

About the Presence of Irregular Precession Motions in a Symmetric

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Abstract

It is generally accepted that the only type of motion present in a symmetric Euler gyroscope (SEG) is regular precession. This paper proves that regular precession is not the only type of motion present, but corresponds only to the well-known initial coordinated Euler angles. At any other initial angles, motions that differ from regular precession occur. In the article, the problem is solved analytically in two stages: first, angular velocities of the gyroscope are determined using differential dynamic equations, at the second stage, as a result of integration of differential matrix kinematic and differential matrix Poisson equations (both with periodic coefficients), final relations about the SEG motion with arbitrary initial Euler angles are derived. Periodic coefficients are the SEG angular velocities that are found as a solution to the dynamic equations. From the obtained general formulas, special formulas of regular precession for particular coordinated initial Euler angles that coincide with the well-known ones are derived.

Index terms—

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I.

2 Annotation

It is generally accepted that the only type of motion present in a symmetric Euler gyroscope (SEG) is regular precession. This paper proves that regular precession is not the only type of motion present, but corresponds only to the well-known initial coordinated Euler angles. At any other initial angles, motions that differ from regular precession occur. In the article, the problem is solved analytically in two stages: first, angular velocities of the gyroscope are determined using differential dynamic equations, at the second stage, as a result of integration of differential matrix kinematic and differential matrix Poisson equations (both with periodic coefficients), final relations about the SEG motion with arbitrary initial Euler angles are derived. Periodic coefficients are the SEG angular velocities that are found as a solution to the dynamic equations. From the obtained general formulas, special formulas of regular precession for particular coordinated initial Euler angles that coincide with the well-known ones are derived. For other initial angles, formulas for irregular precession are obtained. In addition to the solutions for the Euler angles, solutions for the Euler-Krylov angles were found, which in some cases provide a more explicit geometric interpretation of motion. The analytical results are supported by mathematical modeling. In particular, certain conditions were found -the "strong impact" condition when irregular SEG precession for the Euler-Krylov angles occurs in the direction of the rotational pulse, and the sign of the angular velocity of the gyroscope proper rotation changes to the opposite. At the Euler angles, the motions of irregular precession during the "strong" and "weak" impact conditions are qualitatively identical. In relation to the case of regular precession under the "strong" impact conditions, the changes are significant: the angles of precession and nutation become oscillatory, and the angular velocity and the angle of proper rotation change their sign to the opposite.

3 a) Relevance

Modern gyroscopic technology has achieved the highest accuracy in measuring angular motion parameters of moving objects (MO) in the field of classical symmetric Euler gyroscopes with electrostatic suspension. In the US Gravity Probe experiment, the four axially symmetric Euler gyroscopes with electrostatic cryogenic suspension mounted on the astronomical Earth satellite had values of drift angular velocities of less than 10^{-11} angular deg/hr. This, together with the telescope readings, experimentally confirms the Einsteinian general theory of relativity (GTR) by detecting a gyro axis shift with the accuracy of 1% equal to 6.6 angular seconds per year, which is effectively predicted by the GTR [1,2]. It is noted that classical symmetric Euler gyroscopes (SEG) with electrostatic suspensions have drift angular velocities values of 10^{-5} angular deg/hr in terrestrial conditions, which is a better accuracy level than that of fiber optic (FOG) and laser (LG) gyroscopes, i.e. gyros based on new physical measurement principles in which drift angular velocities values are in the range of 10^{-4} - 10^{-3} angular deg/hr, respectively [2]. Considering the fact that rotary classical Euler gyroscopes with magnetic active and magnetic resonance suspensions are still being developed and manufactured, it can be stated that studies concerning angular motions of the rotor's axis of proper rotation, which characterize its errors, are relevant. In this aspect, for a symmetric Euler gyroscope designed for GTR validation [1,4], the parameters of its regular precession are evaluated, i.e. its errors, including the Poinot analysis. A fundamental presentation of the theory of symmetric Euler gyroscopes with the Poinot and McCullagh analyses of motion is given in [5][6].

It should be recalled that elementary particles (electrons, protons, etc.) are essentially Euler gyroscopes [3] (one can say that the entire Universe consists of corpuscular Euler gyroscopes), which also emphasizes the relevance of this study.

4 b) Formulation of the problem

The solution to the problem of inertial motion of a symmetric Euler gyroscope is well known and described in many works, in particular, in [1][2]. This motion is regular precession, characterized by a constant angle of nutation between the kinetic moment axis, superimposed with the inertial basis axis, and the axis of SEG proper rotation. At the same time, the angular velocities of precession and nutation are constant.

The indicated properties have found application in [4] in the process of preparation of an experiment to validate the general theory of relativity using a SEG and a telescope on an artificial Earth satellite when solving the problem of selection of relations between the primary moments of inertia that provide very low angular precession velocities. In the experiment [1], drift angular velocities values were less than 10^{-11} angular deg/hr, which validated the Einsteinian general theory of relativity with an error of less than 1%.

It should be noted that the solution to the problem of regular precession was possible with the following restrictions on the initial Euler angles [6, formulas (2.39), (2.41)]:

where G is the kinetic moment; r_0 is the SEG proper rotation angular velocity component; C is the primary moment of SEG inertia around the same axis.

This paper sets the task of finding the solution to the problem of SEG motion for arbitrary initial angles not only along the precession angle θ , but also along the initial angles of nutation and proper rotation. The Poisson differential kinematic equations are used for this purpose. To clarify the problem formulation, let us cite a statement on this subject from the work [7, p. 79]. The first step in solving the problem is to determine the angular velocities of the body. This is solved analytically regardless of the Euler angles. The second step consists of determining the Euler angles by integrating the kinematic equations due to the angular velocities found in the first step. This long and arduous process is eased by applying the kinetic moment theorem and the method of selection of a coordinate system, one of the axes of which coincides with the kinetic moment vector [5][6], etc. For this article, we chose the way of integration of the matrix differential equations in quaternions, as well as of Poisson equations by means of solving the Cauchy problem with arbitrary initial angles, which is not related to the special selection of a coordinate system, one of the axes of which is directed along the kinetic moment vector of the SEG.

5 II.

On the Influence of Initial Conditions for Kinematic Equations on the Nature of Motions in a Symmetric Euler Gyroscope

In this section, we set the task to clarify the range of values of the initial Euler angles for the kinematic equations of the symmetric Euler gyroscope, with which they are reduced to identities - after substituting their analytical solutions given in [7], as well as the solutions of dynamic equations given in [6]. Since these solutions describe regular precession, we are talking about the initial conditions under which it is observed, and under which it is not.

Dynamic equations for a symmetric Euler gyroscope have the form [7, p. 126]:

$$\begin{aligned} & \ddot{\alpha} + \frac{C}{r} \dot{\alpha} \dot{\beta} \dot{\gamma} = \text{const} \cdot r \cdot \dot{\alpha} \dot{\beta} \dot{\gamma} \cdot \dot{\alpha} \dot{\beta} \dot{\gamma} \\ & C \cdot \ddot{\alpha} + C \cdot \ddot{\beta} + C \cdot \ddot{\gamma} + \frac{C}{r} \dot{\alpha} \dot{\beta} \dot{\gamma} = 0; 0; 0 \end{aligned} \quad (\text{A.1})$$

The kinematic Euler equations [7, p. 115]:

$$\begin{aligned} & \dot{\alpha} \cos \sin \cos \sin \cos \sin \sin r \cdot q \cdot p \\ & \dot{\beta} \cos \sin \cos \sin \cos \sin \sin r \cdot q \cdot p \end{aligned} \quad (\text{A.2})$$

The solutions of these equations obtained in [7, p. 37]:

$$\begin{aligned} & \alpha = \alpha_0 + \omega_\alpha t; \beta = \beta_0 + \omega_\beta t; \gamma = \gamma_0 + \omega_\gamma t \\ & \text{const} \cdot G \cdot C \cdot r \cdot \dot{\alpha} \dot{\beta} \dot{\gamma} = \text{const} \cdot G \cdot C \cdot r \cdot \dot{\alpha} \dot{\beta} \dot{\gamma} \end{aligned} \quad (\text{A.3})$$

7 PROBLEM SOLUTION

or, equivalently, through quaternion matrices [10], [11]:

$$\begin{pmatrix} 0 & 0 & 1 & 1 & ? & ? & ? & ? & T & T & T & N & M & M & N \\ ? & ? & ? & ? & (6) & ? & ? & ? & ? & ? & ? & 0 & 0 & 0 & ; \\ 0 & ; & 0 & ; & 1 & 1 & 1 & 1 & N & M & A & N & M & A & N \\ N & N & N & N & N & N & N & N & N & T & T & ? & ? & ? & ? & ? & ? \end{pmatrix}$$

where N , A are the quaternion matrix and the matrix of directional cosines of the resulting rotation; N_1 , A_1 are the matriciants; N_x , N_y , N_z are the quaternion matrices of the corresponding simplest rotations. At the same time, M and N are the corresponding types of quaternion matrices [10,11].

The matrix of directional cosines of the Euler angles for Fig. 1

[illegible][illegible]

The system (??), (??0) is reduced to an equivalent differential equation with constant coefficients $\frac{dN}{dt} = \lambda N$ (12)

. 00000000;11110123103223013210????????????????????
 ??????????aRaRRaRaPNB????????????????(13). 2 sin; 0; 2 cos; 3 2 1 0
 1ttACRR????????

Given these formulas, we have:

$$2 \cos 0 \quad 0 \quad 2 \sin 0 \quad 2 \cos 2 \sin 0 \quad 0 \quad 2 \sin 2 \cos 0 \quad 2 \sin 0 \quad 0 \quad 2 \cos ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ? \quad ?$$

 $? \quad ? \quad ? \quad ? \quad t \quad t \quad t \quad t \quad t \quad t \quad N(14)$

The equivalence of equations (??) and (12), (??3) is confirmed by the fulfillment of the identity? ? B P N
PN N ? ? ? ? ? ? ? 1 1 ? (15)

The solution to the equation (12) with constant coefficients is the Cauchy formula: $\frac{1}{2\pi i} \int_{\Gamma} \frac{f(z)}{z - \lambda} dz$, where Γ is a contour in the complex plane enclosing the point λ .

where $L(t)$ is the fundamental matrix of solutions; $N Z(0)$ is the matrix of initial values of the angles, equal, by condition, to the identity matrix: $N Z(0)=E$.

After finding the fundamental matrix of solutions and a number of transformations, let us write down the expression (16) in the form: $\frac{1}{2} \sin 2 \cos 2 / 1 2 1 2 0 1 3 0 1 \sin n R a d R d a N D E N$ (17)

After transformations, the matriciant takes the form:

$$\begin{pmatrix} \cos 2 & \sin 0 & 2 \sin 2 & \sin 2 & \cos 2 & \sin 0 & 0 & 2 \sin 2 & \cos 2 & \sin 2 & \sin 0 & 2 \sin 2 & \cos 1 & 1 & 1 & ? & ? & ? & ? & ? & ? & ? & ? & n & a & n \\ n & a & n & R & n & R & n & a & n & R & n & a & N & Z & (18) \end{pmatrix}$$

From the expression (11) we have: $00; 11NNNNNNNNNZTZT????; 1N =) (1??$
 $N ? (k=0,1,2,3). \quad (19)$

In consideration of (??3), (??4) and (??8), the expanded expression for the quaternion matrix N^{-1} is derived below.

Since? ? ? ? 0 0 1 N N N N N N Z T ? ? ?

, we have the following expression for the quaternion matrix of the resulting rotation N for nonzero initial conditions:

```
. 03 02 01 000 1 2 3 1 0 3 2 2 3 0 1 3 2 1 0 ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ? ?  
nnnnnnnnnnnnnnnnnnnnnnnnnnNaaaaaaaaaaaaaaaaaaaa
```

Formulas for the components of the quaternion matrix N: ?
? ? ? ? ? ? ? ? ? ? ? ? 03 0? ? 3 , 0 ? ? i n a i a ? .(21)

We have the explicit form of the formulas for the components of the quaternion matriciant N :

$$N = \begin{pmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}, \quad (22)$$

For regular precession, the angles of the initial orientation and the components of the initial quaternion are expressed by the formulas:

$$\alpha = \arccos\left(\frac{H_z}{H}\right), \quad \beta = \arccos\left(\frac{n_z}{n}\right), \quad \gamma = \arccos\left(\frac{H_z n_x - H n_z x}{H n \sqrt{1-x^2}}\right), \quad \delta = \arccos\left(\frac{H_z n_y + H n_z y}{H n \sqrt{1-x^2}}\right) \quad (23)$$
[illegible]

After that, let us similarly determine the trigonometric functions for the Euler-Krylov angles φ , θ , on the basis of the matrix (8) and its quaternion counterpart [8,9]. We have:

These expressions coincide with formulas (18) [8], confirming the fidelity of the solutions to the problem for zero initial Euler-Krylov angles both in the quaternion form and in the form associated with the application of the Poisson differential kinematic equations.

For arbitrary initial Euler-Krylov angles, explicit solutions can be obtained from relations (24), (25) (in (25), the φ_i must be replaced by values $\varphi_i^0, 0 \leq i \leq n$).

In turn, for the Euler angles we have the following solutions:(27)

sign of the angle of nutation. After transformations, the formulas for determining the Euler angles for the SEG are:

(53)

The expressions (53) suggest that only the change of the sign of the initial angle of nutation -with the other two initial angles unchanged -caused the appearance of irregular precession motions in the Euler gyroscope.

IV.

10 Mathematical Modeling

?? ? ? ? ? ? ? ? ? ? ? ? ? ? ? (M.1)

That is, corresponding to the conditions (23) of regular precession in the Euler angles. The relationship between the Euler and the Euler-Krylov angles is established due to the equality of the respective elements of the matrices (7) and (8). The graphs in Fig. 3 depict the change of the Euler angles for regular precession. The same cannot be said about the graphs in Fig. 4 for the Euler-Krylov angles -where one can see harmonic oscillations for the angles θ and ϕ with a frequency slightly higher than 500 Hz, and for the angle ψ , its increscent property is evident. When applying a stronger rotational pulse around the axis Ox for which $\omega = 4000$, unchanged other conditions for Fig. 3 and 4, the nature Fig. 4 When applying a stronger rotational pulse $\omega > \omega_0$, with unchanged other conditions for Fig. 3 and 4, the nature of the motion does not change (therefore, the graphs are not shown), however, for the Euler angles we Additionally, with unchanged parameters of modeling of SEG motions according to (M.1), (M.2) (figures 3 and 4), but with the sign of the initial angles of nutation reversed and equal to π motion patterns shown in figures 5 and 6 were obtained. In Fig. 5, for the Euler angles, the motion has acquired the character of irregular precession, namely, along θ a vibrational pattern with frequencies slightly above 500 Hz of different amplitudes with oscillation centers shifted by about 0.3 rad. For the angle ϕ , the velocity sign in Fig. 3 has changed to the opposite, and the angle become increscent. The graphs confirm the derived formulas (30).

For the Euler -Krylov angles, the motion is of a qualitatively similar character. At the same time, the motion for the Euler Krylov angles has changed dramatically (Fig. 8).

angle θ began to increase monotonically in the up to 0.45 rad, and the oscillation increased 820 Hz. The angle ϕ remains to be in crescent with superimposed oscillations.

At the same time, the motion for the Euler-Krylov angles has changed dramatically (Fig. 8). The began to increase monotonically in the direction of the rotational pulse action, which is novel.

The angle ψ is still oscillatory in nature with a frequency of 820 Hz around the shifted center of oscillations, and the angle ϕ has changed the sign to the opposite in relation to Fig. ??.

V.

11 Conclusion

According to the results of mathematical modeling, it is shown that the motions that correspond to regular precession in the Euler angles are independent of the magnitude of the angular velocity ω , which is caused by the action of the rotational pulse. However, a change of the sign of the initial angle of nutation leads to a sharp change in the nature of motion -it becomes irregular, which is reflected in the explanation for Fig. 5. The motion along the Euler-Krylov angles radically depends on α : with $R \alpha > 0$, the angle θ becomes monotonically increscent in the direction of the pulse action, and the angle of proper rotation changes the sign of its monotonic rotation to the opposite. Additionally, in the article: As for corpuscular gyroscopes, based on this study, it can be assumed that depending on the application of an external magnetic field over time, not only Larmor precession [14], but also "pseudo-Larmor" precession is possible in them.

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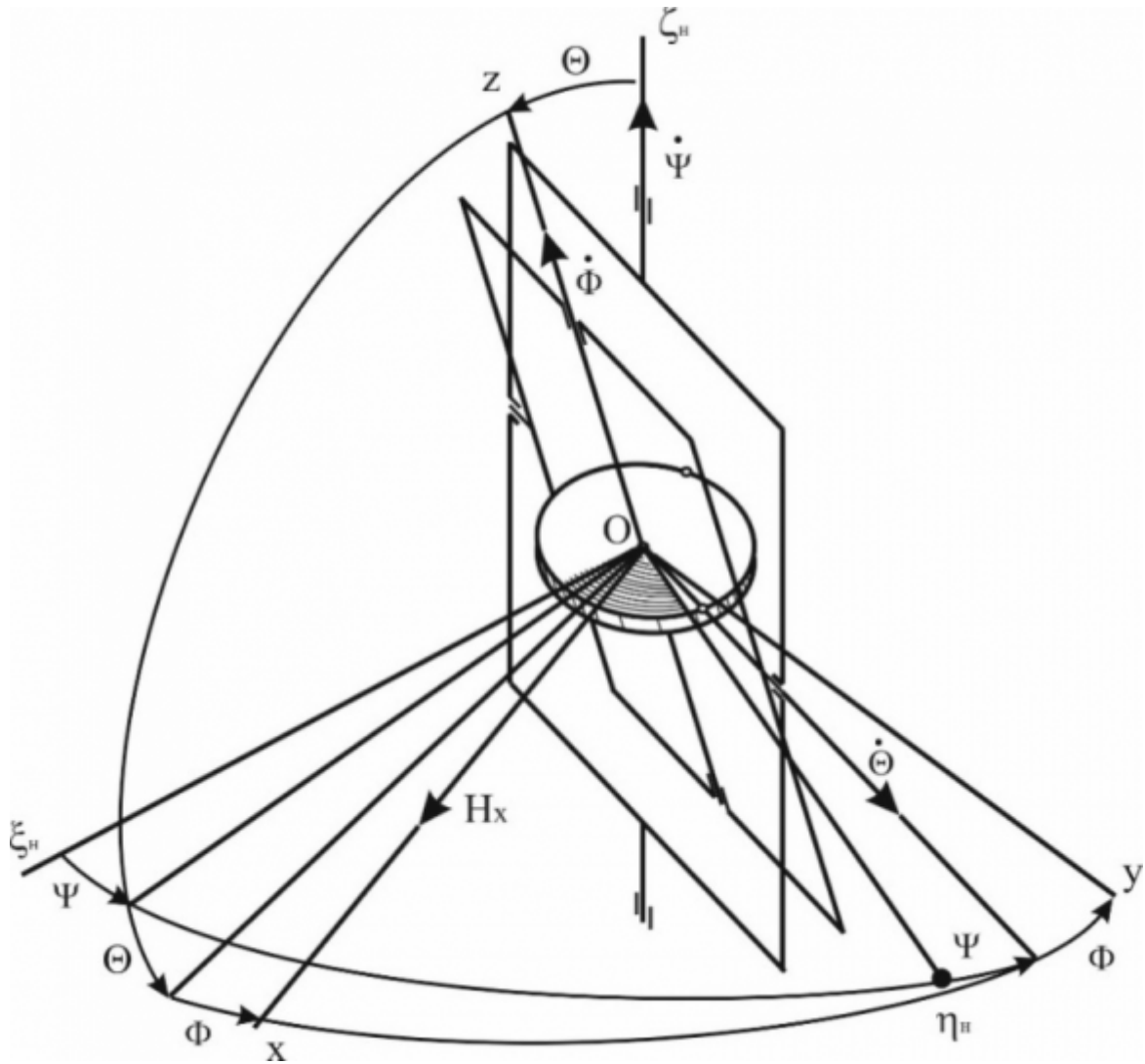


Figure 1:

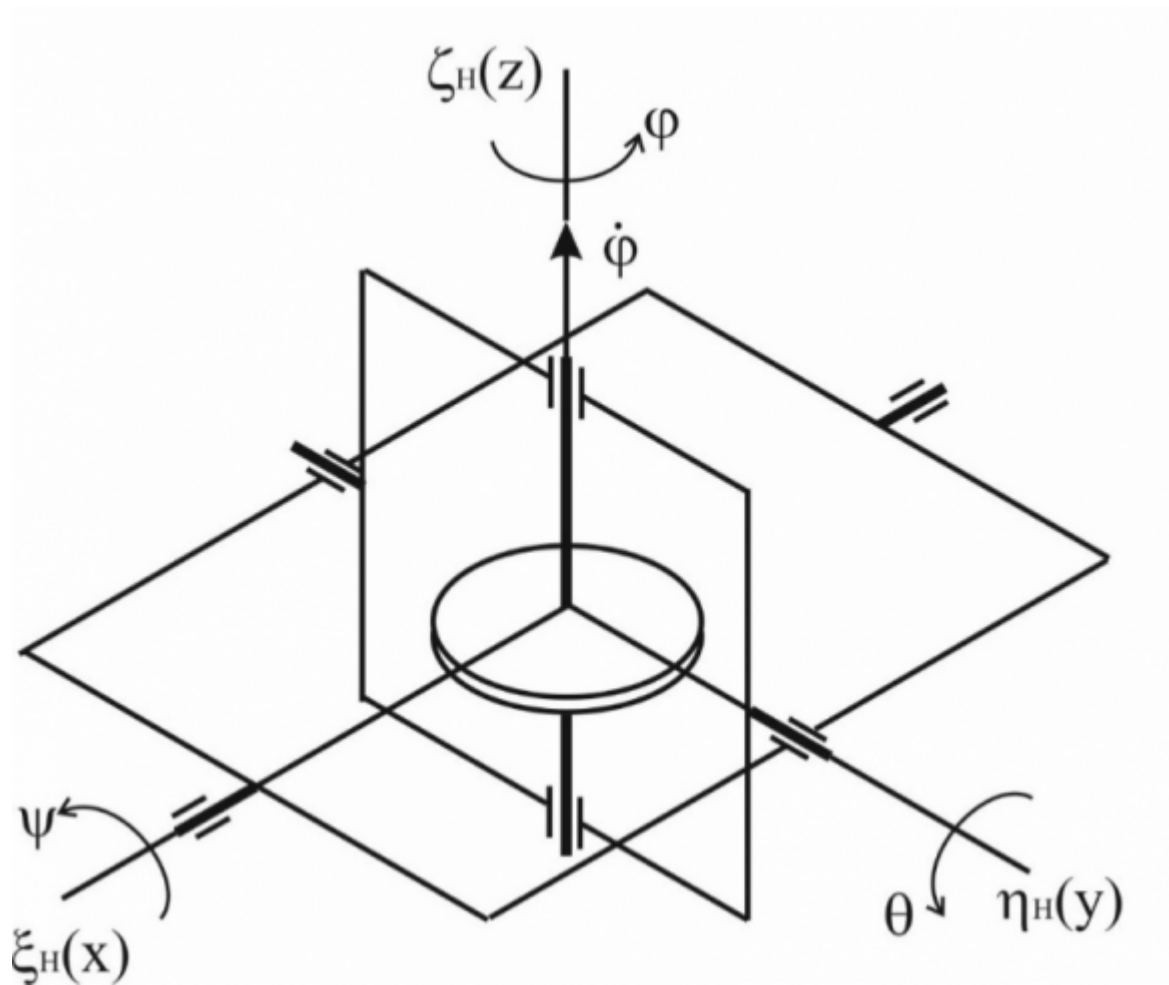


Figure 2:

$$Z(t) = Q(t) \cdot Q^{-1}(0) \cdot Z(0)$$

Figure 3:

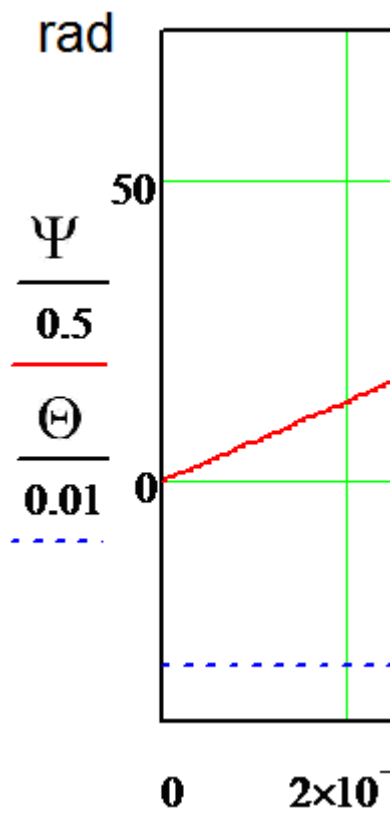


Figure 4:

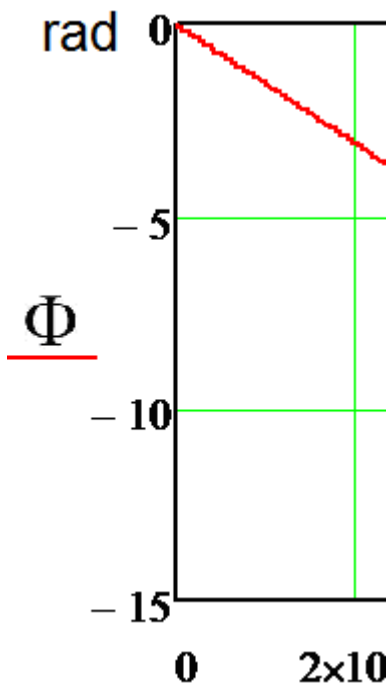


Figure 5:

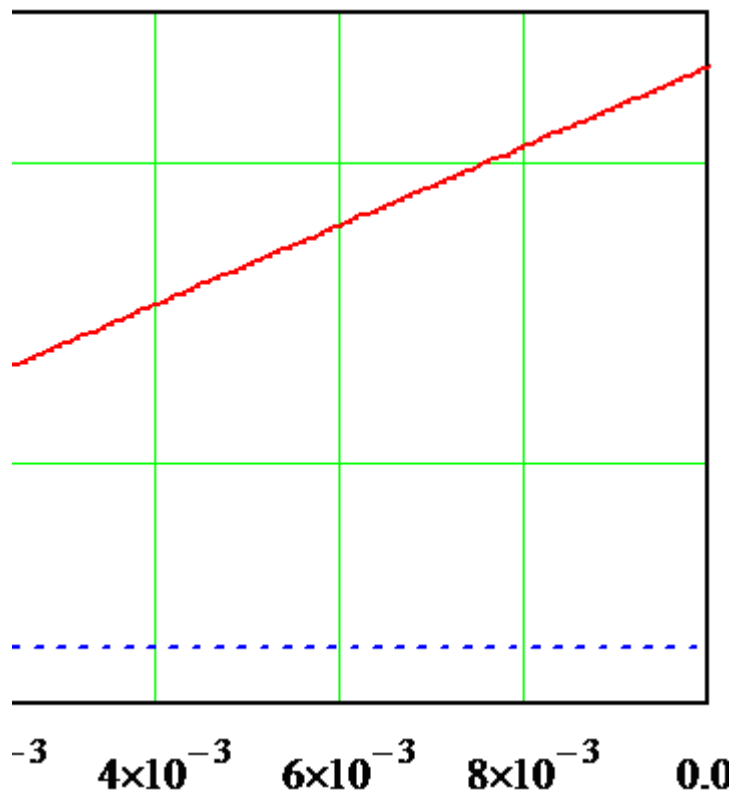


Figure 6:

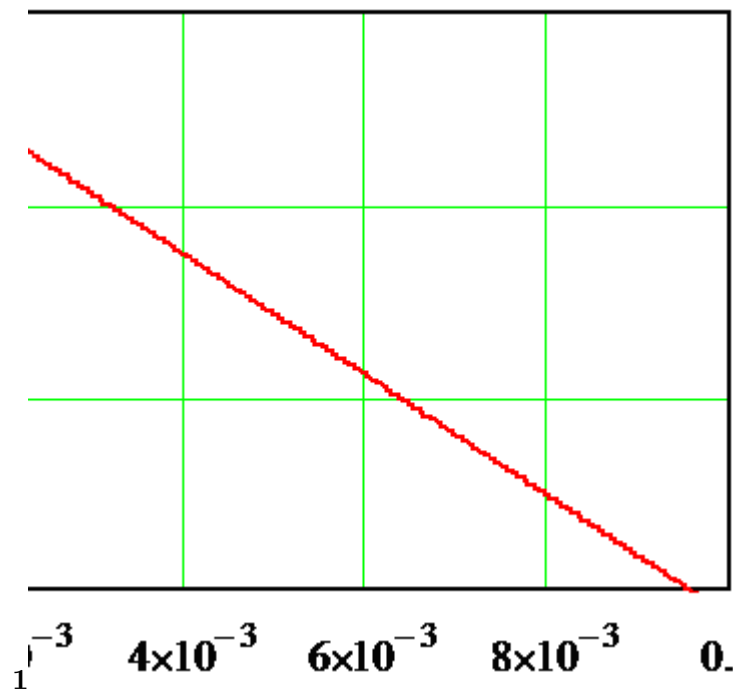


Figure 7: 1 a

t, s
01

Figure 8: For

t, s
01

Figure 9: ;

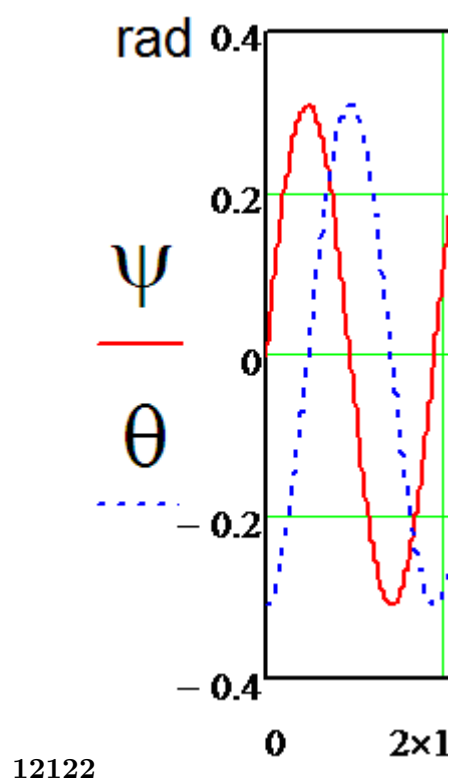


Figure 10: 1 ?? 2 (1 ? 2 ?? 2

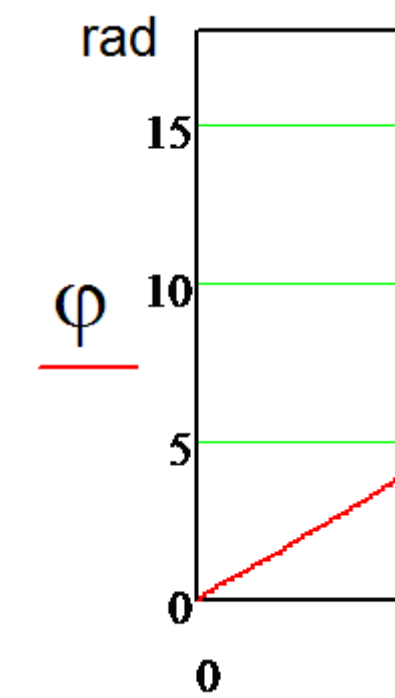


Figure 11: Figures 3 -

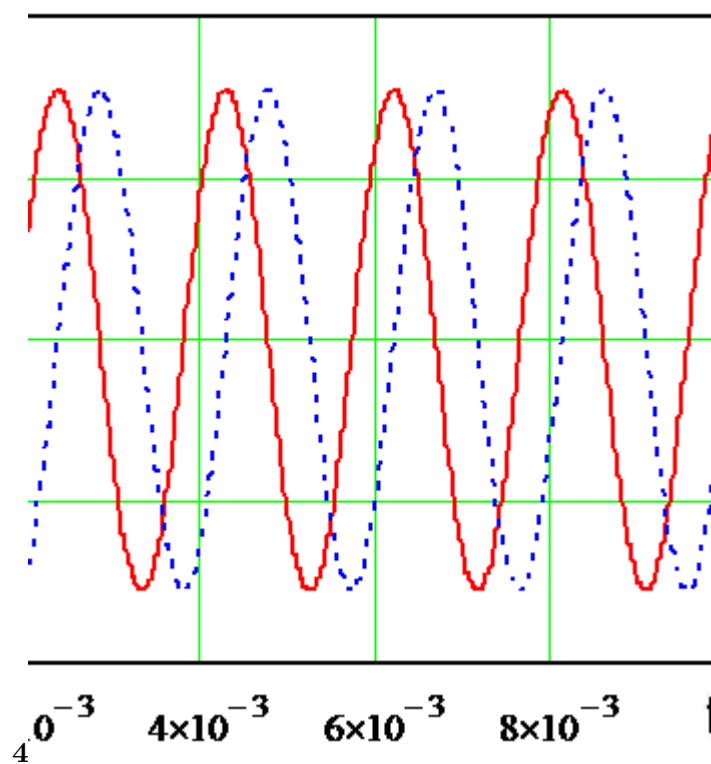


Figure 12: Figures 3 and 4

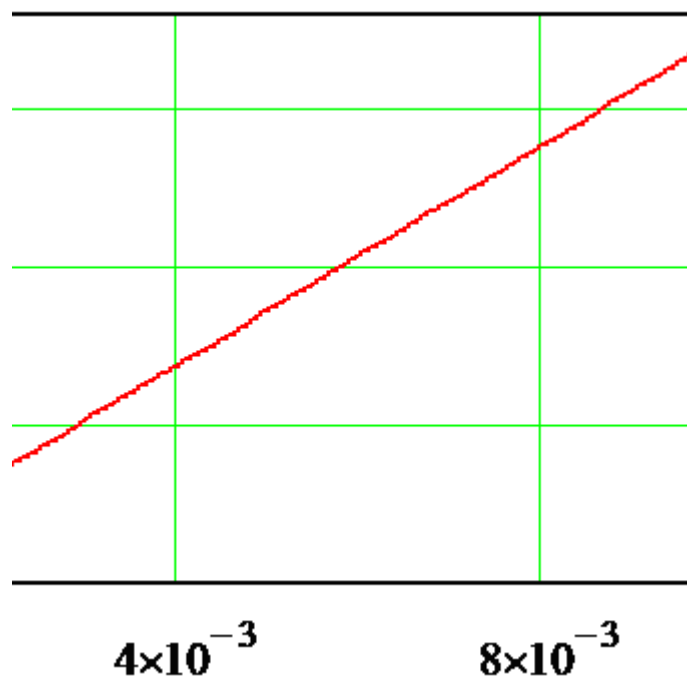


Figure 13:

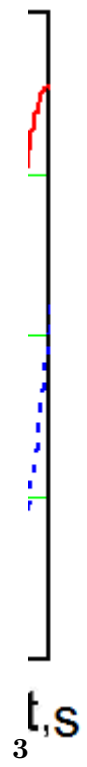


Figure 14: Fig. 3 ©

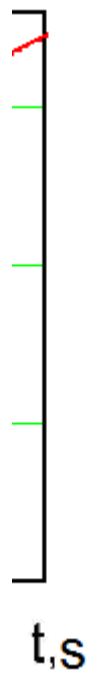


Figure 15:

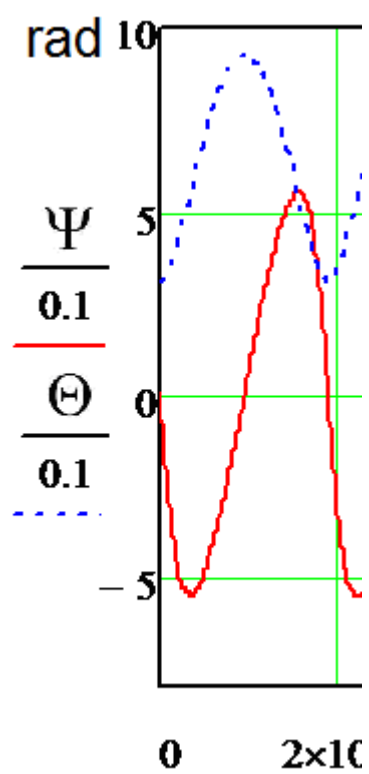


Figure 16:

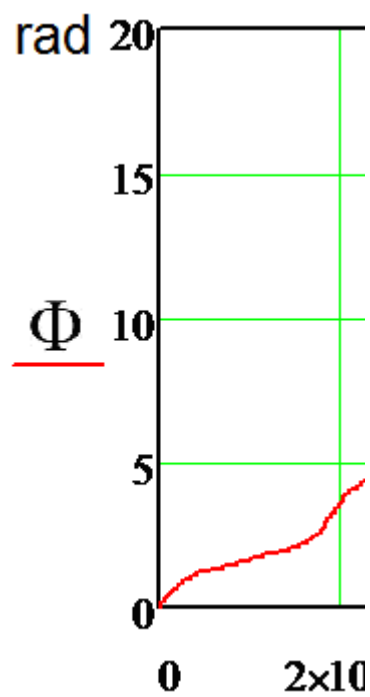


Figure 17: ?

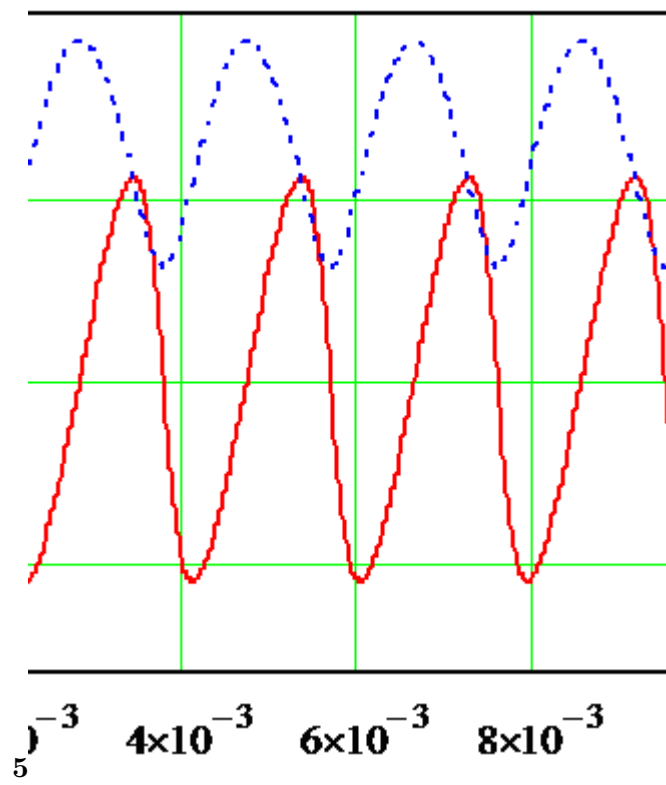


Figure 18: Fig. 5 velocity

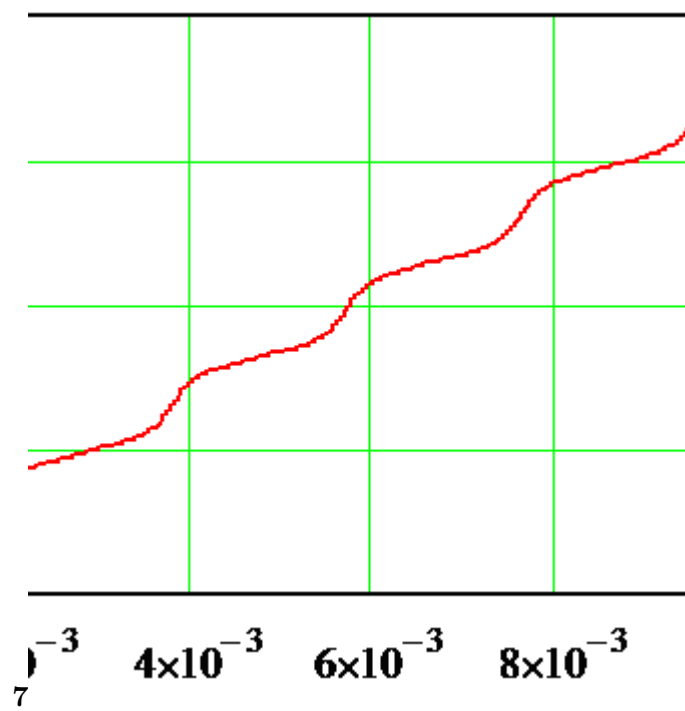


Figure 19: Fig. 7



Figure 20: Figures 7 and 8



Figure 21: Fig. 8

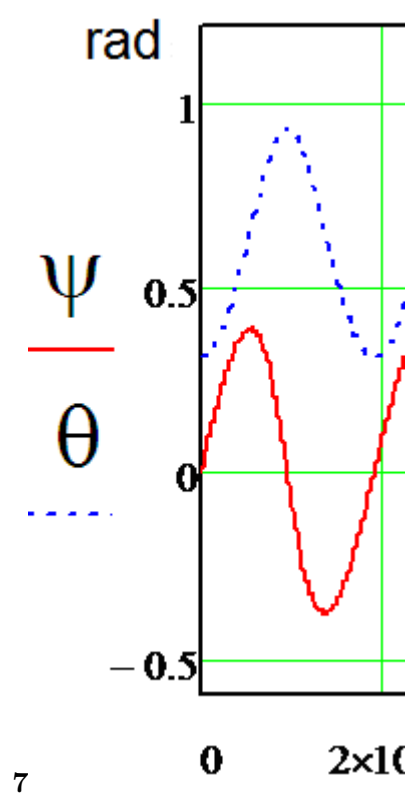


Figure 22: Figures 7

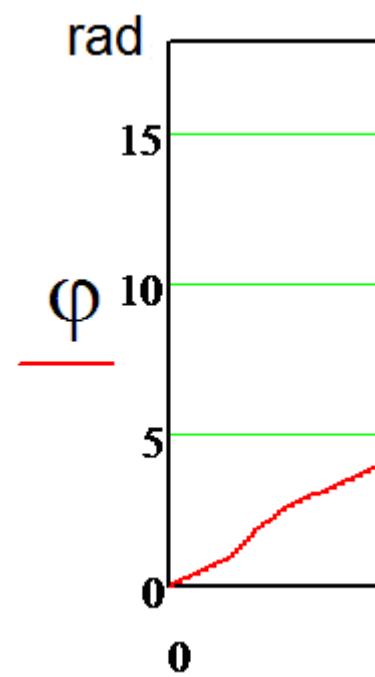


Figure 23: ?

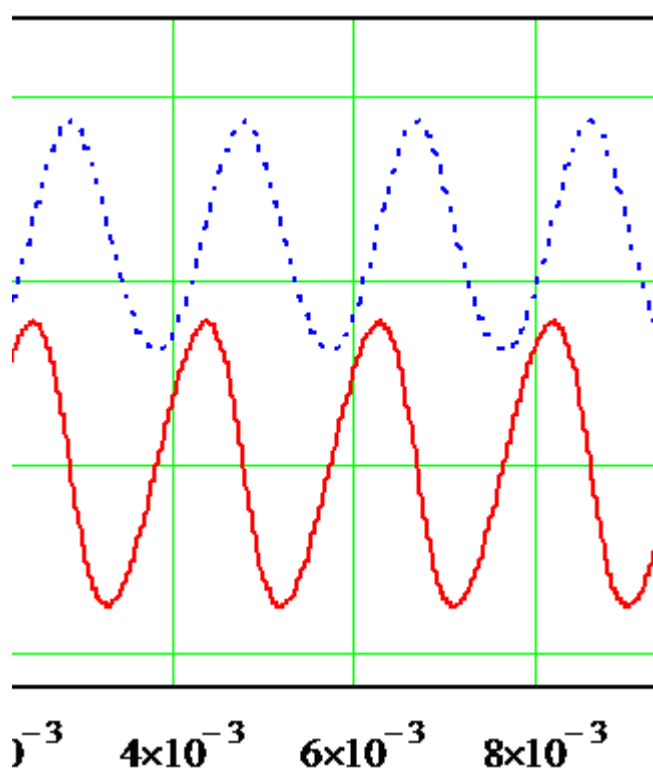


Figure 24:

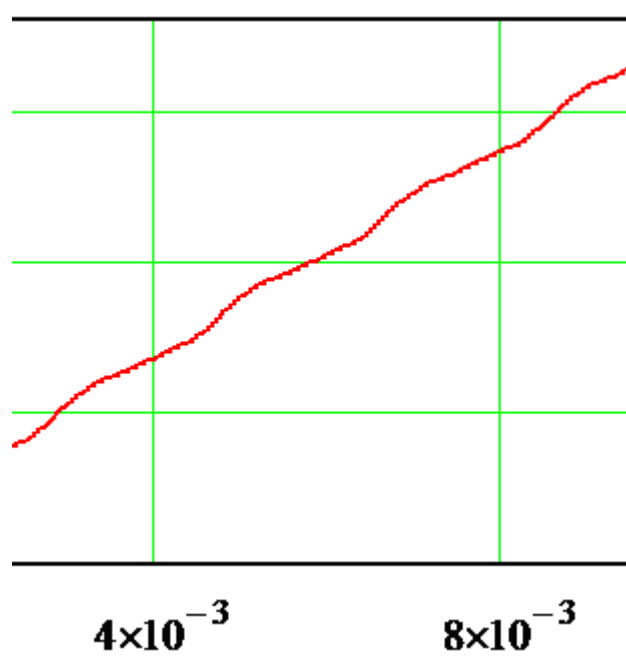


Figure 25:

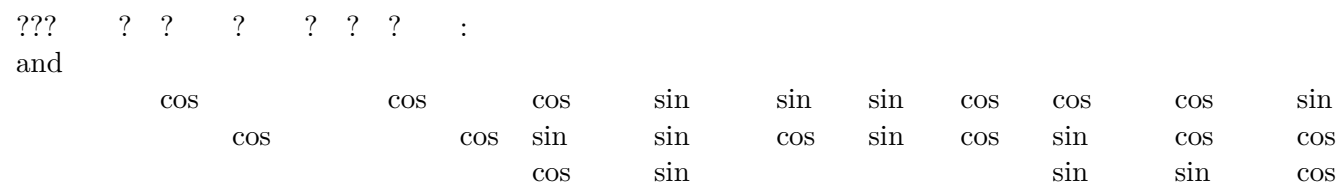


Figure 26:

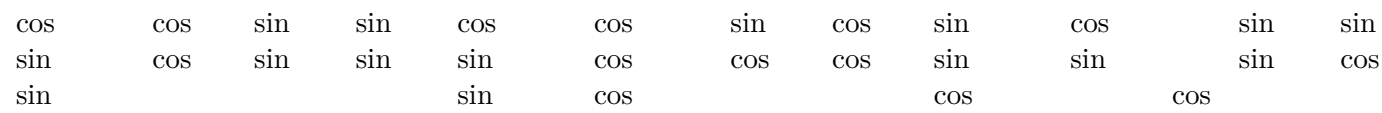


Figure 27:

328 Let us now apply the obtained formulas to the case of regular precession. We use the initial values $\omega = 0$; $\dot{\omega} =$
329 ? in the matrix \hat{M} associated with this type of precession

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