21 A	6.62 A
Open Circuit Voltage	32.9V	29.9 V

The maximum power point (optimum operating point) shown in Figs 8a and 8b of a PV module is function of cell temperature and in solution level and array voltage, as shown in Fig. 8d for a PV called

KC200GT module are given before. A maximum power point tracker (MPPT) is needed to operate the PV array at this optimal point.

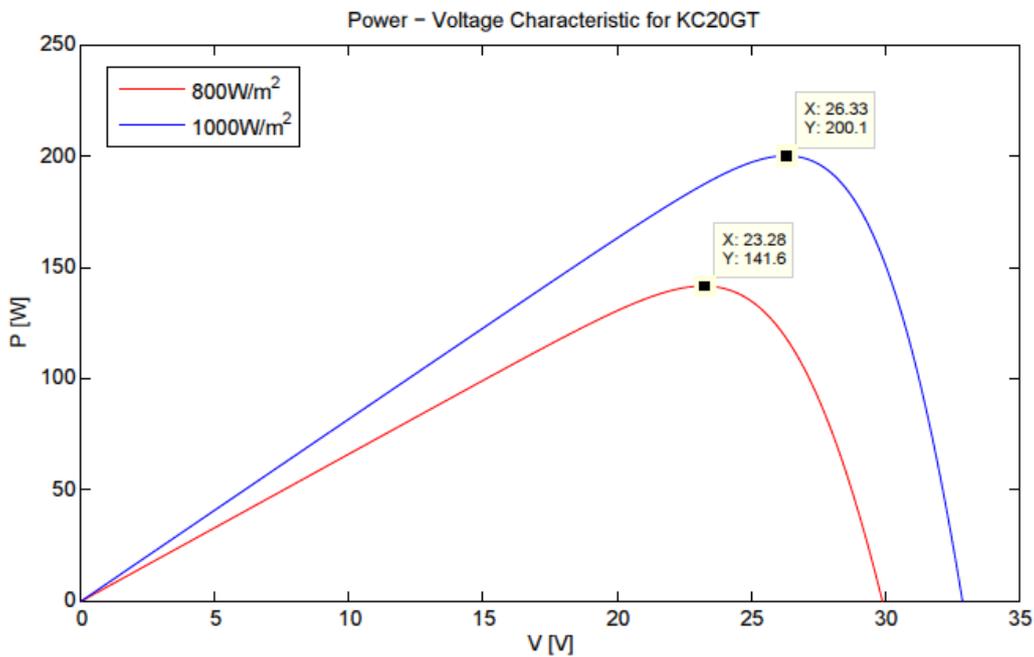


Figure 8e : Power-voltage characteristics of KC200GT, [12]

ii. *Inverters (or converters)*

Inverters convert the generated DC to alternative current (AC) in order to be connected to the utility grid. The modules are connected to the inverters through series strings and parallel strings. The PV systems connected to the grid normally do not have any real influence on the grid voltage. Their voltage operation range are therefore more of a protection function that is used for detecting abnormal utility, rather than regulators

iv. *Module mounting (or tracking) systems*

The modules should be attached to the ground. They can face the sun at fixed tilt angle, or they can be fixed to frames that track the sun.

iii. *Step-up transformers*

Further step-up of the inverters voltage output to that required by the AC grid voltage (e.g. 25kV, 33kV, 38kV, 110kV depending on the grid connection point) is conducted by further step-up transformers; see Figure 9, [1].

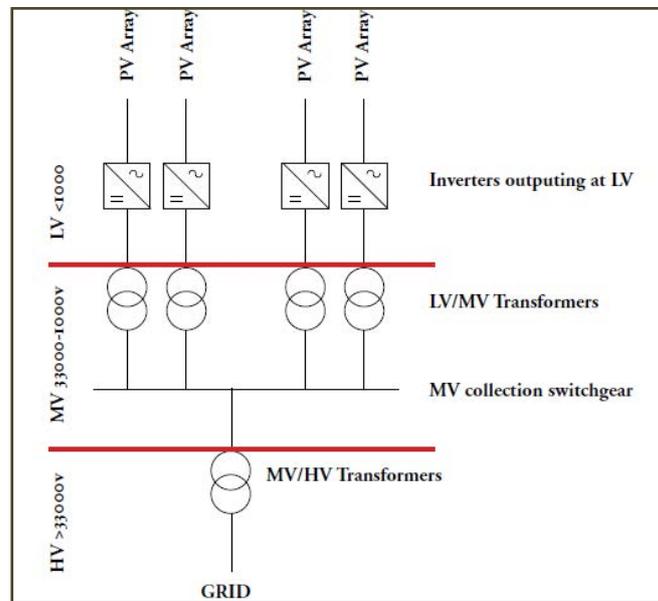


Figure 9 : Typical Transformer Locations and Voltage Levels in a Solar Plant where Export to Grid is at HV, [1]

The substation and metering points are usually located outside the PVPS and typically located on the network operator's property. Connections to the grid network are of major concern when building PVPS in terms of the availability, locality, and capacity. This network should be able to absorb the maximum capacity of the PVSP. The PVPS may be sited at a distance (few kilo-meters) of a suitable grid connection point.

b) Photovoltaic Cell Materials

Most PV cells are manufactured from silicon (Si) that doped with negatively and positively charged semiconductors of phosphorous and boron. When sunlight is received by the PV cell, electrons become free to flow from the negative phosphorus to the positive

boron. The produced DC is obtained through a metal grid covering the cell and external circuit. Besides crystalline silicon (c-Si), and amorphous silicon (a-Si) thin-film technologies, only cadmium telluride (CdTe) has had significant success in utility-scale solar development.

Silicon (Si) material can be mono-crystalline, poly-crystalline and amorphous silicon. Ribbon cast polycrystalline cells are also produced by drawing, through ribbons, flat thin films from molten silicon to reduce the silicon waste by sawing from ingots and thus reduces its cost. Other than silicon materials, gallium arsenide (GaAs), cadmium telluride (CdTe), copper indium diselenide (CIS) and copper indium gallium selenide (CIGS) are used in PV cells manufacturing.

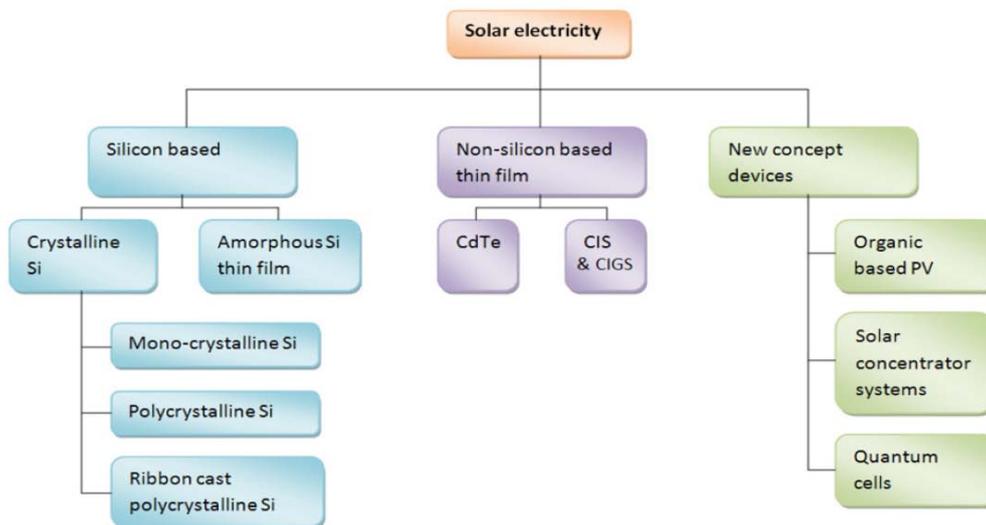


Figure 10a : PV cells material Technology [13]

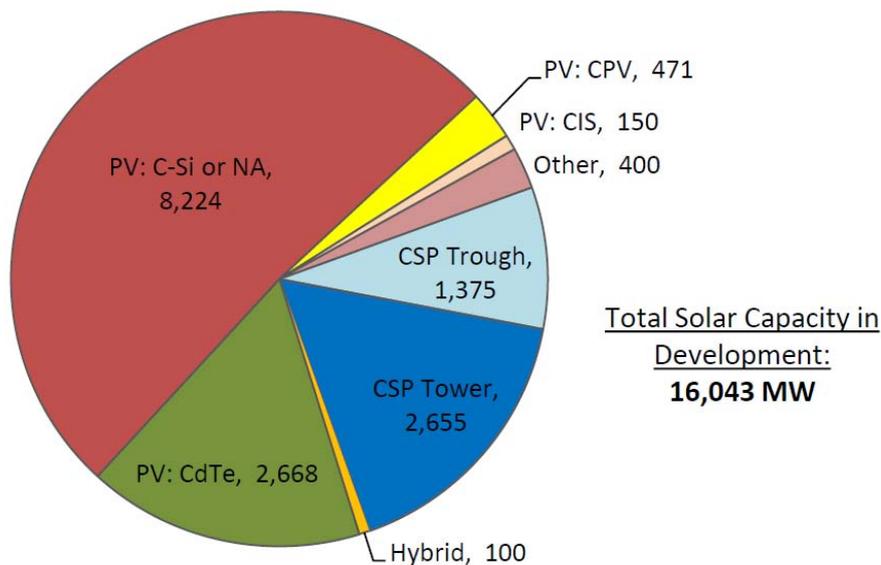


Fig. 10b : Total U.S. utility-scale solar capacity under development (all numbers in MW)

Among the utility scale PV plants in the US, about 24.5% use CdTe, and 74.5% use c-Si, see Fig. 10b. An overview of the different main PV cells materials is given in Fig. 10. The mono-crystalline cells are made of pure silicon, have grey or black color, more efficient (16–24%) than the polycrystalline silicon (14–18%), see Table 3. Solar panel efficiency is the ratio of electric power produced by a PV module to the power of the sunlight striking the module.

The polycrystalline silicon cells are easier to be manufactured (to be sawed from ingots) and thus cheaper but less efficient than the mono-crystalline cells, and have shiny blue color. Amorphous silicon (so called thin-film) cells consist of non-crystallized very thin layers deposited onto a substrate, has brown or red-brown color, reddish brown, and typical efficiency of 4% to 10%, see Table 3. The power per unit area is typically

75–155 Wp/m² for mono-crystalline and poly-crystalline modules, and 40–65 Wp/m² for thin-film modules [13].

The other thin-film cells, other than the amorphous silicon, are Cadmium telluride (CdTe) and Copper indium (gallium) di-selenide (CIGS). The CdTe solar cells are manufactured on as substrate glass with transparent conducting oxide (TCO) layer usually made from fluorinated tin oxide (FTO) as the front contact. This is initially coated with an n-type cadmium sulfide (CdS) window layer and secondary with the p-type CdTe absorber layer. The color is reflective dark green to black and typical cell efficiencies are 9.4–13.8%. The conversion efficiencies of Copper indium selenide (CIS), and Copper indium (gallium) di-selenide (CIGS) cells are shown in Fig. 10b. Values for the highest reported efficiencies of CdTe and CIGS solar cells are shown in Fig. 10b.

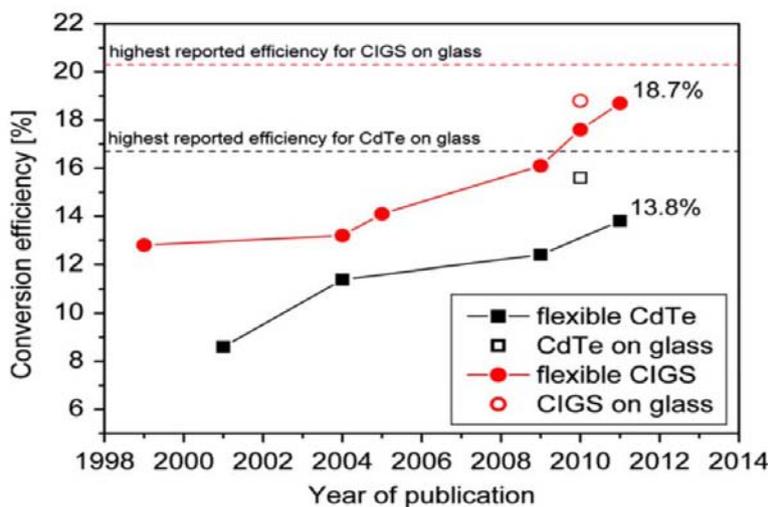


Figure 10c : Conversion efficiencies of flexible CdTe and CIGS solar cells fabricated by low temperature processes. Also shown is the in-house reference on glass and the highest reported efficiency for each technology, [13]

The silicon-based crystalline (c-Si) wafers usually give high solar cells efficiency but at high manufacturing cost. The thin film cells are cheaper but less efficient, [1].

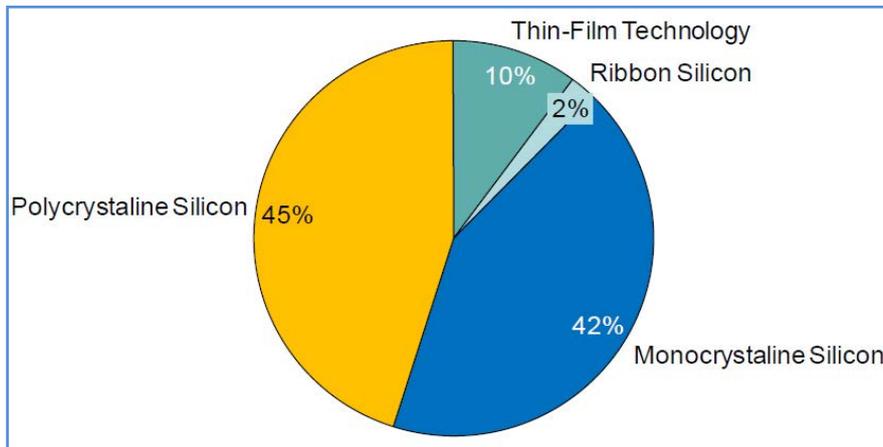


Figure 10d : Market share with regard to technology [in percent], [14]

The characteristics of the cell material affect the cell performance, cost, and methods of manufacture, [3].

In 2010, 78% of the cells used PVPS were wafer-based crystalline silicon modules; and the

percentage of amorphous silicon and cadmium telluride thin film modules was 22%. The solar cell materials are classified in Figure 7, [1], and their main characteristics are given in Table 3, [1].

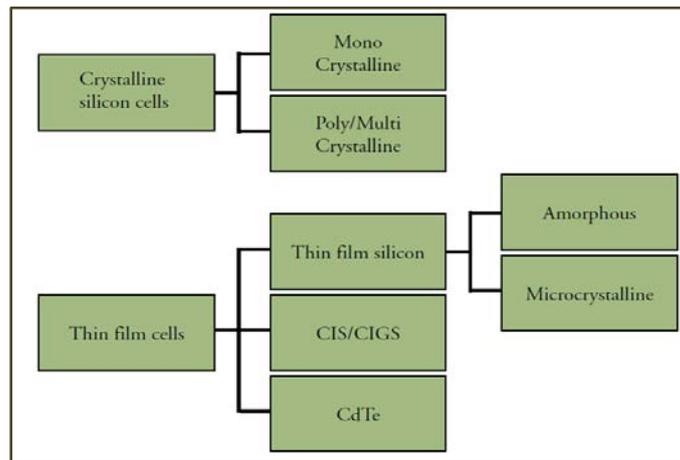


Figure 10b : PV Technology material classes, [1]

Table 3 : Characteristics of various PV technologies, [1]

Technology	Crystalline Silicon	Amorphous Silicon	Cadmium Telluride	Copper Indium Gallium Di-Selenide
Abbreviation	c-Si	a-Si	CdTe	CIGS or CIS
Cost (\$/Wp, 2009)	3.1-3.6	2.5-2.8	2.1-2.8	2.7-2.9
Percentage of Global installed capacity	78%	22%		
Thickness of cell	Thick layers (200-300 μm)	Thin layers (<1 μm)	Thin layers (<1 μm)	Thin layers (<1 μm)
Current commercial efficiency	12-19%	5-7%	8-11%	8-11%
Temperature coefficient for power (typical)	-0.5%/°C	-0.21%/°C	-0.25%/°C	-0.36%/°C

Table 3 shows that the cell efficiencies are in the range of 5-7% for amorphous, and 12-19% for the thick

layers c-Si. The efficiency can reach up to 44.0% with multiple-junction concentrated photovoltaic, [3].

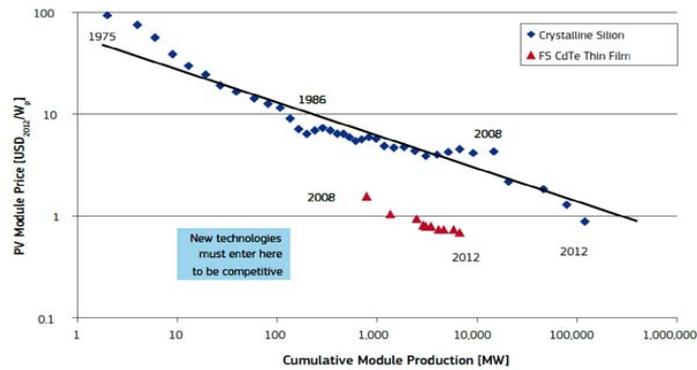


Fig. 10e : Price-experience curve for solar modules (data source: Bloomberg New Energy Finance and PV News), [3]

c) PV Degradation

The performance of PV modules is degraded over time. High degradation occurs in the first year upon initial exposure to light and then it stabilizes. Degradation is mainly affected by used module characteristics. Irreversible light-induced degradation is suffered by c-Si modules due to the presence of boron, oxygen or other chemicals left after cells production. The so called Staebler-Wronski Effect, [15], degrades the amorphous silicon cells, and can cause 10-30% power output reductions in the first six months of exposure to light before stabilization with much less degradation rates. The performance of amorphous silicon cells after stabilization is usually given by the manufacturers. The performance of amorphous silicon is affected by temperature. The modules perform better in hot summer, and drop in cold winter.

Degradation can be caused also by environment effects such as air pollution, dis-coloring or haze of the lamination defects, humidity, and wiring degradation. Degradation can be reduced by regular maintenance and cleaning.

In general, long term of power output degradation rate ranges between 0.3 and 1% per year. Banks often assume a flat rate of degradation rate of 0.5% per annum, [15]. In general, good quality PV modules may be expected to have a useful life of 25-30 years.

III. PV SYSTEM PERFORMANCE

a) PV Cell and Module Ratings

The solar modules are compared with each other based on standard test conditions at normal irradiance rate of 1000 W/m², cell temperature 25°C and Air Mass (AM)=1.5. The AM is corresponding of receiving surface at 37° tilt angle towards the equator facing the sun.

Solar insolation is the integration of irradiance over a specified time, usually day, year or an hour.

Therefore, the insolation has a unit of Watt-hours per square meter. The insolation is usually denoted by H is used for insolation for one day; I is used for insolation for an hour or year. The symbols H and I can represent beam, diffuse or global and can be on surfaces of any orientation. Solar radiation consists of beam (direct) radiation received from the sun without having been scattered by the atmosphere, and diffuse radiation received from the sun after its direction has been changed by scattering in the atmosphere. The sum of the beam and the diffuse solar radiation on a surface, global radiation, is often referred to as total solar radiation. The most common measurements of solar radiation are global radiation on a horizontal surface, referred to as global horizontal radiation.

Peak sun hour is the total number of hours of a day that can receive radiation; it is an equivalent form of insolation and most radiation data is represented using either of these units expressed as kWh/m²/day. The figure below shows the annual insolation map of the United States.

The performance ratio (PR) of the PVPS is defined as percentage ratio of the AC yield to the installed capacity in kWp multiplied by plane array irradiation in kWh/m², [1]. It gives the yield to the maximum nominal output. The PR does not take in consideration the size or the solar resource. A PVPS of high PR converts solar energy to electric power efficiently, and can be achieved by well-designed solar PVPS and not operated in high temperature conditions. The PR of varies between 77% in summer to 82% in winter. Amorphous silicon modules in some PVPS show the opposite effect with high PR in hot summer and low PR in cold winter. Electrical losses decrease the PR, [10], see Table2.

Throughout the components of the system there are electrical losses, which de-rate the conversion from nameplate DC power rating to AC power rating (as explained in Table 4), [16]. Table 4 gives the losses due to the several system components.

Table 4 : De-rate the conversion from nameplate DC power rating to AC power, [16]

Derate Factors for Photovoltaic System Components		
Table 6. PVWatts Default Derate Values		
Component Derate Factors	PVWatts Default	Range
PV module nameplate DC rating	95%	0.80–1.05
Inverter and transformer	92%	0.88–0.98
Mismatch	98%	0.97–0.995
Diodes and connections	100%	0.99–0.997
DC wiring	98%	0.97–0.99
AC wiring	99%	0.98–0.993
Soiling	95%	0.30–0.995
System availability	98%	0.00–0.995
Shading	100%	0.00–1.00
Sun-tracking	100%	0.95–1.00
Age	100%	0.70–1.00
Overall DC-to-AC derate factor	77%	0.09999–0.96001

Table 4 notes that the overall DC-to-AC de-rate factor varies for different PV systems and applications. NREL’s PVWatts tool incorporates a standard de-rate factor of 0.77 (or a 23% loss in output from nameplate DC rating to actual AC energy produced).

The load (or capacity) factor of a PVPS power plant (usually expressed in percentage) is the ratio of the actual output over a period of one year and the target yield (output if it had operated at nominal power the entire year), and is defined as:

$$CF = \frac{\text{Actual yield } E}{\text{Target yield}} = \frac{\text{Annual Energy Generated}(kWh)}{8760(\text{hours/ annum}) \times \text{Installed Capacity}(kWp)}$$

Note that the target yield (dominator) is different from the annual sum of global irradiation, h , that hits the module, and it depends on the specific location. The value of h is to be obtained from measurements, or from an irradiance map, and its units is kWh/m^2 . The relation between the target a_{out} and h is given by:

$$\text{Target yield} = h A \eta_{norm}$$

This gives
$$\frac{\text{Actual yield } E}{\text{Target yield}} = \frac{E}{h A \eta_{norm}} = \eta_{pre} \eta_{rel} \eta_{sys}$$

- Where, η_{nom} = Nominal efficiency
- η_{pre} = Conversion efficiency
- η_{rel} = Relative efficiency
- η_{sys} = system efficiency

The performance ratio is independent from the irradiation h and therefore it is useful to be used to compare systems. The specific final yield, Y_f , (kWh/kWp) is the total annual energy generated E in kWh divided by the nameplate DC power P_0 of the installed modules capacity (kWp), i.e., $Y_f = E/P_0$. Another useful expression is the specific yield to the standard conditions of $1 kW/m^2$ irradiance Y_r . The reference yield Y_r is the total in-plane irradiance H divided by the PV’s reference irradiance G , i.e., $Y_r = H/G$ (hours). Therefore, Y_r is the

number of peak sun-hours or the solar radiation in units of kWh/m^2 . The performance ratio PR is the Y_f divided by the Y_r , i.e., $PR = Y_f/Y_r$ (dimensionless).

Qatar annual global horizontal irradiation GHI are given as: $2055 kWh/m^2$ (minimum), $2160 kWh/m^2$ (maximum), $105 kWh/m^2$ (range) and $2134 kWh/m^2$ (mean), [17]. The fixed tilt PVPS capacity factor plant in sunny areas is about 16%. This means that a PVPS of 100 MWp plant would generate the equivalent energy of 17.7 MW by combined cycle (CC) having 90% CF.

b) Photovoltaic Power Station

The largest solar PVPS as of March 2014 are given in Table 5.

Table 5 : Large-Scale Photovoltaic Power Plants, Ranking 1-50, [18]

Power	Location	Description	Commissioned
320 MWp	Longyangxia Dam, Qinghai Province, China,	Longyangxia Hydro-solar PV Station	2013
250 MW	san Luis Obispo, CA, USA	California Valley Solar Ranch	2012-2013
250 MW	Yuma County, AZ, USA	Agua Caliente Solar Project	2012
214 MW	Charanka, India	PV power plant	2012
200 MWp	Gonghe County, Qinghai Province, China	Charanka Park, Patan district, PV power plant	2012
200 MWp	Golmud, China	Golmud PV power plant	2011
166 MWp	Meuro, Germany	SolarparkMeuro	2011-2012
150 MW	Sonoran desert, AZ, USA	Mesquite Solar I	2011-2012
145 MWp	Neuhardenberg, Germany	SolarparkNeuhardenberg	2012
143.2 MW	Kern County, CA, USA	Catelina Solar Project	2013
139 MW	El Centro, Imperial Valley, CA, USA	Campo Verde Solar Project	2013
128 MWp	Templin, Germany	Solarpark Templin	2012
125 MW	Maricopa County, AZ, USA	Arlington Valley Solar Energy II	2013
115 MWp	Toul-Rosières, France	Centralesolaire de Toul-Rosières	2012
105.56 MWp	Perovo, Ukraine	Perovo I-V PV power plant	2012
100 MW	Chengde, Hebei Province, China	Chengde PV Project Phase I and II	2013
100MW	Jiayuguan, Gansu Province, China	Jiayuguan PV power plant	2013
100 MW	Xitianshan China	Xitianshan I,II,III PV power plant	2012
97 MW	Sarnia, Canada	Sarnia PV power plant [2]	2009-2010
92 MW	Boulder City, NV, USA	Copper Mountain II Solar Facility	2012
91 MW	Briest, Germany	SolarparkBriest	2011
84.7 MWp	Finowfurth, Germany	SolarparkFinowTower I,II	2010-2011
84.2 MWp	Montalto di Castro, Italy	Montalto di Castro PV power plant	2009-2010
84 MW	Lopburi, Thailand	Lopburi PV power plant	2011-2012
82.65 MWp	Ohotnikovo, Ukraine	Ohotnikovo PV power plant	2011
82 MWp	Senftenberg, Germany	SolarparkFinsterwalde I,II,III	2009-2010
80.245 MWp	Finsterwalde, Germany	SolarparkFinsterwalde I,II,III	2009-2010
80 MWp	Eggebek, Germany	SolarparkEggebek	2011
75 MWp	Kalkbult, Northern Cape, South Africa	Kalkbult PV facility	2013
71 MWp	Turnow-Preilack, Germany	SolarparkLieberose	2009-2011
70.556 MW	San Bellino, Italy	San Bellino PV power plant	2010
70 MW	Kagoshima pref., Japan	Kagoshima Nanatsujima Mega Solar Power Plant	2013
70 MW	Wittstock, Germany	Solarpark Alt Daber	2011
69.7 MWp	Crimea, Ukraine	Nikolayevka Solar Park	2013
68 MWp	Sault Ste.Marie, Canada	Starwood SSM I,II,III	2010-2011
67.2 MWp	Losse, France	ParcSolaireGabardan	2009-2011

66 MW	Los Angeles, USA	Alpine Generating Station	2013
60.4 MWp	Karadzhalovo, Bulgaria	Karadzhalovo Solar Park	2012
60 MWp	Crucey, France	Centralesolaire de Crucey	2012
60 MWp	Olmedilla de Alarcon, Spain	Parque solar Olmedilla de Alarcon	2008
58 MW	Boulder City, NV, USA	Copper Mountain I Solar Facility	2010
56 MW	Massangis, France	ParcSolaireMassangis	2012
55 MW	Rajasthan, India	PV power plant in Rajasthan	2013
54.8 MWp	Priozernaya, Ukraine	Priozernaya Solar Park	2013
54 MWp	Straßkirchen, Germany	SolarparkStraßkirchen	2009
52.284 MWp	Walddrenah , Germany	SolarparkWalddrenah	2012
52 MWp	Brandis , Germany	SolarparkWaldpolenz	2007-2008
52 MWp	Tutow, Germany	SolarparkTutow I,II,III	2009-2011
50 MWp	Weidi, China	Weidi Solar Park	2012
50 MW	Alpaugh, CA, USA	SPS Alpaugh solar project	2012

Notes: Power is specified in MWp if DC array power is known. If DC array power is unknown then output power is specified. In some cases, it is unclear if the power is the output or DC array power. Sarnia power plant has AC power of 80 MW. This power was also disclosed in press release. DC array peak power (97 MWp) is unofficial information and is based on personal communication. SolarparkSenftenberg I (18 MWp) was put into service in 2010 and constructed by Phoenix Solar and is a separated project not related to Senftenberg II and III. Last modified: 3/15/2014.

The PVPS can be divided based on its capacity, to mid-capacity station of less than 50 MW, and large capacity plants of 50 MW or more. A NREL report issued in 2012 accounted for 56 PVPS of mid-size ranging from 5- 48 MW each, and total capacity 589.5 MW. There are another 57 PVPS in advance development under development of 20-50 totaling 1,329.5 MW. Concerning large capacity PVPS (greater than 50 MW) in US, there are currently 40 plants of 9,425 MW total capacities in the development stages.

V. POWER CONVERSION

Inverters are required to convert the DC power produced by the modules into AC, which can then be connected to the electrical grid. DC rating to actual AC energy produced. Inverters are solid-state electronic

devices. Inverters can also perform a variety of functions to maximize the output of the plant. These range from optimizing the voltage across the strings and monitoring string performance to logging data, and providing protection and isolation in case of irregularities in the grid or with the PV modules.

Technological improvements are rapidly occurring in many subsectors. For example, micro-inverters can be paired with each PV module, in contrast to centralized inverters, which are paired with a bank of modules. Therefore, if a single micro-inverter fails, only the module paired to the failed inverter is affected, [6]

There are two primary alternatives for configuring this conversion equipment; centralized inverter and string inverter, see Figure 11.

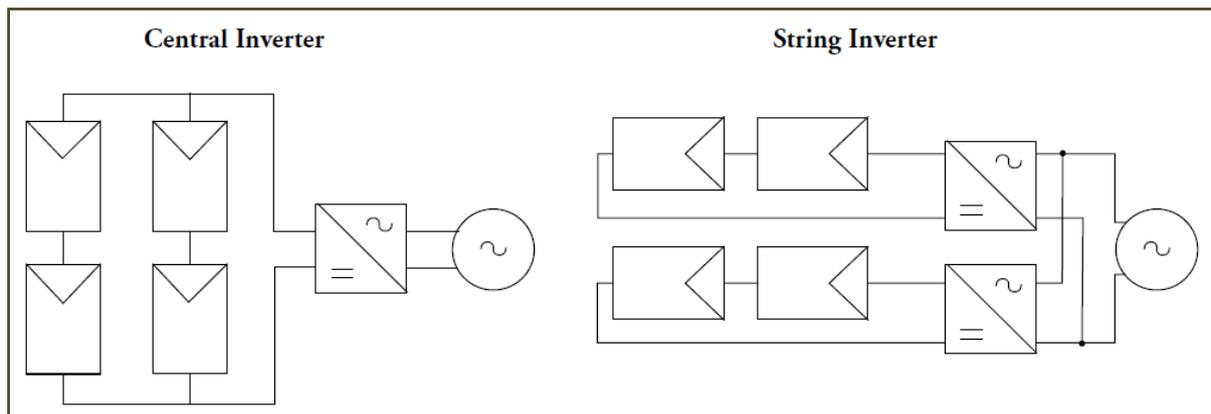


Figure 11 : PV System Configurations, [1]

In central inverters, large numbers of modules are connected in series to form a high voltage string. Strings are then connected in parallel to the inverter, Figure 8. Central inverter configuration is the first choice for many medium and large-scale solar PV plants. Central inverters offer high reliability and simplicity of installation. However, their disadvantages are: increased

mismatch losses and absence of maximum power point tracking for each string. This may cause problems for arrays that have multiple tilt and orientation angles, suffer from shading, or use different module types.

Central inverters are usually three-phase and can include grid frequency transformers.



Figure 12 : First Solar 40-MW CdTe PV Array installed by JUWI Group in Waldpolenz, Germany, [19]

The transformer's location in the Waldpolenz Solar Park, shown in Figure 12 is divided into blocks each with a centralized inverter.

String inverters are substantially lower in capacity, of the order of 10kW, and condition the output of a single array string. This is normally a whole, or part of, a row of solar arrays within the overall plant. String inverters can enhance the efficiency of solar parks, where different parts of the array are experiencing different levels of insolation, for example where arranged at different orientations, or closely packed to minimize site area. While numerous string inverters are required for a large plant, individual inverters are smaller and more easily maintained than a central inverter.

VI. GROUND MOUNTING

PV modules must be mounted on a structure to keep them correctly oriented and provides them with structural support and protection. The mounting structures may be either fixed or tracking. The fixed tilt mounting system is simpler, cheaper and has lower maintenance compared to than tracking systems. The tracking systems are more expensive and more complex, but can be cost-effective in locations with a high proportion of direct irradiation.

Most solar parks use ground mounted (sometimes called free-field or stand-alone) arrays. Land area required for solar parks varies depending on the location, and on the solar modules' efficiency, the slope of the site and the type of mounting used. Fixed tilt solar arrays using typical modules of about 15% efficiency on horizontal sites, need about 10,000 m²/MW.

a) Fixed Tilt

The solar panels in many PV stations are mounted on fixed structures, and thus have fixed inclination calculated to provide the optimum annual output profile, and is generally optimized for each PV power plant according to its location. This helps to maximize the total annual energy yield. These are normally oriented towards the Equator, at a tilt angle slightly less than the latitude of the site. Note that the tilt angle or "inclination angle" is the angle of the PV modules from the horizontal plane. The orientation angle or "azimuth" is the angle of the PV modules relative to south; East is -90° south is 0° and west is 90°.

Fixed tilt mounting systems are simpler, cheaper and have lower maintenance requirements than tracking systems. Frames to carry the PV panels are built first, and then the PV panels are fixed on the frame as shown in Figures 10a-10c, [20].



(a)



(b)





(c)

Figure 13a-13c : Fixed mounting arrangement, [20]

Example of fixed mounted PVPS is the Five Points Solar Station, which has Capacity equal 17.7 MW DC in Fresno County, California. The Five Points project is part of PG&E's 250-MW Utility Owned Generation (UOG) PV Program, a five-year plan for the construction of utility-owned solar PV stations, [15].

b) Seasonally Adjusted Tilt

As the majority of the solar energy is in the direct beam, maximizing collection requires the sun to be visible to the panels as long as possible. The tilt angle can be mechanically adjusted seasonally to optimize output in summer and winter. The angle is

usually adjusted twice or four times per year. These require more land area to reduce internal shading at the steeper winter tilt angle. Because the increased output is typically only a few percent, it seldom justifies the increased cost and complexity of this design. Figure 11 shows the arrangement of seasonally adjusted PV panels in photovoltaic power plant near Alamosa, Colorado. In this plant, the 82-acre tract site is one of the largest PV in the US. The Alamosa Photovoltaic Plant, which went on-line in December 2007, and generates about 8.2 megawatts of power.



Figure 14 : Seasonally adjusted fixed-axis photovoltaic panels at the Sun Edison photovoltaic power plant near Alamosa, Colorado

c) *PV Panels tracking*

Having the direct (beam) radiation, main part of the global radiation, perpendicular on the PV panel surface as much as possible maximizes the energy collected and thus the yield. The main factor affected

the energy contributed by the direct beam is the cosine angle between the incoming light and the panel (angle i). The power lost due to deviation of this angle is given in Table 6, and Fig. 15.

Table 6 : Direct power lost (%) due to misalignment (angle i), [19]

i	Lost = $1 - \cos(i)$	i	hours	Lost
0°	0%	15°	1	3.40%
1°	0.02%	30°	2	13.40%
3°	0.14%	45°	3	30%
8°	1%	60°	4	>50%
23.4°	8.30%	75°	5	>75%

Trackers with accuracies of $\pm 5^\circ$ can deliver greater than 99.6% of the energy delivered by the direct beam plus 100% of the diffuse light. Thus, high

accuracy tracking is not usually used in non-concentrating PV applications.

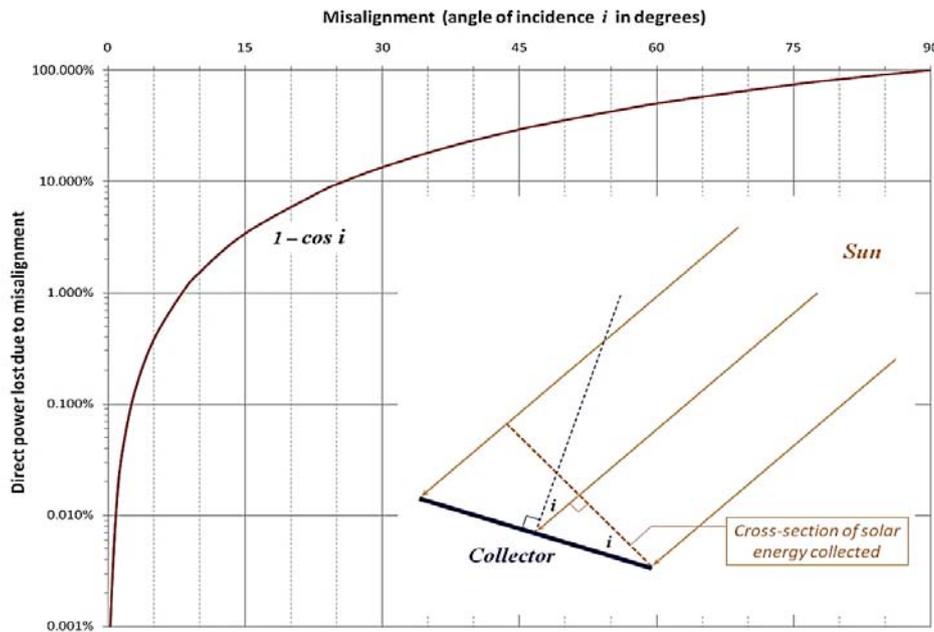


Figure 15 : The effective collection area of a flat-panel solar collector varies with the cosine of the misalignment of the panel with the Sun, [21]

Tracking will always result in a higher energy yield. The amount of the boost however is very much dependent on the location. Generally, locations with a

higher proportion of direct sunlight will benefit more from tracking than locations with a high proportion of diffuse light such as Germany, see Table 4.

Table 4 : Boost of PV system by tracking, [20]

	Flat Panel horizontal surface	Fixed mounting, optimum angle	1-axis tracking	1-axis with seasonal adjustment	2-axes tracking
Energy boost in comparison to optimum tilt	-15%	0%	20%	26%	32%
Initial marginal cost per m ²	0%	5%	10%		20%

Tracking increases the performance ratio of a system. It also results in higher yields for the inverter. Dual-axis tracking systems increase the average total annual irradiation in locations with a high proportion of direct irradiation. Tracking systems follow the sun as it moves. Orienting the solar panels to be normal to the sun's rays maximizes the intensity of incoming direct radiation. The two axis tracking system enables tracking the sun in its daily orbit across the sky, and as its elevation changes throughout the year. The arrays have to be spaced out to reduce inter-shading as the sun moves and the array orientations change. So, it needs more land area. The maximum increased output can be of the order of 30% in locations with high levels of direct radiation, but the increase is lower in temperate climates or when diffuse radiation is significant, due to overcast

conditions. Schematic increase of power output due to the use of dual axis tracking is shown in Figure 12.

Tracking systems are generally the only moving parts employed in a PV power plant. Single-axis trackers either alter the orientation or tilt angle only, while dual-axis tracking systems alter both orientation and tilt angle. Dual-axis tracking systems are able to track the sun more precisely than single-axis systems. Depending on the site and precise characteristics of the solar irradiation, trackers may increase the annual energy yield by up to 27% for single-axis and 37% for dual-axis trackers. Tracking also produces a smoother power output plateau, as shown in Figure 15. This helps meet peak demand in afternoons, which is common in hot climates due to the use of air conditioning units.

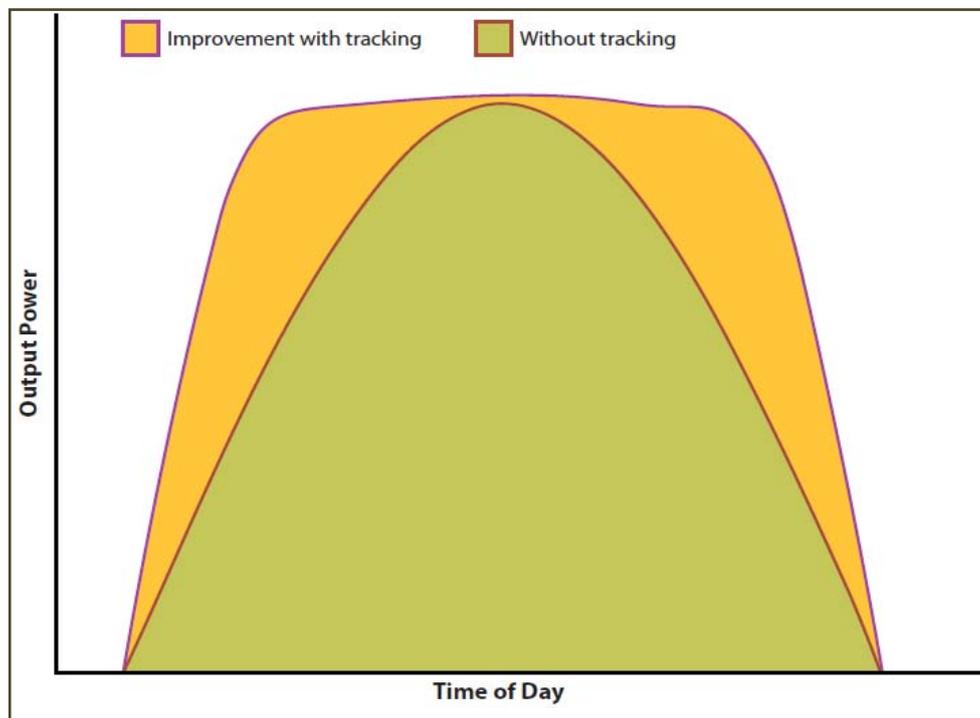


Figure 15 : Benefit of Dual Axis Tracking System, [1]

The tracking system needs additional capital costs for the procurement and installation of the tracking systems (typically \$140-700/kWp); and added additional maintenance costs range from \$2.8-21/kWp per annum, [1]. It also needs additional land area required to avoid shading compared to a free field fixed tilt system of the same nominal capacity. Almost all tracking system plants use crystalline silicon modules because their higher efficiency. This reduces additional capital and operating costs required for the tracking system (per kWp installed). Examples of PVPS using dual axis tracking systems are: Bellpuig Solar Park near Lerida, Spain uses pole-mounted 2-axis trackers; and the Erlasee solar park installed by Solon Mover L plant of 12 MWp, shown in Figure 16.



Figure 16 : Azimuth-altitude dual axis tracker - 2 axis solar tracker, Toledo, Spain

Tracking the sun in one dimension can achieve some of the output benefits of tracking, with a less penalty in terms of land area, capital, and operating

cost. A single axis tracker with roughly 20 degree tilt at Nellis Air Force Base in Nevada, USA is shown in Figure 14.



Figure 17 : Single axis trackers with roughly 20 degree tilt at Nellis Air Force Base in Nevada, USA. The arrays form part of the Nellis Solar Power Plant. Credit: U.S. Air Force photo by Senior Airman Larry E. Reid Jr,

VII. ECONOMY OF PVPS

a) *Levelized Cost of Energy (LEC) of Solar PV Systems*

The levelized cost of energy (LEC) of solar PV systems reflects the price at which energy must be sold to break even over the assumed economic life of the system. In other words, it is the cost incurred to install and maintain an energy-producing system divided by the energy the system will produce over its lifetime of operation:

$$LEC = \text{Life time energy cost} / \text{Life time energy generation}$$

This equation yields a net present value in the familiar cents per kilowatt-hour (kWh) of electricity generated. This is an assessment of the economic lifetime energy cost and energy production and can be applied to essentially any energy technology. It is frequently used to evaluate a technology or energy system against electricity purchased from the grid. The LEC equation takes into account system costs, as well as factors including financing, insurance, operations and maintenance (O&M), depreciation and any applicable incentives. Installed costs are a primary driver for solar PV systems as they lack fuel costs and require minimal O&M.

By knowing that the EP produced by PVPS is higher than the EP retail price, it is required to identify if and when the declining LEC of solar PV intersects with

the increasing retail electricity prices. The term frequently used to describe this intersection is “grid parity”. The installed cost of solar PV systems is the largest component of the LEC.

The installed price of utility-scale systems varies significantly across projects. In the US, among 49 projects completed in 2011, for example, installed prices ranged from \$2.4/W to \$6.3/W, reflecting the wide variation in project size (from 2 MW to 35 MW), differences in system configurations (e.g., fixed-tilt vs. tracking and thin-film vs. crystalline modules), and the unique characteristics of individual projects, [20]. It is noticed that for very large PVPS plant of 187.5 MWP DC one-axis utility-scale ground mount, the estimated cost was \$4.40/WP DC, or \$ 5.9/W (by considering 0.75 De-rate Factor from DC to AC). So, for Qatar and 50 MW plant in Qatar if 20% increase is assumed the price would be \$7.04, and the plant will cost 352 million (M). In another study for India, 169 Indian Rupee (\$3)/W were reported. Again, if this for peak DC, and by considering 0.75 De-rate Factor from DC to AC it would be \$4/W, [1].

A study to calculate the LEC by North Carolina State University indicated that for 10 MW plant made the following assumptions: the installed cost is \$3.75 -\$5/W, economic life of system is 20 years, fixed operation and maintenance is \$50-65 kW/year, capacity factor 15-28%, the LEC is \$0.24-0.46/kWh, [21]. The cost breakdown was given in Fig. 18.

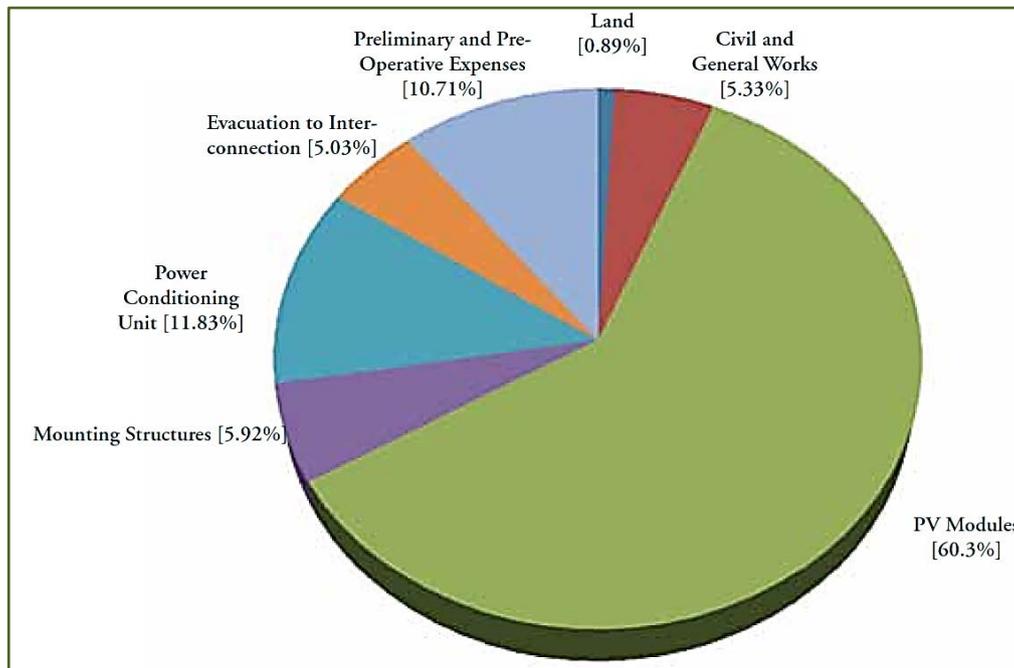


Figure 18 : Cost breakdown of PV power station, [21]

b) *Utilities as Contractual Intermediaries*

The utility in Qatar is acting as contractual intermediary agent between the power producer and the customers. The owner of the power plant sells power

output from the plant (it is solar PV system here) to the utility, which, in turn, sells the power back to the site host/end-user.

This arrangement protect consumers (rates and reliability) and to ensure a highly functioning electric grid. By having a single entity control the system, a utility can balance constantly changing supply and demand to ensure reliability and keep the electricity flow on the grid optimized and safe.

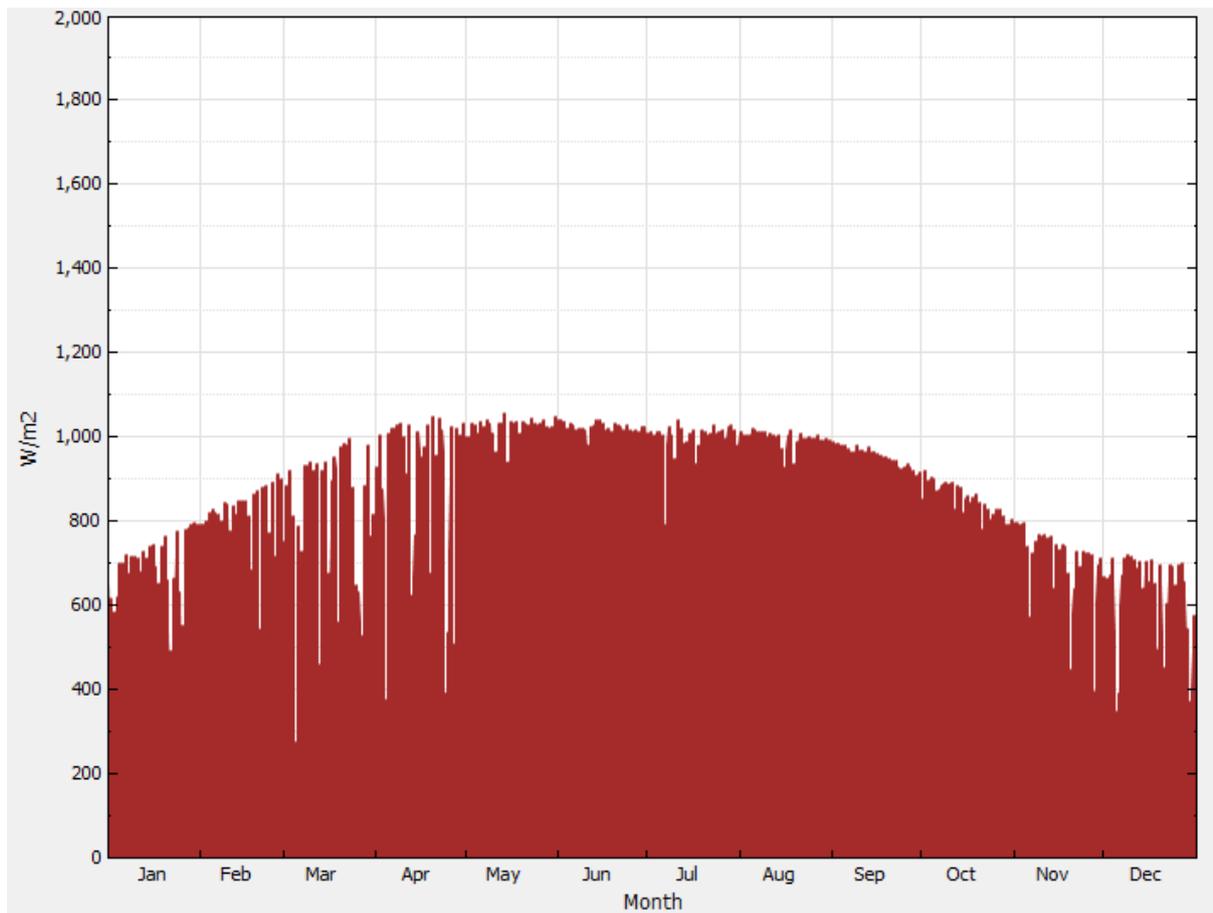
program developed by NREL is used here to figure the PVPS characteristics, capital cost, and the EP output cost in \$/kWh. Because no boundary conditions data are available for Qatar, the environmental data for Cairo, Egypt is used in running the program. This environmental data is given as follows:

VIII. SUGGESTED PVPS USING FLAT TYPE PV SYSTEM AND SAM COMPUTER PROGRAM RESULTS

A PV array of 20 MW_{dc} is suggested in this study. The System Advisor Model (SAM) computer

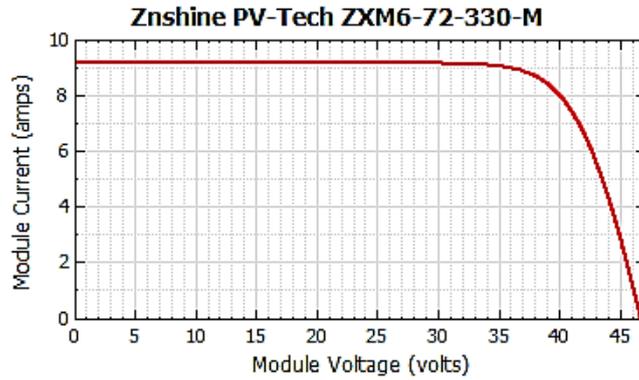
Location Information					
City	ABU DHABI	Time Zone	GMT 4	Latitude	24.43 deg
State	-	Elevation	27 m	Longitude	54.65 deg

Weather Data Information (Annual)				
Direct Normal	2294.9 kWh/m ²	Dry-bulb Temp	27.1 °C	View hourly data...
Global Horizontal	2204.6 kWh/m ²	Wind Speed	3.6 m/s	



Global Horizontal, W/m²

PV Module: Znshine PV-Tech ZXM6-72-330-M

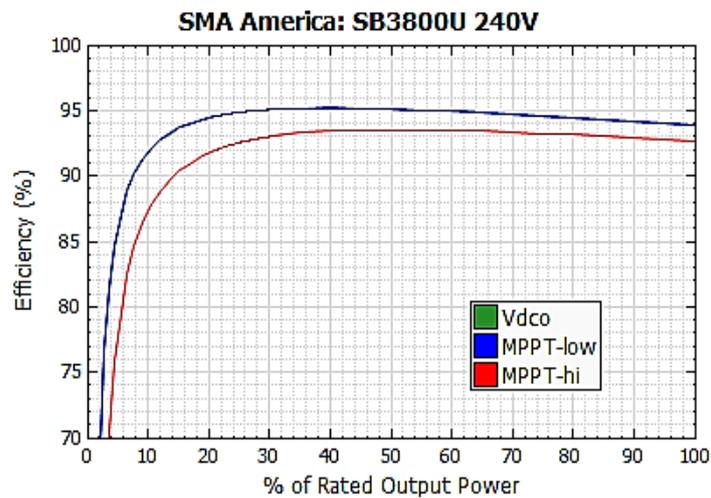


Efficiency	17.00 %	Temperature Coefficients	
Maximum Power (Pmp)	329.998 Wdc	-4.530e-001 %/C	-1.495e+000 W/C
Max Power Voltage (Vmp)	38.15 Vdc		
Max Power Current (Imp)	8.65 Adc		
Open Circuit Voltage (Voc)	46.72 Vdc	-3.200e-001 %/C	-1.495e-001 V/C
Short Circuit Current (Isc)	9.19 Adc	5.600e-002 %/C	5.146e-003 A/C

Physical Characteristics

Material	Mono-c-Si	Module Area	1.941 m ²	Number of Cells	72
----------	-----------	-------------	----------------------	-----------------	----

Inverter: SMA America: SB3800U 240V



CEC weighted efficiency	94.5729	%
European weighted efficiency	94.2751	%
Maximum AC power	3800	Wac
Maximum DC power	4050.67	Wdc
Power consumption during operation	24.7974	Wdc
Power consumption at night	0.161	Wac
Nominal AC voltage	240	Vac
Maximum DC voltage	600	Vdc
Maximum DC current	18	Adc
Minimum MPPT DC voltage	250	Vdc
Nominal DC voltage	251.5	Vdc
Maximum MPPT DC voltage	480	Vdc

The array consists of 60646 modules of 117,702 m² total area, connecting by 7580 strings, or 8 modules per string. The nameplate capacity and Vmp are at module reference conditions. Voc (string) is 373.76 V at 1000 W/m² irradiance and 25°C, and Vmp (string) =

305.2 V. The module material is mono Crystalline Silicon (c-Si). The number of used inverters is 5500, V dc max (dc-inverter) 600 V. Data on module's dimensions are given as:

Actual Layout			
Modules	Inverters		
Nameplate capacity	20011 kWdc	Total capacity	20900 kWac
Number of modules	60640	Total capacity	22278.7 kWdc
Modules per string	8	Number of inverters	5500
Strings in parallel	7580	Maximum DC voltage	600 Vdc
Total module area	117702 m ²	Minimum MPPT voltage	250 Vdc
String Voc	373.76 V	Maximum MPPT voltage	480 Vdc
String Vmp	305.2 V		

Nameplate capacity and string Vmp are at module reference conditions. String Voc is at 1000 W/m² incident irradiance and 25 °C cell temperature.

Module

Orientation: Landscape

Length: 2.884 m

Width: 0.673 m

Number of Cells along Length: 12

Number of Cells along Width: 6

Number of Bypass Diodes: 3

Characteristics from Module Page

Area: 1.941 m² Number of cells: 72



The string wiring is shown as follows:

Array

String Wiring

Number of Strings along Bottom Side Length m

Number of Modules along Bottom Row Spacing m

Number of Modules along Side Number of Rows

Layout from Array Page

Modules per String Strings in Parallel

Ground Reflectance

Use albedo in weather file if it is specified

Monthly ground reflectance (albedo)

Tilted Surface Radiation Model (Advanced)

Isotropic
 HDKR
 Perez

-Radiation Components-

Beam and diffuse
 Total and beam

Interconnection Derates (AC)

AC wiring losses (0..1)

Step-up transformer losses (0..1)

Total interconnection derate (0..1)

Land Area

Packing factor

Total land area acres

The tracking and orientation are given as:

-String Configuration-

Strings in array

Strings allocated to subarray

-Tracking & Orientation-

Azimuth

N = 0

Tilt

Vert = 90

Fixed

1 Axis

2 Axis

Azimuth Axis

Tilt (deg)

Azimuth (deg)

Tracker rotation limit (deg)

Shading mode for 1 axis tracking

Ground coverage ratio (GCR)

Details of capital cost are given by:
Direct capital cost

Direct Capital Costs						
Module	60640 units	0.3 kWdc/unit	20011 kWdc	\$ 3.302	\$/Wdc	\$ 66,076,481.82
Inverter	5500 units	3.8 kWac/unit	20900 kWac	\$ 0.61	\$/Wac	\$ 12,749,000.00
Balance of system, equipment	0 \$		0.458 \$/Wdc	0 \$/m2		\$ 9,165,060.17
Installation labor	8100 \$		0 \$/Wdc	0 \$/m2		\$ 8,100.00
Installer margin and overhead	0 \$		0 \$/Wdc	0 \$/m2		\$ 0.00
Contingency				0 %		\$ 0.00
Total Direct Cost						\$ 87,998,641.98

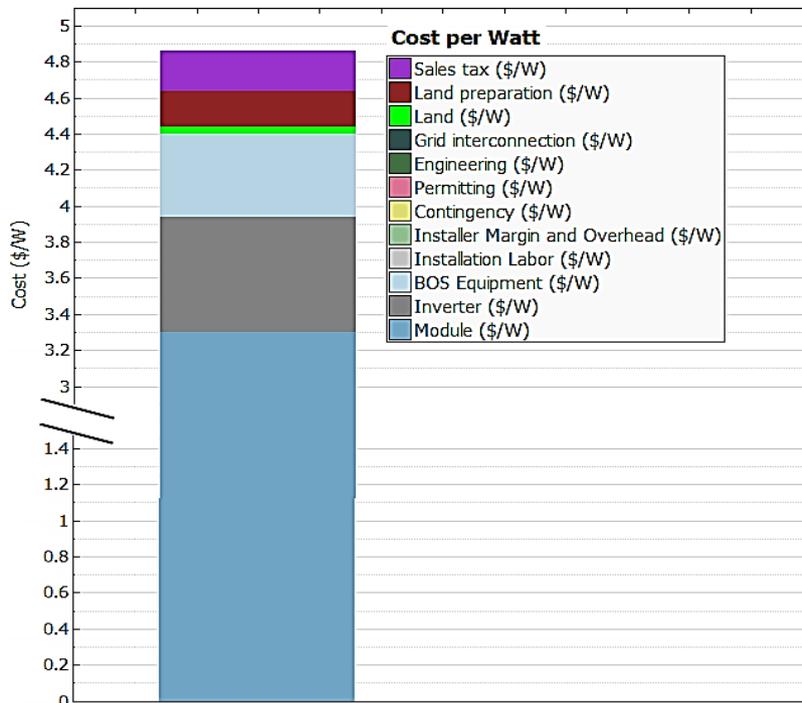
Indirect cost

Indirect Capital Costs				
	% of Direct Cost	Cost \$/Wdc	Fixed Cost	Total
Permitting, Environmental Studies	0 %	0.00	\$ 0.00	\$ 0.00
Engineering	0 %	0.00	\$ 800.00	\$ 800.00
Grid interconnection	0 %	0.00	\$ 0.00	\$ 0.00
Land Costs				
Total Land Area	72.7106 acres			
Cost \$/acre				
Land	0.00	0 %	0.04	\$ 800,441.94
Land preparation	0.00	0 %	0.20	\$ 4,002,209.68
Sales Tax of	5 %	applies to	100 %	of Direct Cost
				\$ 4,399,932.10
Total Indirect Cost				\$ 9,203,383.72

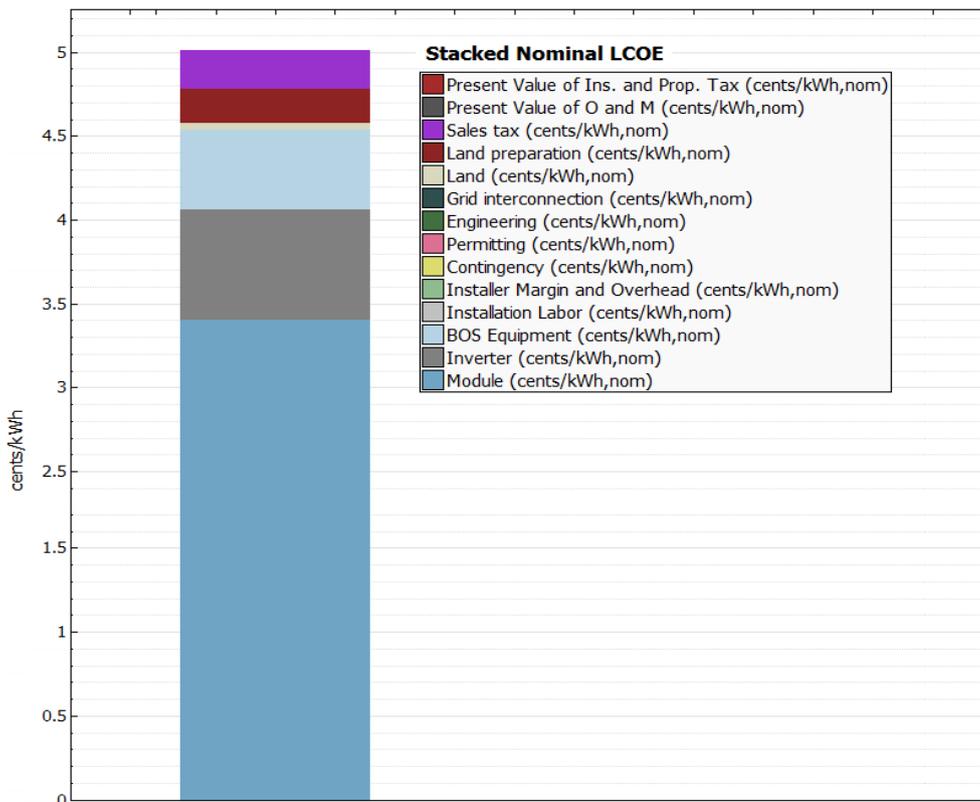
Total installed cost excluding any financing cost

Total Installed Costs	
Total Installed Cost	\$ 97,202,025.70
Total Installed Cost per Capacity (\$/Wdc)	\$ 4.86
Total Installed Cost excludes any financing costs	

The itemized capital cost is given as:



The levelized Electricity Cost (LEC) is given as:



IX. CONCLUSIONS

The technology and economics of the PV power station is presented in this paper. The main components

of the PV power plants including the solar PV modules, module mounting and tracking systems, inverters (or converters), and step-up transformers was outlined. It reviews the materials of the PV cells, the PV cells

degradation, and the existing PV power plant. Utility PV power plants around the world were also reviewed.

The System Advisor Model (SAM) software developed by National Renewable Energy Laboratory (NREL) has been used to predict the total direct capital cost of the 20 MW PV plant as \$88.0 million (M), and total installed cost as \$ 97.202 M; or \$4.86 M/MW. This is almost half the cost of the CSP using parabolic trough plants. The LCOE, as given by the computer program is \$0.16/kWh. The main disadvantage of the PV power station is non-dispatch ability.

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Evaluation of Residual Stress Field by X-Ray Diffraction using the Sen²Ψ Method on E308L Stainless Steel Weld Overlays

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Abstract- The application of metal coatings by welding of steels and alloys resistant to corrosion is an alternative quite reasonable economically facing the manufacturing of solid components in these alloys. However there is a lack of information about the mechanical / metallurgical changes caused by thermal cycles of welding due to the application of these coatings, especially about its influence on the level of residual stresses, that it is essential to understand the phenomenon of stress-assisted corrosion. The aim of this study was to evaluate the level of the superficial residual stresses in coatings of AWS E308L T-1 stainless steel applied by the FCAW process on steel plates of ASTM A36. The measurements of residual stresses were performed by a portable X - ray diffractometer on the surface of the coatings and the main results obtained were that the nature of residual stresses on the surface of the coatings always showed compressive and it was not observed linearity of the magnitude of the residual stresses as a function of the welding heat input.

Keywords: *welding, stainless steel, residual stress, x-ray diffraction.*

GJRE-J Classification : *FOR Code: 291899p*



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Abstract- The application of metal coatings by welding of steels and alloys resistant to corrosion is an alternative quite reasonable economically facing the manufacturing of solid components in these alloys. However there is a lack of information about the mechanical / metallurgical changes caused by thermal cycles of welding due to the application of these coatings, especially about its influence on the level of residual stresses, that it is essential to understand the phenomenon of stress-assisted corrosion. The aim of this study was to evaluate the level of the superficial residual stresses in coatings of AWS E308L T-1 stainless steel applied by the FCAW process on steel plates of ASTM A36. The measurements of residual stresses were performed by a portable X - ray diffractometer on the surface of the coatings and the main results obtained were that the nature of residual stresses on the surface of the coatings always showed compressive and it was not observed linearity of the magnitude of the residual stresses as a function of the welding heat input.

Keywords: welding, stainless steel, residual stress, x-ray diffraction.

I. INTRODUCTION

Global economic pressures and characteristics of national producing basins have increasingly led refineries use heavy oil in their processes. This requires the materials used in plant of exploration and processing of oil and its derivatives excellent corrosion resistance and good mechanical properties [1]. Among

these metallic materials stainless steels and nickel super alloy can be highlighted [2, 3, 4]. However, the manufacture of solid components using these alloys makes them uneconomical equipment. Therefore, a viable alternative to such a problem is the application of metallic coatings by welding of these alloys on structural and piping carbon steel [5].

However, the welding of dissimilar materials leads to rise residual stress level due to difference between coefficients of thermal expansion and contraction [6, 7]. It is well known that the welding residual stresses have very significant influence on the fatigue life, crack growth, acceleration (or retardation) of corrosion processes assisted by tension and other effects [8, 9, 10].

In this context, this study aimed to evaluate the level of residual stresses in surface coatings of stainless steel E308L T1 applied by the FCAW process in plates of ASTM A36 steel.

II. MATERIALS AND METHODS

a) Base Metal and Consumables

In this paper ASTM A36 steel was used as the base metal and the AWS E308L T1 with a diameter of 1.2 mm as filler metal. Table 1 shows the chemical composition of the materials used according to the manufacturer.

Table 1 : Chemical composition (weight%) of employed consumables

	C	Cr	Ni	Mo	Mn	Si	Cu	P	S	Fe
AWS E-308L	0,03	19,5 - 22,0	9,0 - 11,0	0,75	1,0 - 2,5	0,30 - 0,65	0,75	-	-	Balance
ASTM A 36	0,18-0,23	-	-	-	0,30-0,60	-	-	0,03 máx.	0,05 máx.	Balance

b) Experimental Procedure

The welds were made by FCAW (Flux Core Arc Welding) process. An electronic welding source and a system of data acquisition for control of welding parameters was used. The welding procedure was performed without restriction and without weaving. The specimens consist of three weld beads deposited in the flat position by varying the voltage (U), the wire feed

speed (Fr) and travel speed (Ts) generating different values of heat input (H). The values of the welding parameters used in the experiments are shown in Table 2. Un overlap of 1/3 weld beadwidth was used as illustrated in Figure 1. Inclination of the welding torch ($\theta = 15^\circ$ to the vertical), welding direction "pushing" mode current with reverse polarity CC+ and contact tip away from the workpiece of 15 mm have been kept fixed.

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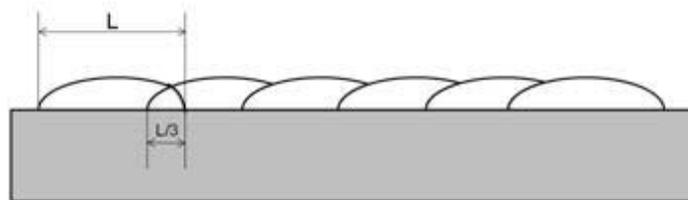


Figure 1 : Illustration of the overlap between the beads

Table 2 : Welding parameters

	U (volts)	Fr (m/min)	Ts (cm/min)	H (kJ/cm)
1	30,0	6,0	20,0	19,65
2	40,0	6,0	20,0	26,19
3	30,0	7,0	20,0	21,20
4	40,0	7,0	20,0	28,19
5	30,0	6,0	26,0	10,97
6	40,0	6,0	26,0	14,51
7	30,0	7,0	26,0	11,82
8	40,0	7,0	26,0	15,99
9	35,0	6,5	23,0	16,94
10	35,0	6,5	23,0	17,52
11	35,0	6,5	23,0	17,44

The stress analysis was performed on the surface of the coatings using a portable X-ray diffractometer (Figure 2).

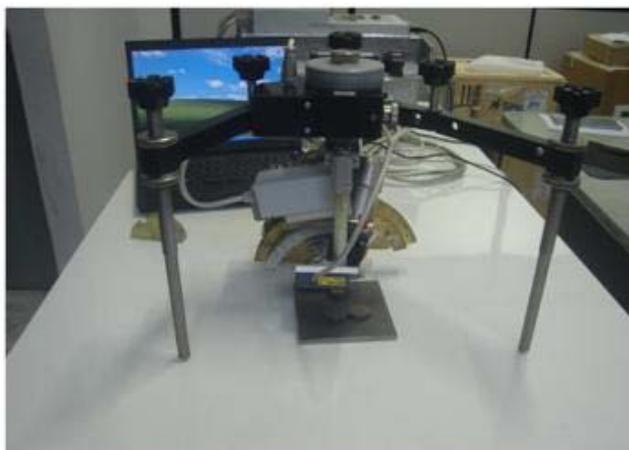


Figure 2 : Portable diffractometer used in the analysis of residual stresses

To examining austenitic steels planes {220} with a wavelength of $\lambda = 2.2896 \text{ \AA}$, and peak angle (2θ) = 128.84° were used. The wavelength is produced by a anodic tube of chromium [11].

M and F means the beginning, middle and end of coating.

The methodology used was the sen²ψ with measurements for $\psi = 0^\circ, 30^\circ, 33^\circ, 35^\circ, 37^\circ, 40^\circ, 42^\circ, 43^\circ, 44^\circ$ and 45° . Analyses were performed in three distinct regions of all coatings. Figure 3 illustrates the arrangement of the points that were analyzed, where I,

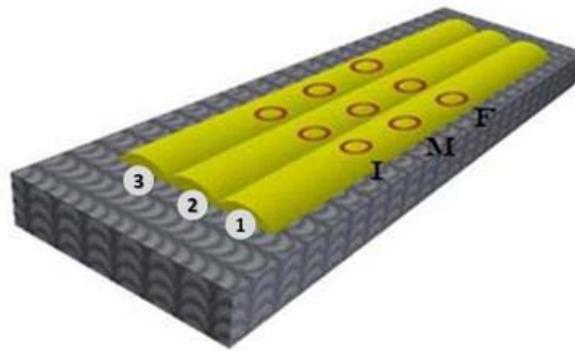


Figure 3 : Analyzed regions on the coatings

III. RESULTS

In Figures 4 to 9 are plotted the longitudinal and transverse fields of residual stresses for coatings FCAW

- 1, 2 and 5. These coatings were chosen because were done using extremes and intermediates heat input values: very low ($H_5 = 10.97 \text{ kJ / cm}$), very high ($H_2 = 26.19 \text{ kJ / cm}$) and intermediate ($H_1 = 19.65 \text{ kJ / cm}$).

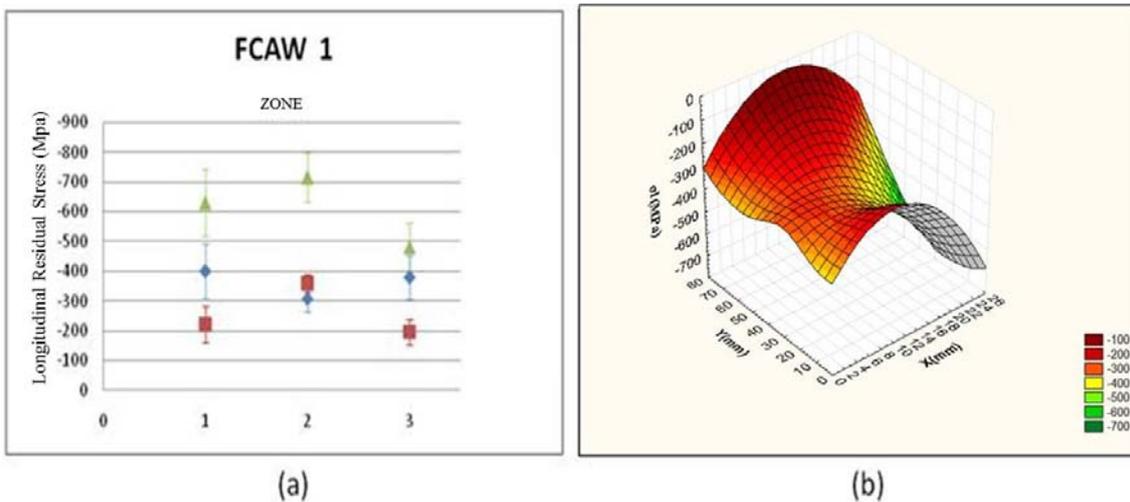


Figure 4 : Mean values of longitudinal residual stresses and their respective uncertainties (a); Field of longitudinal residual stresses into the coating (b)

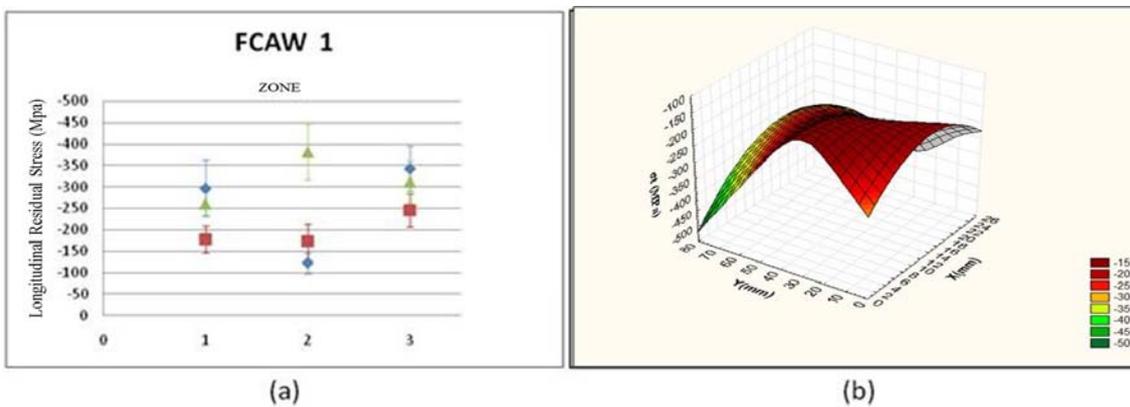


Figure 5 : Mean values of transverse residual stresses and their respective uncertainties (a); Field transverse residual stresses (b) in coating

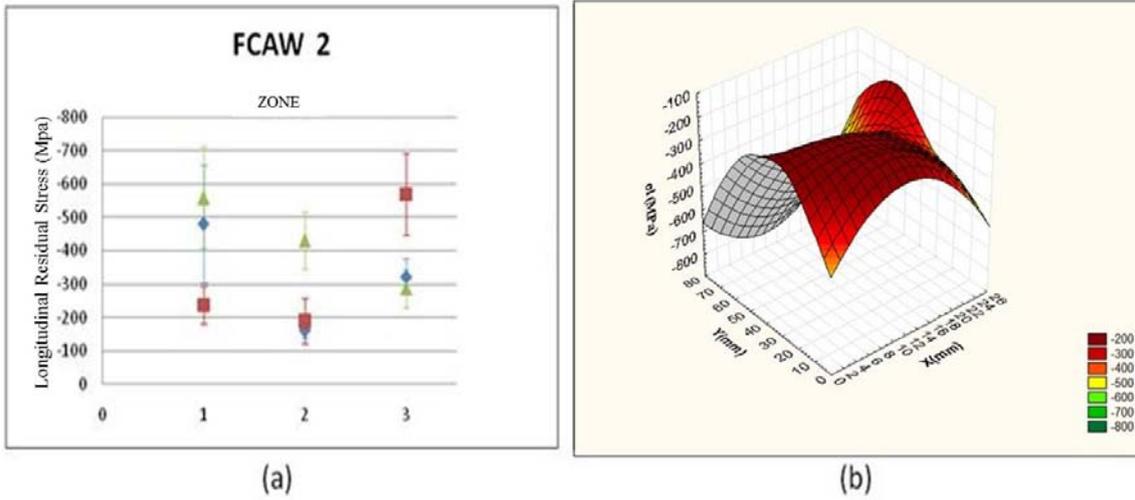


Figure 6 : Mean values of longitudinal residual stresses and their respective uncertainties (a); Field of longitudinal residual stresses into the coating -2(b)

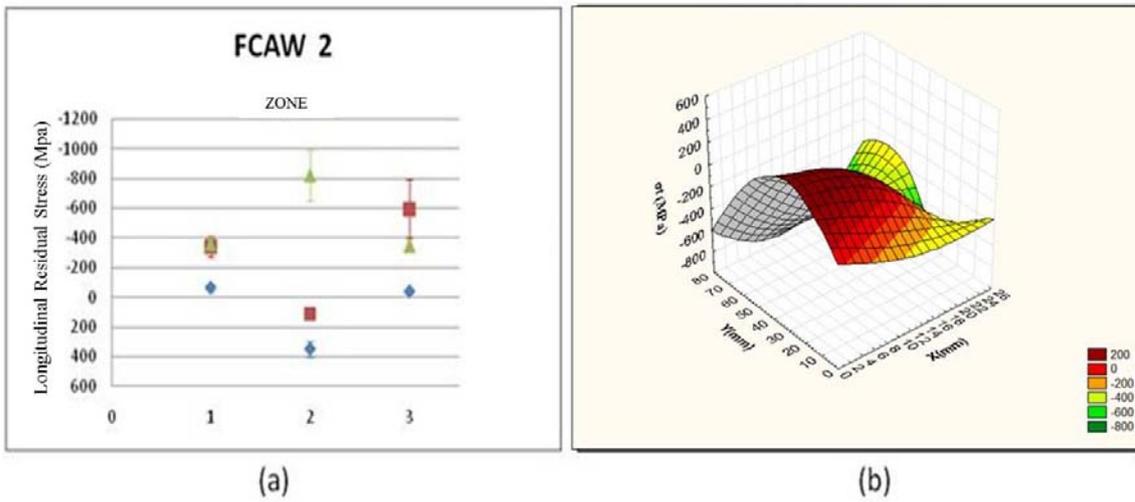


Figure 7 : Mean values of transverse residual stresses and their respective uncertainties (a); Field transverse residual stresses into the coating - 2 (b)

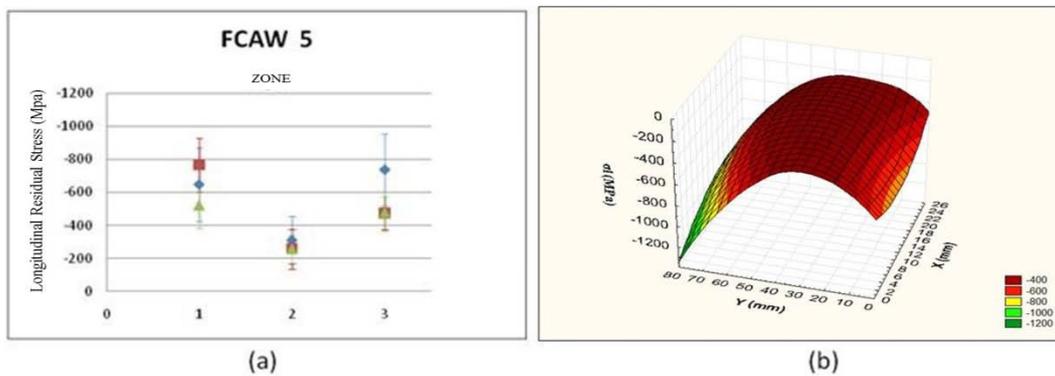


Figure 8 : Mean values of longitudinal residual stresses and their respective uncertainties (a); Field of longitudinal residual stresses into the coating - 5 (b)

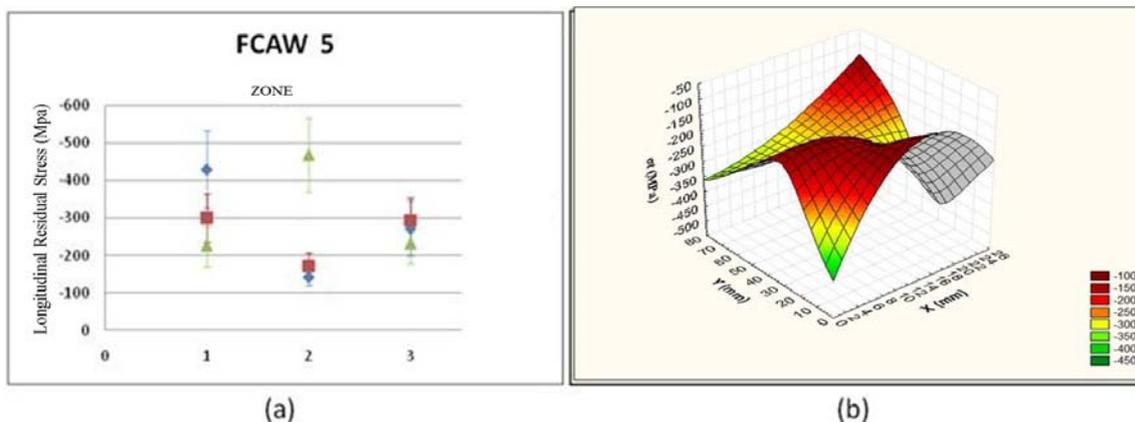


Figure 9 : Mean values of transverse residual stresses and their respective uncertainties (a); Field transverse residual stresses for into the coating - 5 (b)

From these graphs, it is noted that the central region of the coatings is more compressive than the periphery, and that the residual stress field develops compressive values in the top coating with decrease and increase again at the end of coating. Results from the literature [12, 13] indicate that the 18-8 alloys (% Cr-% Ni) show higher variation of the residual stresses along the width of the weld bead. These fluctuations indicate that the residual stresses are not constant in the irradiated areas.

Figure 4 (a) shows that the maximum value of longitudinal residual stress was about 700 MPa in modulus in region 2 at the end of the coating. The minimum value in modulus was 200 MPa, in the beads 1 and 3 through the coating. Figure 5 (a) shows that the transverse residual stresses are smaller in magnitude than the longitudinal. The maximum transverse residual stress was 380 MPa, compared to 700 MPa (longitudinal residual stresses) in the same region.

It can be seen from the Figure 7 (a) the presence of residual tensile stresses at the beginning

and middle of the coating bead, subsequently evolving into a state of compressive stress at the end of the coating. Bezerra et al. [14] hypothesized that welding defects such as discontinuities and lack of fusion of the weld metal can contribute to justify the presence of these tensile stresses in these regions.

Figures 8 and 9 show a distribution of the residual stress field similar to previous coatings, although the magnitude of the residual stress values due to the change significantly greater amount of welding energy with which these coatings were applied. Cardoso [15] found that joints subjected to different welding energies have similar distribution of residual stress on the surface, but may have different magnitudes of stresses.

The nature of the distribution of residual stress is closely linked to the nature of electric arc welding characteristic of each process [16].

Table 3 shows the average values of longitudinal and transverse residual stresses and their deviations and coefficients of variation.

Table 3 : Average residual stresses for all conditions of welding by FCAW process

	Longitudinal (MPa)			Transverse (MPa)		
	Average	Desviation	CV	Average	Desviation	CV
1	-410	68	0,1658	-257	44	0,1712
2	-344	88	0,2558	-233	70	0,3004
3	-373	74	0,1984	-260	54	0,2077
4	-591	143	0,2420	-278	54	0,1942
5	-494	144	0,2915	-167	63	0,3772
6	-498	110	0,2209	-286	53	0,1853
7	-393	119	0,3028	-250	57	0,2280
8	-470	111	0,2362	-361	75	0,1847
9	-506	142	0,2806	-195	65	0,3333
10	-456	109	0,2390	-251	110	0,4382
11	-390	107	0,2744	-241	65	0,2697

Table 3 gives an idea of the magnitude of the average longitudinal and transverse residual stresses in the coatings. It is noteworthy that the transverse residual stresses values had lesser magnitude than the longitudinal residual stress, corroborating the results of KOU [7]. All coefficients of variation (CV) are observed near 0.3; as criteria for acceptance in this work it was established that values lower than 0.5 CV are needed to express the average representative form, and CV values close to 0.3 are considered optimal.

The nature of the average residual stresses in the coatings was compression, however, it was observed during the individual analyzes some tensile residual stresses value, indicating the probable presence of welding defects. The compressive nature of the stresses in these coatings can be explained due to a physical characteristic of the filler metal involved. During

the process of solidification austenitic stainless steel undergoes two phase transformations, the first liquid metal to δ ferrite, body-centered cubic structure, the second transformation is the δ ferrite to austenite structure which is cubic face centered. In this last phase transformation, there is an expansion of the unit cells, since the parameter of the face-centered cubic structure network is greater than the body-centered cubic [17]. This expansion opposes the nature of the substrate, which is ferritic, which tends to contract during the solidification process, combined with the mismatch between the coefficient of thermal expansion of stainless steel (17 - 19 mm m⁻¹ K⁻¹) and the structural substrate generates compressive residual stresses on the surface of the coating [12, 13, 18].

Figure 10 illustrates the influence of heat input on the level of residual stresses.

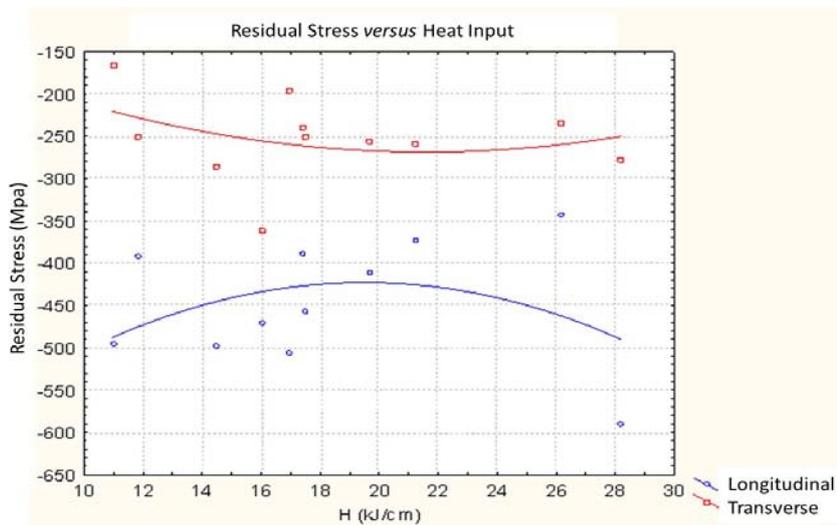


Figure 10 : Influence of heat input on the level of longitudinal and transverse residual stresses, and the interaction between the stress levels in the FCAW process

In this paper, the results do not indicate a linear relationship between the levels of residual stresses and the values of welding energy. This non-linearity can be attributed to competitive phenomena governed by the heat input and its directly influence the magnitude and nature of the residual stresses as:

- The the volumetric fraction of δ ferrite present in the weld metal, whose amount is proportional to the welding energy [19]. From there it can be hypothesize that the presence of δ ferrite can induce compressive residual stresses present in the weld metal;
- The size of the molten pool, where a larger restriction is imposed on the welded assemblies with lower energy, since they do not have major penetrations, which results in a large volume of all that remains solid during welding thereby increasing the restriction and consequently raising the level of residual stress;

- Welded specimens with higher heat input have a large amount of deposited material which results in high levels of residual stress.

Figure 11 shows the volumetric fraction of δ ferrite, evaluated qualitatively by the method of meshes [20]. The CP 5 coating was applied using a heat input $H = 10.97$ kJ / cm, this welding condition presented 23 mesh nodes on the δ ferrite. The CP 8 was applied using a heat input $H = 15.99$ kJ / cm with present 45 mesh nodes on the δ ferrite. In general, it can be noticed that increasing heat input occurs an increase in the the volumetric fraction of δ ferrite present in the weld metal. Comparing the information in Table 3 with those in Figure 11it can be noted that this increase results in a slight reduction of the longitudinal residual stresses and there is an increase of transverse residual stresses.

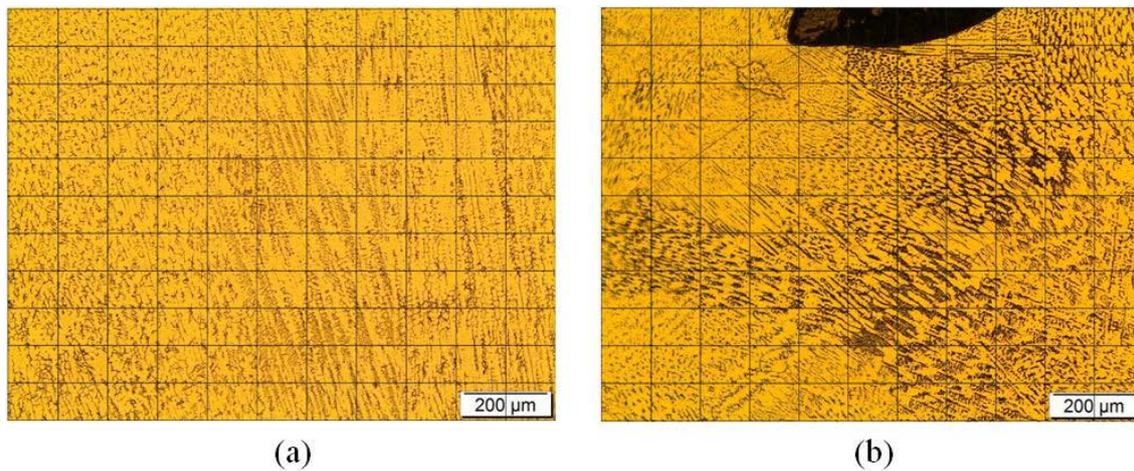


Figure 11 : Volumetric fraction of δ ferrite in the weld metal of the specimens CP 5 (a) and CP 8 (b)

Thus, for the range of low to medium welding heat input the magnitude of the longitudinal welding residual stress decreases with increasing welding energy due to the increased the volumetric fraction of δ ferrite present in the weld metal. However, there is a sharp increase of the transverse residual stresses.

Although it may be inferred that when the longitudinal residual stresses become less compressive the transverse residual stresses tend to become more compressive in order to keep a balance between these stresses.

IV. CONCLUSIONS

- The surface residual stresses in the coatings were predominantly compressive. The central region of the coatings always showed less compression than the periphery;
- It was not observed a linear relationship between the magnitude of the residual stresses and the welding heat input due to competitive phenomena such as precipitation of δ ferrite in the weld metal and the volume of material deposited in the application of coatings using high welding heat input;
- In general, the microstructure observed in the weld metal was austenite with δ ferrite, and the volumetric fraction of δ ferrite may play a significant influence on the magnitude of the residual stresses.

V. ACKNOWLEDGEMENTS

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The Design and Simulation Patterns in Ultrasonic Wedges for Non-Destructive Testing

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Abstract- The design and simulation patterns in ultrasonic wedges for non-destructive testing are critical to the assessment and characterization of a material or structure. Its significance is the mitigation of failures in members. The approach is relevant mostly to non-destructive testing of very expensive and critical mechanical members requiring high level of reliability in operation. In this paper, the design and simulation of ultrasound through both straight and angle wedges are discussed. Also, the characterization of the material using pulse-echo and through-transmission ultrasonic methods is illustrated.

Keywords: design, simulation, ultrasonic wedges, nondestructive testing.

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The Design and Simulation Patterns in Ultrasonic Wedges for Non-Destructive Testing

Ibe, G. A

Abstract- The design and simulation patterns in ultrasonic wedges for non-destructive testing are critical to the assessment and characterization of a material or structure. Its significance is the mitigation of failures in members. The approach is relevant mostly to non-destructive testing of very expensive and critical mechanical members requiring high level of reliability in operation. In this paper, the design and simulation of ultrasound through both straight and angle wedges are discussed. Also, the characterization of the material using pulse-echo and through-transmission ultrasonic methods is illustrated.

Keywords: design, simulation, ultrasonic wedges, non-destructive testing.

I. INTRODUCTION

The performance and effectiveness of ultrasonic wedges in non-destructive testing cannot be optimised except appropriate design simulation approach is adopted. Design and simulation of the ultrasound will fix problems of wedge configuration and the interface angle [1]. Ultimately, this approach will enable the simulation of the sound wave into the test sample.

In ultrasonic testing, both longitudinal and shear waves can be transmitted into the specimen. However, refracted shear wave is exploited in angle beam inspection because of its low attenuation [2]. Most

importantly, when refracted shear waves are utilized only in the inspection, the refracted longitudinal waves align with the material interface, enabling easy and accurate interpretation of signals [3]. The angle of the incident beam at which the parallel alignment of the longitudinal waves with the specimen surface occurs, is called the first critical angle.

Apart from the benefit of having one wave mode in the sample, the critical angle allows the inspection of sample surfaces such as weldments. For this mode-converted system, the transfer of energy is optimised in steel. Also the defect sensitivity is enhanced in the presence of shear waves [3].

II. WEDGE DESIGN

To optimally design an ultrasonic wedge, some basic specifications must be made. The following are the specifications that were used in the study.

- Longitudinal waves are refracted into the test sample at 90° .
- Wedge is Rexolite with longitudinal sound velocity at 2362.2 m/s.
- The test sample is made of alloy steel (X90CrMoV18) with shear sound velocity at 2478 m/s.

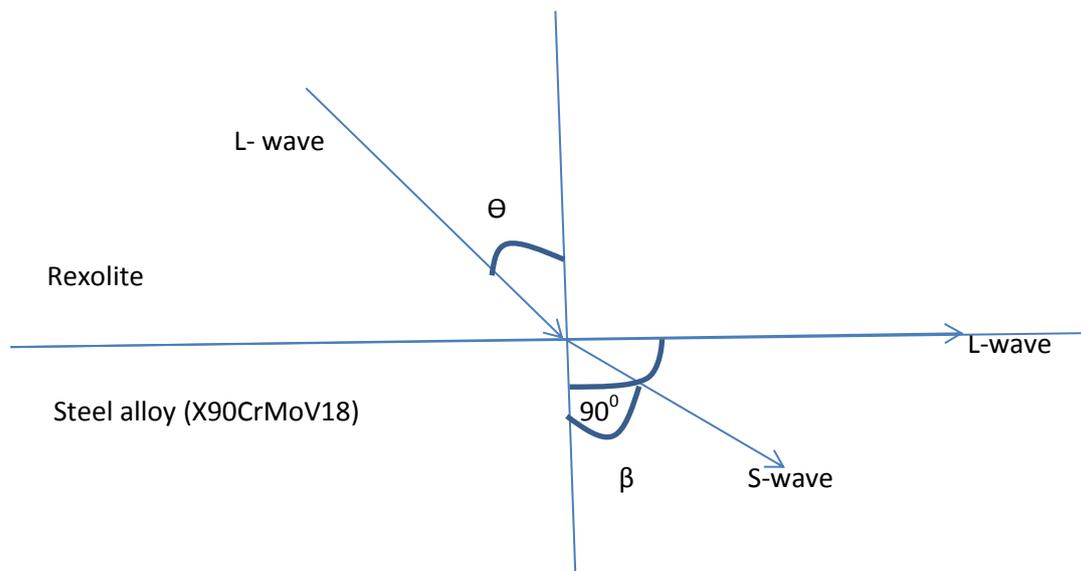


Fig. 1 : Refraction of shear waves into the sample

From Fig.1, applying Snell's law:

$$\frac{\sin \theta}{\sin 90} = \frac{v}{u} \text{-----(1)}$$

Where θ = incident wedge angle;

Longitudinal refracted angle = 90° at the interface;

v = longitudinal sound velocity in Rexolite;

u = Rayleigh wave velocity (2900 m/s).

Substituting in equation (1),

$$\frac{\sin \theta}{\sin 90} = \frac{2362.2}{2900}$$

$$\theta = \text{asin} (2362.2/2900 \sin 90) = 54.54^\circ$$

Hence, θ is beyond the first critical angle in steel (i.e. 27.5°). However, it is below the second critical angle (i.e. 57°). Therefore, only shear waves would be refracted into the unit under test.

Also, the refracted shear angle β in the specimen can be estimated as follows:

$$\frac{\sin 54.54}{\sin \beta} = \frac{2362.2}{2478}$$

$$\beta = \text{asin} \left(\frac{2478}{2362.2} \sin 54.54 \right) = 58.72^\circ$$

III. MATERIAL CHARACTERIZATION

The sound velocity of the specimen was measured using pulse-echo and through- transmission ultrasound. With this approach, the sound velocity of the specimen of known thickness can be found. Conversely, the sample thickness can be tested for material of known sound velocity especially in stress corrosion control [4].

As will be shown subsequently, the pulse-echo method gives more accurate results than the through transmission. Another advantage with the pulse-echo technique is that it is more amenable to ultrasonic testing because it requires only one scanning surface of the specimen [5].

For a steel block of thickness 20.3mm, the longitudinal sound velocity, using the two ultrasonic techniques, is compared as follows:

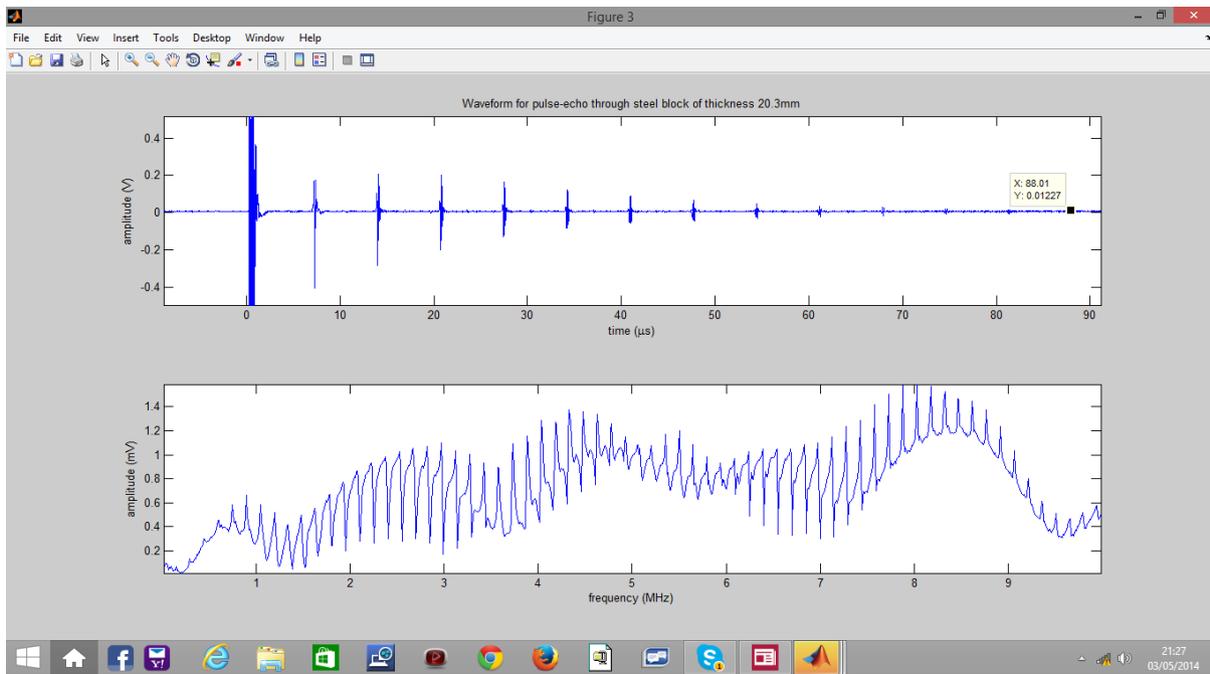


Fig. 2 : Waveform of pulse-echo ultrasound in the steel block

From Fig. 2, the time of flight is the time interval between the pickup and the first echo divided by 2.

$$\text{Hence, time of flight, } T_f = \frac{7.1-0.2}{2} = 3.45\mu\text{s}$$

$$\begin{aligned} \text{Sound velocity in the steel} &= \frac{\text{Thickness of steel block}}{\text{Time of flight}} \\ &= \frac{20300 \text{ mm}}{3.45\text{s}} \\ &= 5,884.1\text{m/s} \end{aligned}$$

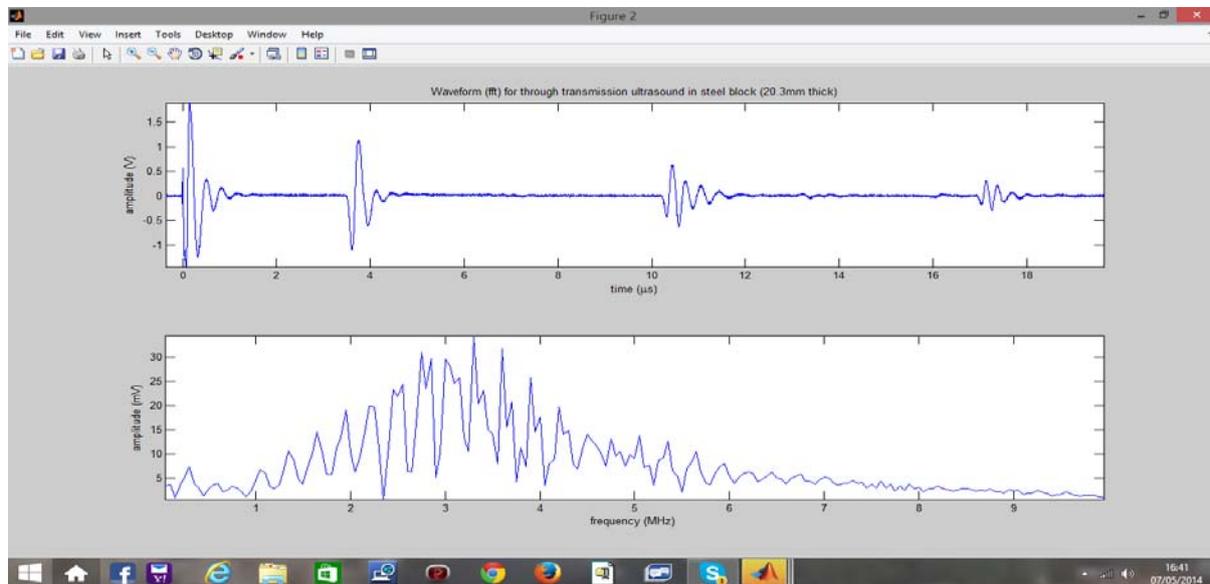


Fig. 3: Waveform of through transmission ultrasound in the steel block

From Fig. 3, the time of flight is the interval between the pickup and the corresponding point on the first reflection.

$$\text{Sound velocity in the specimen} = \frac{\text{Thickness of steel block}}{\text{Time of flight}}$$

$$\begin{aligned} \text{From the graph, time of flight, } T_f &= 3.48\mu\text{s} - 0.014\mu\text{s} \\ &= 3.47\mu\text{s} \end{aligned}$$

$$\begin{aligned} \text{Hence, longitudinal sound velocity in the steel} &= \frac{0.0203\text{ m}}{3.47e-6(\text{s})} \\ &= 5,850\text{m/s} \end{aligned}$$

IV. SIMULATION OF SOUND WAVE IN THE STEEL SAMPLE

A 10mm diameter piston transducer with a centre frequency of 1MHz was used to transmit

a) Zero Degree Interface Angle

ultrasound into the specimen through a Rexolite wedge of 0 and 20° interface angles. The directivity patterns for both longitudinal and shear waves were plotted in matlab using the directivity function.

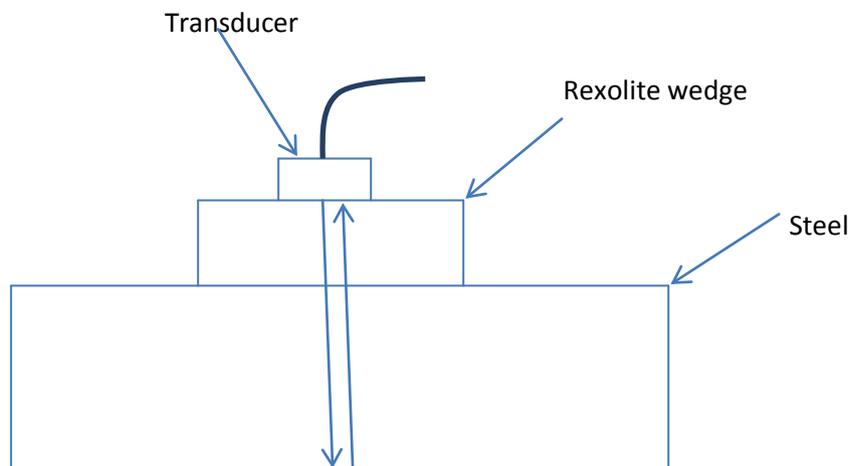


Fig. 4: Ray propagation from Rexolite into steel at 0° Interface angle



The directivity patterns for Fig. 4 are displayed as follows:

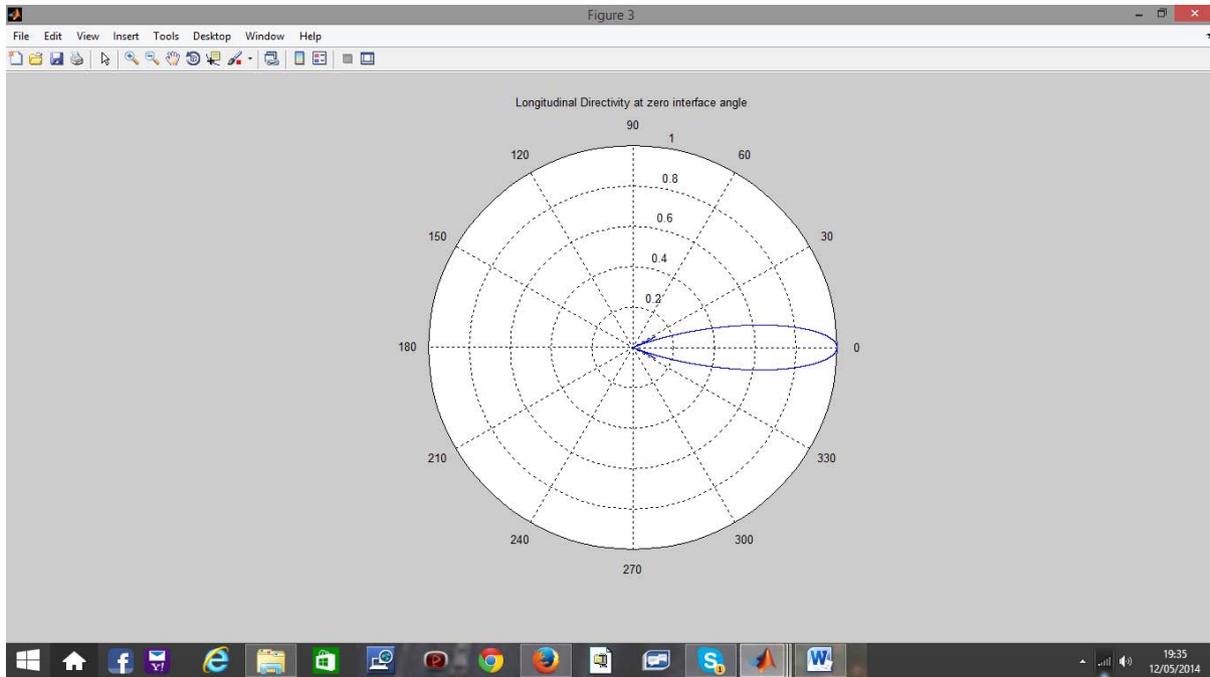


Fig. 5 : Directivity pattern of longitudinal wave in Steel/Rexolite at 0° interface angle

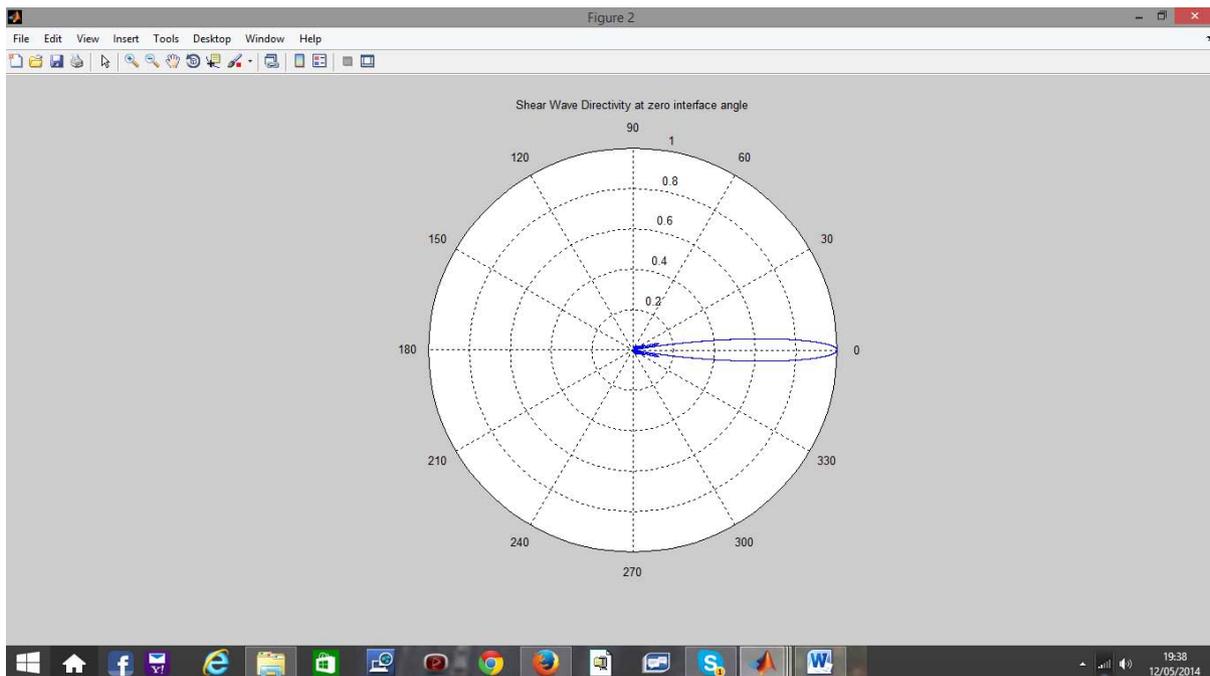


Fig. 6 : Directivity pattern of shear wave in Steel/Rexolite at 0° interface angle

b) Twenty Degree Interface Angle

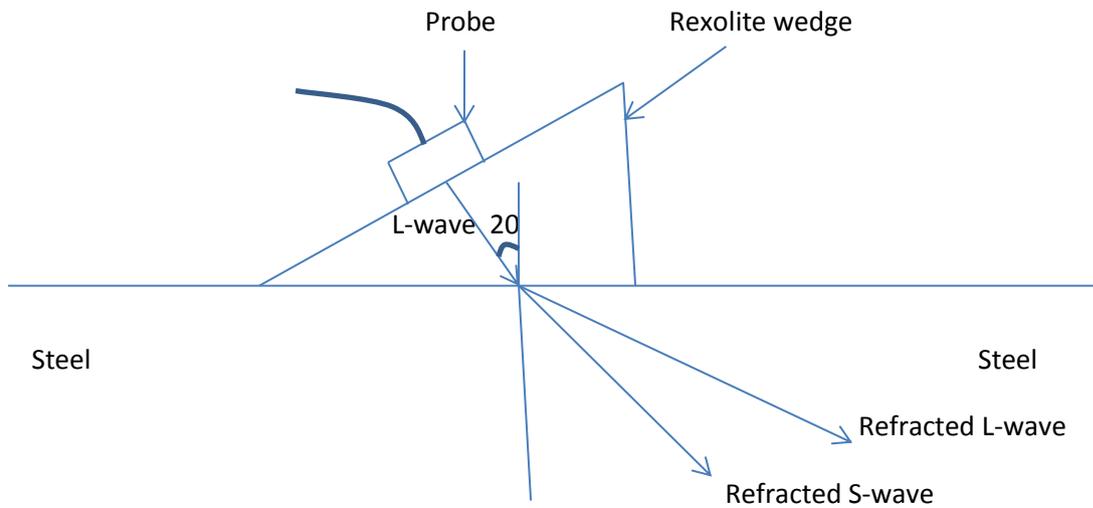


Fig. 7: Ray propagation through 20° Rexolite wedge into Steel

The directivity patterns for Fig. 7 are shown below.

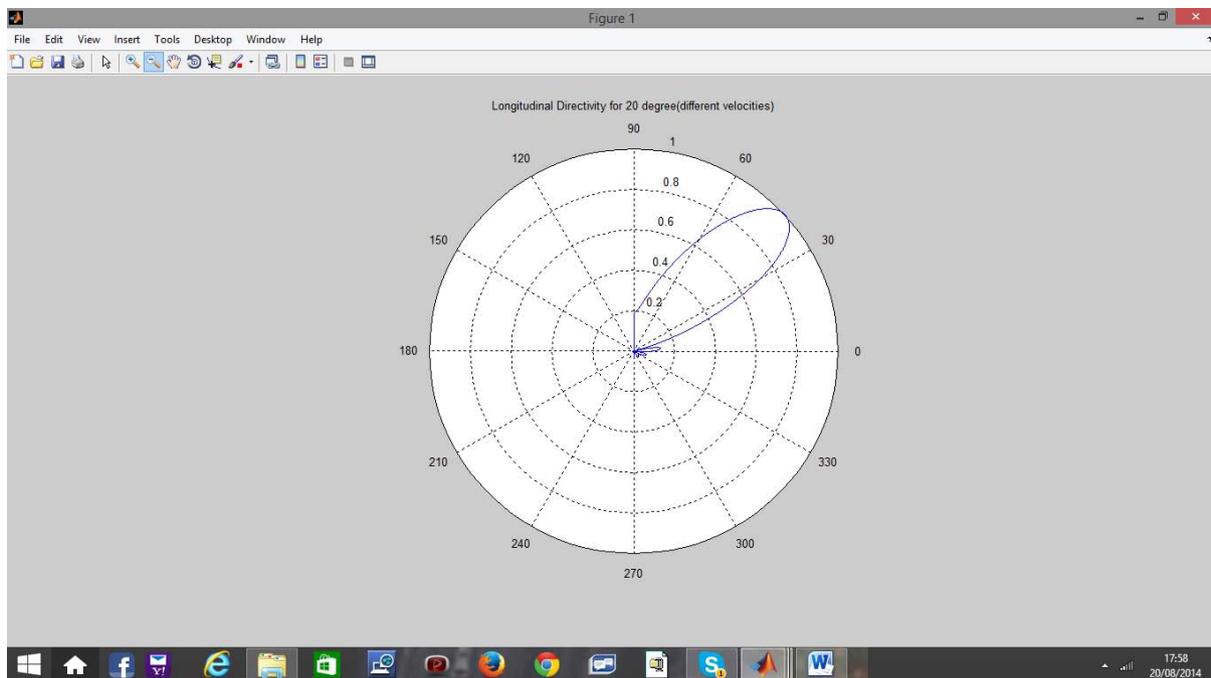


Fig. 8: Directivity pattern of longitudinal wave in Steel/Rexolite at 20° interface angle, 1MHz

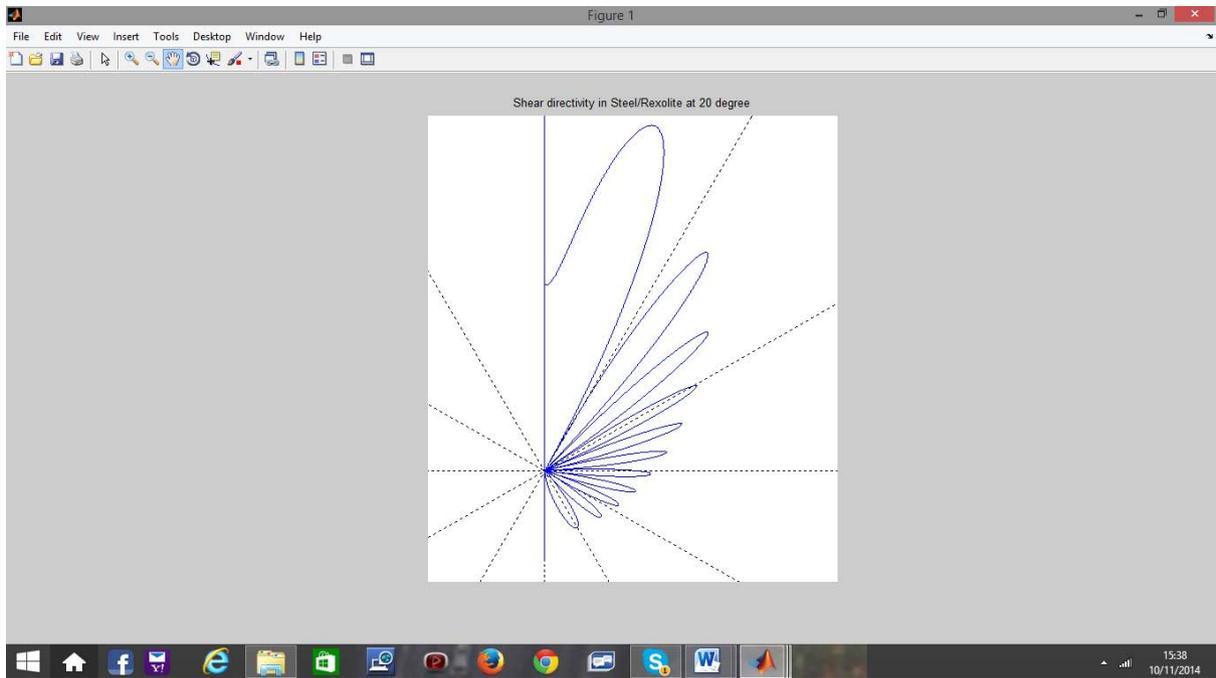


Fig. 9 : Directivity pattern of shear wave in Steel/Rexolite at 20° interface angle, 1MHz

c) Reducing The Centre Frequency of the Transducer From 1mhz To 0.5mhz.

The simulation patterns are illustrated below.

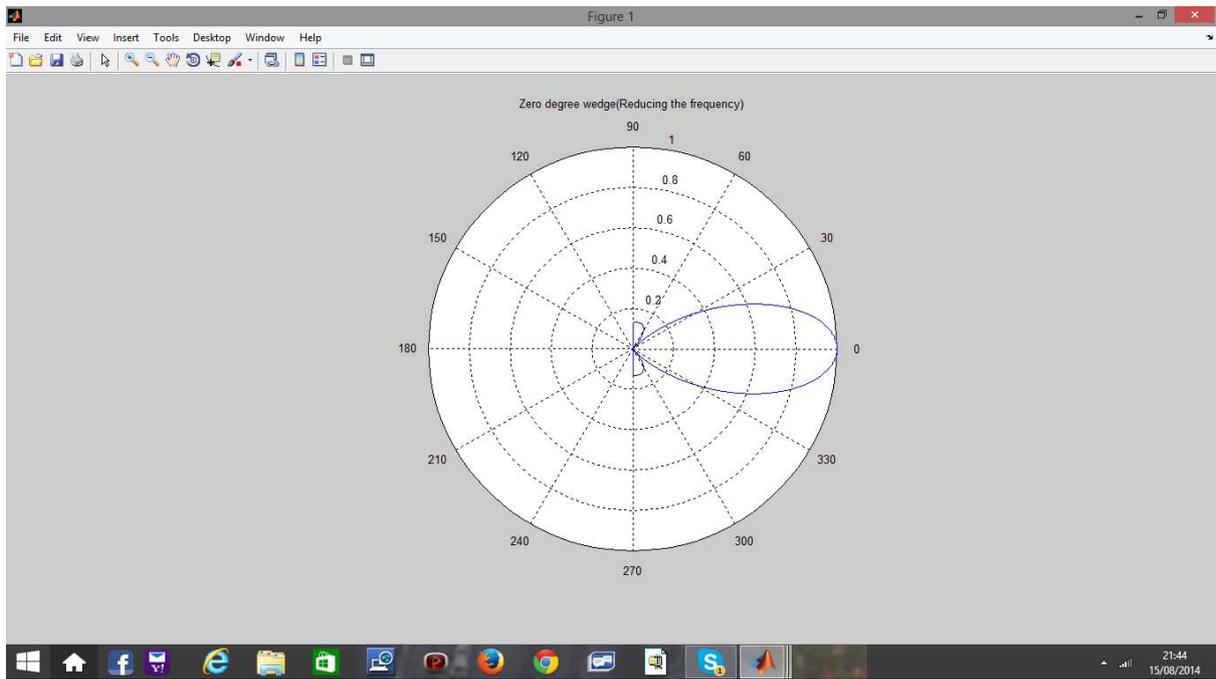


Fig.10 : Directivity pattern of longitudinal wave in Steel/Rexolite at 0° angle and 0.5MHz

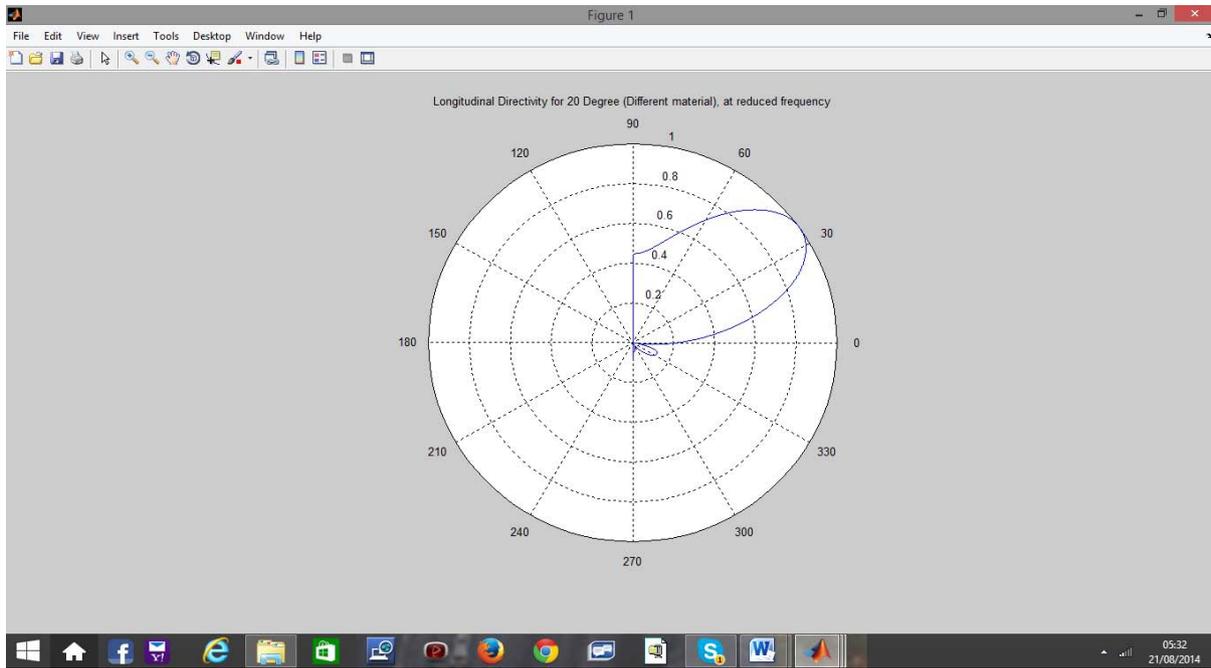


Fig.11 : Directivity pattern of longitudinal wave in Steel/Rexolite at 20° angle and 0.5MHz

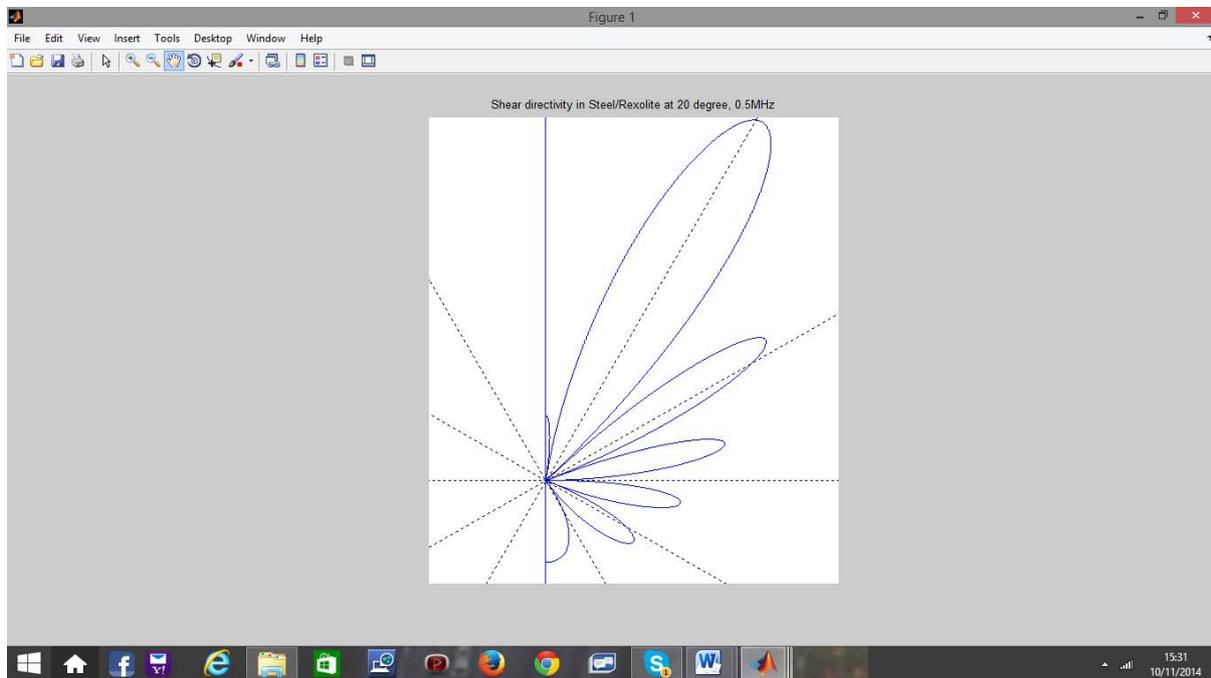


Fig.12 : Directivity pattern of shear wave in Steel/Rexolite at 20° angle and 0.5MHz

V. DISCUSSION OF RESULTS

For the experiments on material characterization (see section 3), the ideal longitudinal sound velocity of steel is 5890m/s. This literature value could not be attained due to errors of instruments, operator and temperature variations. Also, the pulse-echo method gave more accurate measurement than the through transmission technique as seen from the small deviation when compared to the literature value.

From Fig. 5, the longitudinal directivity is less directional and less focussed than the shear directivity in Fig. 6. Moreover, both patterns are symmetrical, indicating absence of mode conversion. Directional ultrasonic signals are sensitive to small flaws and flaws parallel to the wave direction [6]. Hence, this informs why shear waves are mostly preferred in ultrasonic testing.

In Fig. 8, the longitudinal directivity pattern for the 20 degree wedge is asymmetrical and has a pronounced main lobe while in Fig. 9; the pattern is more distorted with many side lobes.

By reducing the centre frequency of the finite transducer, the following observations are made:

- a) The longitudinal directivity patterns for the 0 and 20° interface wedges are more omnidirectional compared to the previous patterns in Figs. 5 and 8.
- b) In Fig. 12, the shear directivity pattern for the 20 degree wedge is larger with fewer side lobes compared to that in Fig. 9.

Finally, it can be seen that the asymmetrical and distorted form of the patterns for the 20 degree angle wedge is due to mode conversion at the material interface. The directivity patterns for the straight wedge are symmetrical and on axis due the absence of mode conversion. Mode conversion is effected when the incident angle is not perpendicular to the interface in the presence of impedance mismatch [7].

VI. CONCLUSION

Appropriate wedge design and simulation will greatly facilitate the modeling of ultrasonic wedges. The approach will determine the critical incident angle which is one of the input parameters to numerical modeling.

Also, the analytical simulation done in this study can provide a reasonable picture of the numerical approach.

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Kinetic Induktance Charges and its Role in Classical Electrodynamics

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Keywords: *electrodynamics; maxwell equation; kinetic inductance of charges; dielectric constant; magnetic permeability.*

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Kinetic Inductance Charges and its Role in Classical Electrodynamics

F. F. Mende

Abstract- The dielectric and magnetic constant of material media are the parameters, which are used during writing Maxwell equations. However, there is still one not less important material parameter, namely a kinetic inductance of charges, which has not less important role, than the parameters indicated. Unfortunately, importance and fundamentality of this parameter in the works on electrodynamics, until now, is not noted, and kinetic inductance is present in all equations of electrodynamics implicitly. This work is dedicated to the examination of the role of the kinetic inductance of charges in the electrodynamics of material media and to the restoration of its rights as the fundamental parameter, on the importance that not less meant than dielectric and magnetic constant.

Keywords: *electrodynamics; maxwell equation; kinetic inductance of charges; dielectric constant; magnetic permeability.*

I. INTRODUCTION

In the existing scientific literature occurs only the irregular references about the kinetic the inductance of charges [1-3]. the most in detail physical essence of this parameter and its place in the description of electrodynamic processes in the conductors is examined in work [4]. In this work is introduced also the concept of surface kinetic and field inductance, which earlier was not introduced:

$$L_K = \frac{1}{\omega |\vec{H}_T(0)|^2} \text{Im} \int_0^\infty \vec{j}^* \vec{E} dz,$$

$$L_H = \frac{1}{|\vec{H}_T(0)|^2} \int_0^\infty |\vec{H}_T|^2 dz,$$

Where L_K and L_H - surface kinetic and field inductance, \vec{E} - the tension of electric field, \vec{j}^* - the complexly conjugate value of current density, \vec{H}_T - tension of magnetic field; $\vec{H}_T(0)$ - the value of the tension of magnetic field on the surface; ω - frequency of the applied field. These relationships are valid for the case of the arbitrary connection between the current and

the field in the metals and the superconductors, and they reveal the physical essence of surface kinetic and field inductance.

a) *Plasmo-like media*

The equation of motion of electron takes the following form:

$$m \frac{d\vec{v}}{dt} = e\vec{E} \quad (1.1)$$

where m - mass electron, e - electron charge, \vec{E} - tension of electric field, \vec{v} - speed of the motion of charge. In the work [2] it is shown that this equation can be used also for describing the electron motion in the hot plasma.

Using an expression for the current density

$$\vec{j} = ne\vec{v}, \quad (1.2)$$

from (1.1) we obtain the current density of the conductivity of the free electrons

$$\vec{j}_L = \frac{ne^2}{m} \int \vec{E} dt \quad (1.3)$$

in relationships (2.2) and (2.3) the value n represents the specific density of charges. After introducing the designation

$$L_k = \frac{m}{ne^2} \quad (1.4)$$

we find

$$\vec{j}_L = \frac{1}{L_k} \int \vec{E} dt \quad (1.5)$$

In this case the value of L_k presents the specific kinetic inductance of charge carriers [4-7]. Its existence connected with the fact that charge, having a mass, possesses inertia properties.

Pour on relationship (1.5) it will be written down for the case of harmonics:

$$\vec{j}_L = -\frac{1}{\omega L_k} \vec{E}_0 \cos \omega t \quad (1.6)$$

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For the mathematical description of electrodynamic processes the trigonometric functions will be here and throughout, instead of the complex quantities, used so that would be well visible the phase relationships between the vectors, which represent electric fields and current densities. From relationship (1.5) and (1.6) is evident that presents inductive current, since its phase is late with respect to the tension of electric field to the angle.

During the presence of summed current it is necessary to consider bias current

$$\vec{j}_\varepsilon = \varepsilon_0 \frac{\partial \vec{E}}{\partial t} = \varepsilon_0 \vec{E}_0 \cos \omega t$$

Is evident that this current bears capacitive nature, since its phase anticipates the phase of the tension of electrical to the angle $\frac{\pi}{2}$. Thus, summary current density will compose

$$\vec{j}_\Sigma = \varepsilon_0 \frac{\partial \vec{E}}{\partial t} + \frac{1}{L_k} \int \vec{E} dt$$

or

$$\vec{j}_\Sigma = \left(\omega \varepsilon_0 - \frac{1}{\omega L_k} \right) \vec{E}_0 \cos \omega t \quad (1.7)$$

Introducing the plasma frequency

$\omega_0 = \sqrt{\frac{1}{L_k \varepsilon_0}}$, relationship (1.7) it is possible to rewrite

$$\vec{j}_\Sigma = \omega \varepsilon_0 \left(1 - \frac{\omega_0^2}{\omega^2} \right) \vec{E}_0 \cos \omega t \quad (1.8)$$

If in the conductor are ohmic losses, then total current density determines the relationship

$$\vec{j}_\Sigma = \sigma \vec{E} + \varepsilon_0 \frac{\partial \vec{E}}{\partial t} + \frac{1}{L_k} \int \vec{E} dt$$

where σ - conductivity of metal.

b) Dielectrics

In the existing literature there are no indications that the kinetic inductance of charge carriers plays some

role in the electrodynamic processes in the dielectrics. However, this not thus [7]. This parameter in the electrodynamics of dielectrics plays not less important role, than in the electrodynamics of conductors.

Let us examine the simplest case, when oscillating processes in atoms or molecules of dielectric obey the law of mechanical oscillator.

$$\left(\frac{\beta}{m} - \omega^2 \right) \vec{r}_m = \frac{e}{m} \vec{E}, \quad (2.1)$$

Where \vec{r}_m - deviation of charges from the position of equilibrium, β - coefficient of elasticity, which characterizes the elastic electrical binding forces of charges in the atoms and the molecules. Introducing the resonance frequency of the bound charges of

$$\omega_0 = \frac{\beta}{m}$$

we obtain from (2.1):

$$\vec{r}_m = -\frac{e \vec{E}}{m(\omega^2 - \omega_0^2)}. \quad (2.2)$$

is evident that in relationship (2.2) as the parameter is present the natural vibration frequency, into which enters the mass of charge. This speaks, that the inertia properties of the being varied charges will influence oscillating processes in the atoms and the molecules.

Since the general current density on Wednesday consists of the bias current and conduction current

$$\text{rot} \vec{H} = \vec{j}_\Sigma = \varepsilon_0 \frac{\partial \vec{E}}{\partial t} + ne \vec{v}$$

that, finding the speed of charge carriers in the dielectric as the derivative of their displacement through the coordinate

$$\vec{v} = \frac{\partial \vec{r}_m}{\partial t} = -\frac{e}{m(\omega^2 - \omega_0^2)} \frac{\partial \vec{E}}{\partial t}$$

from relationship (2.2) we find

$$\text{rot} \vec{H} = \vec{j}_\Sigma = \varepsilon_0 \frac{\partial \vec{E}}{\partial t} - \frac{1}{L_{kd}(\omega^2 - \omega_0^2)} \frac{\partial \vec{E}}{\partial t} \quad (2.3)$$

But the value

$$L_{kd} = \frac{m}{ne^2}$$

presents the kinetic inductance of the charges, entering the constitution of atom or molecules of dielectrics, when to consider charges free. Then relationship (2.3) can be rewritten

$$\text{rot}\vec{H} = \vec{j}_{\Sigma} = \varepsilon_0 \left(1 - \frac{1}{\varepsilon_0 L_{kd} (\omega^2 - \omega_0^2)} \right) \frac{\partial \vec{E}}{\partial t} \quad (2.4)$$

But, since the value

$$\frac{1}{\varepsilon_0 L_{kd}} = \omega_{pd}^2$$

represents the plasma frequency of charges in atoms and molecules of dielectric, if we consider these charges free, then relationship (2.4) takes the form:

$$\text{rot}\vec{H} = \vec{j}_{\Sigma} = \varepsilon_0 \left(1 - \frac{\omega_{pd}^2}{(\omega^2 - \omega_0^2)} \right) \frac{\partial \vec{E}}{\partial t} \quad (2.5)$$

Let us examine two limiting cases:

- If $\omega = \omega_0$, then from (2.5) we obtain

$$\text{rot}\vec{H} = \vec{j}_{\Sigma} = \varepsilon_0 \left(1 + \frac{\omega_{pd}^2}{\omega_0^2} \right) \frac{\partial \vec{E}}{\partial t} \quad (2.6)$$

In this case the coefficient, confronting the derivative, does not depend on frequency, and it presents the static dielectric constant of dielectric. As we see, it depends on the natural frequency of oscillation of atoms or molecules and on plasma frequency. This result is intelligible. Frequency in this case proves to be such small that the inertia properties of charges it is possible not to consider, and bracketed expression in the right side of relationship (1.7) presents the static dielectric constant of dielectric. Hence immediately we have a prescription for creating the dielectrics with the high dielectric constant. In order to reach this, should be in the assigned volume of space packed a maximum quantity of molecules with maximally soft connections between the charges inside molecule itself.

- The case, when $\omega \gg \omega_0$, is exponential. Then

$$\text{rot}\vec{H} = \vec{j}_{\Sigma} = \varepsilon_0 \left(1 - \frac{\omega_{pd}^2}{\omega^2} \right) \frac{\partial \vec{E}}{\partial t}$$

and dielectric is converted in the plasma. The obtained relationship coincides with the case of plasma (1.8).

Now it is possible to examine the question of why dielectric prism decomposes polychromatic light into monochromatic components or why rainbow is formed. So that this phenomenon would occur, it is necessary to have the frequency dispersion of the phase speed of electromagnetic waves in the medium in question. If we to relationship (2.5) add the first equation of Maxwell, then we will obtain:

$$\begin{aligned} \text{rot}\vec{E} &= -\mu_0 \frac{\partial \vec{H}}{\partial t} \\ \text{rot}\vec{H} &= \varepsilon_0 \left(1 - \frac{\omega_{pd}^2}{(\omega^2 - \omega_0^2)} \right) \frac{\partial \vec{E}}{\partial t} \end{aligned} \quad (2.7)$$

That we will obtain the wave equation

$$\nabla^2 \vec{E} = \mu_0 \varepsilon_0 \left(1 - \frac{\omega_{pd}^2}{\omega^2 - \omega_0^2} \right) \frac{\partial^2 \vec{E}}{\partial t^2}$$

If one considers that

$$\mu_0 \varepsilon_0 = \frac{1}{c^2}$$

where C - speed of light, then no longer will remain doubts about the fact that with the propagation of electromagnetic waves in the dielectrics the frequency dispersion of phase speed will be observed. In the formation of this dispersion it will participate immediately three, which do not depend on the frequency, physical quantities: the self-resonant frequency of atoms themselves or molecules, the plasma frequency of charges, if we consider it their free, and the dielectric constant of vacuum.

II. CONCLUSION

This examination showed that this parameter as the kinetic inductance of charges characterizes electromagnetic processes in the conductors and the dielectrics and has the same fundamental value as the dielectric and magnetic constant of these media. Unfortunately, this important circumstance is not noted not only in the existing scientific literature, but also in the works of Maxwell.

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A Thesis Report on the Application of Preventive Replacement Strategy on Machines in Perspective of Cement Industry in Bangladesh

By M. M. Israfil Shahin Seddiqe, Avizit Basak, Md. Rifaul Islam
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Abstract- The operation of a particular component in deteriorating condition will lead to a high machine downtime. This is due to the failure of component at unexpected time. As a result it will increase cost of maintenance and production lost. One of the solutions to this matter is to use Preventive Maintenance (PM). In industries, Preventive Maintenance (PM) is not a new practice to minimize the sudden breakdown of production machine. PM will be performed at predetermine intervals to provide a balance between failure cost and component utilization (aging). Therefore, the objective of this paper is to introduce the preventive maintenance strategy for determining an optimal replacement time for component that deteriorates over time. In this thesis we consider a particular type of machine (Gear Motor 7.5 KW.) from Holcim Bangladesh Ltd. where machines are subject to maintenance. To maximize the benefit from operating the machine two replacement models are used. Among them, one model is used to determine an optimal replacement policy which tells us, when equipment reaches a particular age, whether or not it should be replaced or continue to be operated to minimize the total operating cost. Another model is used to determine the optimal interval between the preventive replacements to minimize the total cost and to operate the machine to the time which is determined by first model. We have determined to find the preventive replacement cost and also the maximum time at which we can use the machine without replacing it. Some time it is more economical to replace the machine rather than maintenance it. So it is most important to find out the age at which the replacement will be most economical to replace rather than maintenance.

Keywords: preventive replacement, mitigate degradation, etc.

GJRE-J Classification : FOR Code: 091599



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A Thesis Report on the Application of Preventive Replacement Strategy on Machines in Perspective of Cement Industry in Bangladesh

M. M. Israfil Shahin Seddiq^α, Avizit Basak^σ, Md. Rifaul Islam^ρ & Md. Omar Faruk Akanda^ω

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Keywords: preventive replacement, mitigate degradation, etc.

I. INTRODUCTION

What is maintenance and why is it performed? Past and current maintenance practices in both the private and government sectors would

imply that maintenance is the actions associated with equipment repair after it is broken. The dictionary defines maintenance as follows: “the work of keeping something in proper condition; upkeep.” This would imply that maintenance should be actions taken to prevent a device or component from failing or to repair normal equipment degradation experienced with the operation of the device to keep it in proper working order. Unfortunately, data obtained in many studies over the past decade indicates that most private and government facilities do not expend the necessary resources to maintain equipment in proper working order. Rather, they wait for equipment failure to occur and then take whatever actions are necessary to repair or replace the equipment. Nothing lasts forever and all equipment has associated with it some predefined life expectancy or operational life.

The need for maintenance is predicated on actual or impending failure – ideally, maintenance is performed to keep equipment and systems running efficiently for at least design life of the component(s). As such, the practical operation of a component is time-based function. If one were to graph the failure rate a component population versus time, it is likely the graph would take the “bathtub” shape shown in Figure 1.1.1. In the figure the Y axis represents the failure rate and the X axis is time. From its shape, the curve can be divided into three distinct: infant mortality, useful life, and wear-out periods.

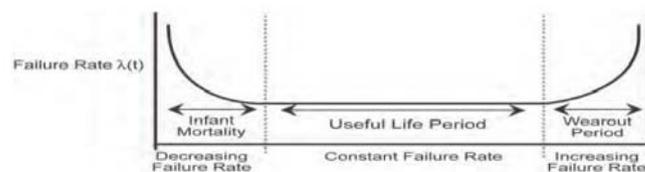


Figure 1.1.1 : Component failure rate over time for component population

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a) *Types of maintenance*

The maintenance is mainly two types and they are planned and unplanned maintenances. And these two types of maintenance are divided into some other types of maintenance.

Planned maintenance: It is an organized maintenance work carried out as per recorded procedures having control.

Breakdown maintenance: It is an emergency based policy in which the plant or equipment is operated until it fails and then it is brought back into running condition by repair. The maintenance staff locate any mechanical, electrical or any other fault to correct it immediately.

Corrective maintenance: It is a maintenance task performed to identify, isolate, and rectify a fault so that the failed equipment, machine, or system can be restored to an operational condition within the tolerances or limits established for in-service operations.

Routine maintenance: It refers to maintenance work that is normally planned for, and performed on a routine basis. Most of the time, routine maintenance can be, and is frequently, carried out without locking out a machine. It involves minor jobs such as cleaning, lubrication, inspection and minor adjustment. It needs very little investment in time and money.

Preventive maintenance: Actions performed on a time- or machine-run-based schedule that detect, preclude, or mitigate degradation of a component or system with the aim of sustaining.

b) *Benefits of Maintenance*

- Increase functional reliability of production facilities.
- Enables product and service quality to be achieved through correctly adjusted, serviced and operated equipment.
- Maximize the useful life of the equipment.
- Minimize the total production or operating costs directly attributed to equipment service and repair.
- Minimize the frequency of interruptions to production by reducing breakdowns.
- Maximize the production capacity from the given equipment resources.
- Enhance the safety of manpower.

II. PROBLEM STATEMENT

Holcim Bangladesh Ltd. is one of the leading cement manufacturing companies in Bangladesh. They have a number of heavy machines these machines are subject to maintenance according to traditional approach. The used traditional maintenance is actually scheduled maintenance and due to this practice a lot of problem occurs. Loss of production, repair and replacement cost, low productivity, long lead time and low reliability of plant machineries result from this type of maintenance.

Due to production loss productivity decreases and which results in the decrease of profitability. Repair/Replacement cost is required to restore the equipment in functioning condition. Since equipments are subject to breakdown then the reliability of the equipments to remain in functioning condition is low. Lower reliability and loss of production lead to take longer lead time of delivery. Moreover bottlenecking and more work in process inventory occur due to this practice.

In Holcim Bangladesh Ltd. the total activity is divided into three stages and they are the unloading system, cement production system and packing and delivery system. In these three stages there are many machines required. The machines in these three systems are Hydraulic crane, Belt conveyor, pay loader, motor 1000kw and packer machines. Most of the machines are consisted of different number of Gear motor 7.5kw. All the failures happen in the industry are more often failure of this motor. So here we have chosen to this motor in our calculation.

Maintenance practice is required to overcome above problems which removes loss of production, increase reliability of the equipment, decreases repair/replacement costs and leads the company towards success. Planned maintenance practice is one of those practices which can overcome above problems.

a) *Objectives*

The objectives of this paper is

- To investigate the economic advantages in implementing appropriate replacement process of equipment with physical impairment,
- To find out an optimal preventive maintenance interval based on the cost.

III. THEORETICAL BACKGROUND

a) *Optimal replacement policy for equipment whose operating cost increases with use (finite time horizon). (A.K.S JARDINE)*

i. *Construction of model*

- I = age of the equipment (in years) since last replaced with n periods of time to go until the end of production plan.
- J = age of the equipment (in years) since last replaced with $(n-1)$ periods of time to go to the end of production plan.
- $C(a)$ = operating cost (in Taka) for one period when the equipment is of age 'a' since last replaced at the start of the operating period.
- C_r = cost of replacement (in Taka).
- $C(I,J)$ = total cost (in Taka) of starting with the equipment of age I at the start of a period & having the equipment of age J at the end of the year.

- $C_n(l)$ = total cost (in Taka) of operating & replacing the equipment over next n years having age l at the start = $[C(l,J) + f_{n-1}(J)]$
 - C_f is the cost of a failure replacement .
 - $f(t)$ is the probability density function of the equipment's failure times.
 - The replacement policy is to perform preventive replacements at constant intervals of length t_p , irrespective of the age of the equipment, and failure replacements occurs as many times as required in interval $(0, t_p)$.
 - The objective is to determine the optimal interval between preventive replacements to minimize the total expected replacement cost per unit time.
- $f_n(l)$ = minimum value of $C_n(l) = \min_j [C(l,J) + f_{n-1}(J)]$
- Where, $f_{n-1}(J)$ = cost of best decision taken over the remaining $(n-1)$ years.
- b) *Optimal interval between preventive replacements of equipment subject to breakdown. (A.K.S JARDINE)*
- ii. *Construction of model*
- C_p is the cost of a preventive replacement.

The total expected cost per unit time, for preventive replacement at time t_p , denoted by $C(t_p)$ is

$$C(t_p) = \frac{\text{Total expected cost in interval } (0, t_p)}{\text{Length of interval}}$$

Total expected cost = cost of a preventive replacement in interval $(0, t_p)$ + Expected cost of failure replacement

$$= C_p + C_f H(t_p)$$

Where $H(t_p)$ is the expected number of failures in interval $(0, t_p)$.

Length of interval = t_p

Therefore

$$C(t_p) = \frac{C_p + C_f H(t_p)}{t_p}$$

This is a model of the problem relating replacement interval t_p to total cost $C(t_p)$.

c) *Determination of $H(t_p)$*

There is a process by which the expected number of failures $H(t_p)$ in an interval of length t_p can be obtained. The expected number of failures $H(t_p)$ can be determined by discrete method.

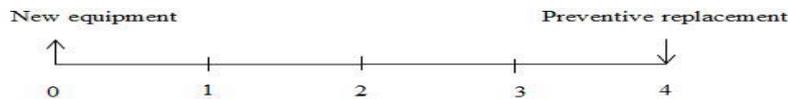


Figure : Illustration of the expected no. of failures in interval (0,4)

In general

$$H(T) = \sum_{t=0}^{T-1} [1 + H(T - i - 1)] \int_i^{i+1} f(t) dt ; T \geq 1$$

IV. DATA ANALYSIS AND CALCULATION

Table: Break down and repair reports

Machine name	No of failure (1year)	Total repair time (min/hours)(in 1 yr)	Total failure cost (1 year)	Data collection time
Gear Motor (7.5KW)	02	45 hrs	20000	(15/01/2011-15/01/2012)
Hydraulic Crane	03	24 hrs	50-100K	(15/01/2011-15/01/2012)
Chain Conveyor	04	32 hrs	100-500K	(15/01/2011-15/01/2012)
Bucket Elevator	02	40 hrs	50-100K	(15/01/2011-15/02/2012)
Water Pump	02	20 hrs	10-20K	(15/01/2011-15/01/2012)

From the above chart
For Gear Motor (7.5KW):

a) *Optimal replacement policy for equipment whose operating cost increases with use (finite time horizon)*

The objective of this policy is to perform replacements in such a way that the total cost of operating & replacing the motor over next 3 years is minimized having age $l=1$ years at the start.

Under this policy the relevant data are considered as follows-

Age of motor since last replaced, a in years	0	1	2	3
Operating cost* for one period C(a) in Tk.	4, 37,670	4, 92,720	5, 47,200	6, 02,160

Replacement cost, $C_r = 1, 50,000$ Tk.

b) *Operating cost calculation*

Number of motor = 100

Number of operator = 2

Number of shift = 2

Salary of operator per month = 12000 Tk.

Total monthly salary = $12000 \times 2 \times 2 = 48000$ Tk.

1st year, operator salary per machine = $48000/100 = 480$ Tk.

Operator salary per machine per year = $480 \times 12 = 5760$ Tk.

Similarly,

2nd year, Operator salary per machine per year =

$$\frac{14000 \times 4 \times 12}{100} = 6720 \text{ Tk.}$$

3rd year, Operator salary per machine per year =

$$\frac{15000 \times 4 \times 12}{100} = 7200 \text{ Tk.}$$

4th year, Operator salary per machine per year =

$$\frac{17000 \times 4 \times 12}{100} = 8160 \text{ Tk.}$$

Power of a Gear motor, $P = 7.5$ kw.

Per unit rate of electricity = 8 Tk. (for year 1)

Working days = 300 per year

Operating hours per day, $t = 24$ h

Thus total kWh,

$$W = Pt = 7.5 \times 24 = 180 \text{ kWh/day} = 180 \times 300 = 54000 \text{ kWh/year}$$

Since 1 unit = 1 kWh

The cost of operating a motor in a year = $54000 \times 8 = 432000$ Tk. per year

This is taken to be the electricity bill for the first year of operation.

Hence, total operating cost:

$$\text{For First year, operating cost} = 432000 + 5760 = 437670 \text{ Tk.}$$

Similarly,

Taking different values of the per unit rate for 2nd, 3rd & 4th year i.e. 9 Tk., 10Tk. & 11 Tk. respectively, the total cost can be calculated as above-

$$\text{For second year, operating cost} = (54000 \times 9) + 6720 = 492720 \text{ Tk.}$$

$$\text{For third year, operating cost} = (54000 \times 10) + 7200 = 547200 \text{ Tk.}$$

$$\text{For fourth year, operating cost} = (54000 \times 11) + 8160 = 602160 \text{ Tk.}$$

c) *Calculation*

Cost matrix

J \ I	0	1	2	3
0	∞	4,37,670	∞	∞
1	∞	5,87,670	4,92,720	∞
2	∞	5,87,670	∞	5,47,200
3	∞	5,87,670	∞	6,02,160

For 0 year to go,

$$f_0(l) = 0 \text{ for all possible values of } l.$$

For 1 year to go,

$$f_1(l) = \min_j [C(l,J) + f_0(J)] = \min_j [C(l,J)] \text{ since } f_0(J) = 0$$

Now when $l = 0$,

$$f_1(0) = \min_j [C(0,J)] = \min \begin{bmatrix} C(0,0) \\ C(0,1) \\ C(0,2) \\ C(0,3) \end{bmatrix} = \min \begin{bmatrix} \infty \\ 4,37,670 \\ \infty \\ \infty \end{bmatrix} \quad \leftarrow \text{ i.e. continue}$$

$$f_1(1) = \min_j [C(1,J)] = \min \begin{bmatrix} C(1,0) \\ C(1,1) \\ C(1,2) \\ C(1,3) \end{bmatrix} = \min \begin{bmatrix} \infty \\ 5,87,670 \\ 4,92,720 \\ \infty \end{bmatrix} \quad \leftarrow \text{ i.e. continue}$$

Similarly for I = 2, 3 the values of f₁(2), f₁(3) can be determined & can be computed and shown in a table as below-

Table 1 : For 1 year to go

I (year)	0	1	2	3
J (year)	1	2	3	1
Action to take at start of period	C	C	C	R
f ₁ (I)	4,37,670	4,92,720	5,47,200	5,87,670

C = continue & R = replace

For 2 years to go,

$$f_2(I) = \min_j [C(I,J) + f_1(J)]$$

When I = 0,

$$f_2(0) = \min_j [C(0,J) + f_1(J)] = \min \begin{bmatrix} C(0,0) + f_1(0) \\ C(0,1) + f_1(1) \\ C(0,2) + f_1(2) \\ C(0,3) + f_1(3) \end{bmatrix} = \min \begin{bmatrix} \infty \\ 930390 \\ \infty \\ \infty \end{bmatrix} \quad \leftarrow \text{ i.e. continue}$$

In similar process as used for table 1, tables can be made for 2, 3 and 4 years to go-

Table 2 : For 2 years to go

I (year)	0	1	2	3
J (year)	1	2	1	1
Action to take at start of period	C	C	R	R
f ₂ (I)	9,30,390	10,39,920	10,80,390	10,80,390

C = continue & R = replace

Table 3 : For 3 years to go

I (year)	0	1	2	3
J (year)	1	2	1,3	3
Action to take at start of period	C	C	R & C	R
f ₃ (I)	13,68,060	15,73,110	16,27,590	16,27,590

C = continue & R = replace

The replacement policy can be summarized as below-

Periods to go (year)	3	2	1
Decision	Continue (Table 3)	Replace (Table 2)	Continue (Table 1)

d) *Optimal interval between preventive replacements of equipment subject to breakdown*

Labor cost per maintenance per personnel in a month = 16750 Tk.

Spare parts (Accessories) cost (Ball bearing / bearing Sleeve) in a year = 3000 Tk. (1500*2)

Spare parts (Accessories) cost (Ball bearing / bearing Sleeve) in a month = 250 Tk. (3000/12)

Production rate = 30 tons (3000kg) per hour = 60 bag (3000/50) per hour

Net income per bag = 100 Tk.

Failure maintenance time in a year = 45 hrs

Failure maintenance time in a month = 3.75 hrs (45/12)

Failure maintenance cost in a year = 20000 Tk.

Failure maintenance cost in a month = 1667 Tk. (20000/12)

i. C_p Calculation

Preventive maintenance cost = 17000 Tk. (16750+250)

$C_p = 17000$ Tk.

ii. C_f Calculation

Production loss in a month in unit = 225 bags per month (60*3.75)

Production loss in a month in taka = 22500 Tk. (225*100)

C_f = Preventive maintenance cost + Failure maintenance cost in a month + Production loss in a month in taka

$$= (17000+22500+1667) \text{ Tk.}$$

$$= 41167 \text{ Tk.}$$

μ and σ Calculation

Table : Failure analysis

No of failure	Failure occur at (month)
1 st	4
2 nd	10
3 rd	16
4 th	21

From the above chart if we calculate then we get,

$$\mu = 5.25 \text{ months}$$

$$\sigma = .96 \text{ month}$$

$$C(t_p) = \frac{C_p + C_f H(t_p)}{t_p}$$

For $t_p = 1$ month,

$$H(1) = [1 + H(0)] \int_0^1 f(t) dt = \int_0^1 f(t) dt = \varphi(-5.25) - \varphi(-4.43) = 0$$

$$C(1) = \frac{17000 + 41167 \times 0}{1} = 17000 \text{ Tk.}$$

For $t_p = 2$ months,

$$H(2) = [1 + H(1)] \int_0^1 f(t) dt + [1 + H(0)] \int_1^2 f(t) dt = [1 + 0] \times 0 + [\varphi(-3.49) - \varphi(-4.43)]$$

$$= 0 + [3.6 \times 10^{-4} - 0]$$

$$= 3.6 \times 10^{-4}$$

$$C(2) = \frac{17000 + 41167 \times 3.6 \times 10^{-4}}{2} = 8507 \text{ Tk.}$$

$$\text{For } t_p = 3 \text{ months, } H(3) = [1 + H(2)] \int_0^1 f(t) dt + [1 + H(1)] \int_1^2 f(t) dt + [1 + H(0)] \int_2^3 f(t) dt$$

$$= 1.00036 \times 0 + 1 \times 0.00036 + [\varphi(-2.34) - \varphi(-3.39)]$$

$$= .0036 + (9.55 \times 10^{-3} - 3.6 \times 10^{-4})$$

$$= .00955$$

$$C(3) = \frac{17000 + 41167 \times .00955}{3} = 5798 \text{ Tk.}$$

For $t_p = 4$ months,

$$H(4) = [1+H(3)] \int_0^1 f(t)dt + [1+H(2)] \int_1^2 f(t)dt + [1+H(1)] \int_2^3 f(t)dt + [1+H(0)] \int_3^4 f(t)dt$$

$$= 0 + 3.6 \times 10^{-4} + 0.00919 + [\varphi(-1.302) - \varphi(-2.34)]$$

$$= 0.09635$$

$$C(4) = \frac{17000 + 41167 \times 0.09635}{4} = 5241 \text{ Tk.}$$

For $t_p = 5$ months,

$$H(5) = [1+H(4)] \int_0^1 f(t)dt + [1+H(3)] \int_1^2 f(t)dt + [1+H(2)] \int_2^3 f(t)dt + [1+H(1)] \int_3^4 f(t)dt + [1+H(0)] \int_4^5 f(t)dt$$

$$= [1 + .00955] * 0.00036 + [1 + 0.00036] * 0.00919 + 0.09635 + [\varphi(-0.26) - \varphi(-1.302)]$$

$$= 0.4158$$

$$C(5) = \frac{17000 + 41167 \times 0.4158}{5} = 6823 \text{ Tk.}$$

The above results can be shown in a table as below-

t_p (months)	1	2	3	4	5
C(t_p) (Taka)	17000	8507	5798	5241	6823

V. LIMITATIONS

The thesis activity performed here is not out of limitations. The identified limitations are:

- ❖ The research is based on history data and not on real time data.
- ❖ The research is made up with failure data not failure modes.
- ❖ The mathematical formulations did not take into account the time it requires to perform preventive replacements because the model assumed that time to be very short, compared to the mean time between replacements.
- ❖ The calculated value of mean (μ) is not 100% accurate because of lack of the data of failure occurs.
- ❖ For these type of model application, it is necessary to know the characteristics of the machines over a long operating time i.e. performance of machine, number of failure occur during operating etc.
- ❖ The model of replacement decisions are developed by A. K. S. Jardine in separate conditions but this thing is neglected in this thesis.

VI. CONCLUSION

From the above model machine will be replaced after 2 years since the age of machine is 1. And the total

cost of replacement and operation to the three years would be 15, 73,110 Tk.

According to the above method the optimal preventive replacement is to perform at 4 months interval.

VII. RECOMMENDATIONS

The following recommendations are forwarded for the industry and related Bangladeshi industries that are executing maintenance work in their regular activity

- ❖ Replacement or repair cost, benefit from operating the equipment, labor cost etc. should be collected with high degree of accuracy.
- ❖ Statistical analysis of the data should be done frequently.
- ❖ Proper documentation of each activity should be kept.

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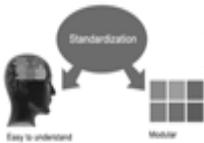
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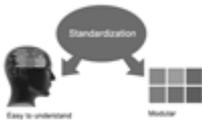


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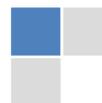
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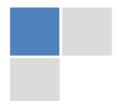
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- If you desire, you may place your figures and tables properly within the text of your results part.

Figures and tables

- If you put figures and tables at the end of the details, make certain that they are visibly distinguished from any attach appendix materials, such as raw facts
- Despite of position, each figure must be numbered one after the other and complete with subtitle
- In spite of position, each table must be titled, numbered one after the other and complete with heading
- All figure and table must be adequately complete that it could situate on its own, divide from text

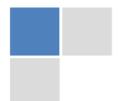
Discussion:

The Discussion is expected the trickiest segment to write and describe. A lot of papers submitted for journal are discarded based on problems with the Discussion. There is no head of state for how long a argument should be. Position your understanding of the outcome visibly to lead the reviewer through your conclusions, and then finish the paper with a summing up of the implication of the study. The purpose here is to offer an understanding of your results and hold up for all of your conclusions, using facts from your research and generally accepted information, if suitable. The implication of result should be visibly described. Infer your data in the conversation in suitable depth. This means that when you clarify an observable fact you must explain mechanisms that may account for the observation. If your results vary from your prospect, make clear why that may have happened. If your results agree, then explain the theory that the proof supported. It is never suitable to just state that the data approved with prospect, and let it drop at that.

- Make a decision if each premise is supported, discarded, or if you cannot make a conclusion with assurance. Do not just dismiss a study or part of a study as "uncertain."
- Research papers are not acknowledged if the work is imperfect. Draw what conclusions you can based upon the results that you have, and take care of the study as a finished work
- You may propose future guidelines, such as how the experiment might be personalized to accomplish a new idea.
- Give details all of your remarks as much as possible, focus on mechanisms.
- Make a decision if the tentative design sufficiently addressed the theory, and whether or not it was correctly restricted.
- Try to present substitute explanations if sensible alternatives be present.
- One research will not counter an overall question, so maintain the large picture in mind, where do you go next? The best studies unlock new avenues of study. What questions remain?
- Recommendations for detailed papers will offer supplementary suggestions.

Approach:

- When you refer to information, differentiate data generated by your own studies from available information
- Submit to work done by specific persons (including you) in past tense.
- Submit to generally acknowledged facts and main beliefs in present tense.



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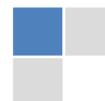
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<i>Methods and Procedures</i>	Clear and to the point with well arranged paragraph, precision and accuracy of facts and figures, well organized subheads	Difficult to comprehend with embarrassed text, too much explanation but completed	Incorrect and unorganized structure with hazy meaning
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<i>References</i>	Complete and correct format, well organized	Beside the point, Incomplete	Wrong format and structuring



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