



GLOBAL JOURNAL OF RESEARCHES IN ENGINEERING: J
GENERAL ENGINEERING
Volume 14 Issue 7 Version 1.0 Year 2014
Type: Double Blind Peer Reviewed International Research Journal
Publisher: Global Journals Inc. (USA)
Online ISSN: 2249-4596 & Print ISSN: 0975-5861

Analytical Solution of a Guided Simply Supported Beam under Three Point Bending

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GJRE-J Classification : FOR Code: 099999



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Analytical Solution of a Guided Simply Supported Beam under Three Point Bending

Shabbir Ahmed ^α & Mohammad Ikthair Hossain Soiket ^σ

Abstract- In the present paper, elastic field parameters of a guided simply supported beam under three point bending has been investigated by a new approach called displacement potential formulation. Guided beams involve mixed mode of boundary conditions which the classical beam theory cannot handle. In displacement potential formulation, all the elastic field parameters has been expressed in terms of a single function Ψ which gives rise to a fourth order partial differential equation. The solution of this equation has been determined by trial and error process satisfying all the boundary conditions. This solution is then inserted to reduce the partial differential equation into an ordinary one. The obtained analytical solution of the beam has been presented as graphs. The stress concentration occurs mainly at the point of application of the load and the support at two corners. The analytical solution has further been verified by finite element analysis.

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I. INTRODUCTION

A beam may be considered as one of the most commonly used structural elements in engineering applications. A beam is said to be a deep beam when the depth is comparable to its span. Design of deep beams based on classical Euler bending theory can be seriously erroneous, since the simple theory of flexure takes no account of the effect of normal pressures on the top and bottom edges of the beam caused by the loads and reactions (Chow, Conway and Winter, 1952). The effect of normal pressures on the stress distribution in deep beams is such that the distribution of bending stresses on vertical sections is not linear and the distribution of shear stresses is not parabolic. Consequently, a plane transverse section does not remain plane after bending, and the neutral axis does not lie at the mid-depth, which eventually causes the basis of classical theory to be violated.

In an attempt to make up the limitation, different theories as well as methods of solution have been reported in the literature (Conway, Chow and Morgan, 1951; Conway and Ithaca, 1953; Murty, 1984; Suzuki, 1986).

However, each solution possesses certain limitations, and eventually none of the solutions are found to conform to all the physical characteristics of the problem for deep beam appropriately. Even, photoelastic studies (Uddin, 1966), finite element analysis (Hardy and Pipelzadeh, 1991) and finite difference solutions (Ahmed, Idris and Uddin, 1966; Ahmed, Khan and Uddin, 1998; Ahmed, Idris and Uddin, 1999; Ahmed, Hossain and Uddin 2005; Akanda, Ahmed and Uddin, 2002) have also been carried out for deep beams on two supports, mainly because all the physical conditions imposed on the beam could not be fully taken into account in the analytical methods of solution. Among the existing mathematical models of elasticity for the plane boundary-value problems, the stress function approach and the displacement formulation are noticeable. The stress function approach accepts boundary conditions in terms of loading only; boundary restraints cannot be satisfactorily imposed on it. On the other hand, the displacement formulation involves extreme difficulty especially when the boundary conditions are a mixture of restraints and stresses. As a consequence, serious attempts had hardly been made in the past for stress analysis using this formulation. As such, neither of the existing formulations is suitable for solving problems of mixed boundary conditions.

Further, the use of standard structures, like beams, columns, etc. with guides on part or full of their bounding surfaces is receiving increased importance in order to satisfy precise and strict design criteria in many of the engineering applications. Guided boundaries usually help in reducing the level of deformation in the structural elements, which eventually resist the change of the original shape of the bounding surfaces under loading. But structures with guided boundaries always remain away from the scope of analytical solutions, because the physical conditions of guided boundaries need to be mathematically modelled in terms of a mixed mode of boundary conditions.

Since the exact analytical solution of mixed-boundary-value elastic problems, is beyond the scope of existing mathematical models of elasticity, the use of a new mathematical formulation will be investigated to analyze the elastic behavior of a guided deep beam under three point bending loading and support arrangements. It would be worth mentioning that, as far as the reporting in the literature is concerned, the author has not come across any reliable study of the present

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problem. Therefore, the analytical solution for a guided deep beam under three point bending is yet to be developed.

II. BOUNDARY CONDITIONS

The physical conditions at different boundaries of the beam are expressed mathematically as follows:

- $u_x = 0$ at the edge of $x = 0$
- $u_x = 0$ at the edge of $x = L$
- $\sigma_{xy}(0, y) = 0$ at the edge of $x = 0$
- $\sigma_{xy}(L, y) = 0$ at the edge of $x = L$
- $\sigma_{xy}(0, y) = 0$ at the edge of $y = 0$
- $\sigma_{xy}(0, y) = 0$ at the edge of $y = D$
- The lateral stress at the edge of $y = D$ is related to the applied load for the three point bending. Since the point load is actually acting over a certain area of the beam, for instance it can be considered for the length of $x=0.45L$ to $0.55L$. Again it is considered that the load intensity is σ_0 . Therefore, the magnitude of point load, $P = 0.1L\sigma_0$. Then for $x=0.45L$ to $0.55L$
- Similarly, the lateral stress at the edge of $y = 0$ is related to the reactions at the support. In this case $\sigma_{yy}(x, 0) = \sigma_0/2$ for $x=0$ to $0.1L$ and $0.9L$ to L .

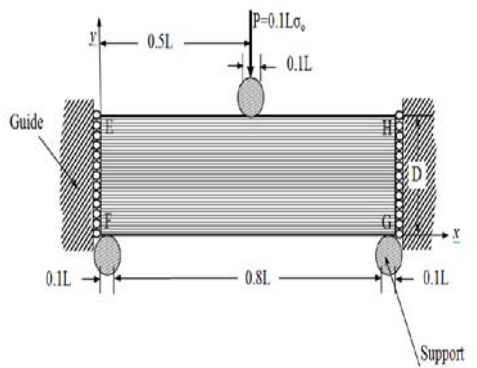


Fig. 1: Geometry and loading (symmetric) of the guided simply supported beam under three point bending

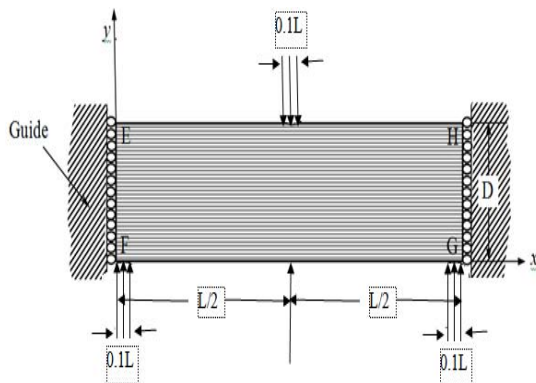


Fig. 2: Analytical model of the guided beam

III. ANALYTICAL SOLUTION

The equation of equilibrium for isotropic material is as follows (Timoshenko and Goodier, 1970):

$$\frac{\partial^4 \psi}{\partial x^4} + 2 \frac{\partial^2 \psi}{\partial x^2 \partial y^2} + \frac{\partial^4 \psi}{\partial y^4} = 0 \quad (1)$$

The expressions of displacement and stress components in terms of function $\psi(x, y)$ are as follows (Nath, Ahmed and Afsar, 2006):

$$u_x(x, y) = \frac{\partial^2 \psi}{\partial x \partial y} \quad (2a)$$

$$u_y(x, y) = -\frac{1}{1+\mu} \left[2 \frac{\partial^2 \psi}{\partial x^2} + (1-\mu) \frac{\partial^2 \psi}{\partial y^2} \right] \quad (2b)$$

$$\sigma_{xx}(x, y) = -\frac{E}{(1+\mu)^2} \left[\frac{\partial^3 \psi}{\partial x^2 \partial y} - \mu \frac{\partial^2 \psi}{\partial y^2} \right] \quad (2c)$$

$$\sigma_{yy}(x, y) = -\frac{E}{(1+\mu)^2} \left[(2+\mu) \frac{\partial^3 \psi}{\partial x^2 \partial y} + \frac{\partial^3 \psi}{\partial y^3} \right] \quad (2d)$$

$$\sigma_{xy}(x, y) = -\frac{E}{(1+\mu)^2} \left[\frac{\partial^3 \psi}{\partial x^3} - \mu \frac{\partial^3 \psi}{\partial x \partial y^2} \right] \quad (2e)$$

The potential function $\psi(x, y)$ is first assumed in a way so that the physical conditions of the two opposing guided ends are automatically satisfied. At the same time solution has to satisfy the 4th order partial differential equation. After a long trial and error process, the solution of the governing equation (1) is thus approximated as follows:

$$\psi(x, y) = \sum_{m=1}^{\infty} Y_m(y) \cos \alpha x + K y^3 \quad (3)$$

where, $Y_m = f(y)$, $\alpha = (m\pi/L)$, K is an arbitrary constant and $m = 1, 2, 3, \dots, \infty$.

Derivatives of equation (3) with respect to x and y are substituted in Eq. (1) and following equation is obtained:

$$Y_m''' - 2\alpha^2 Y_m'' + \alpha^4 Y_m = 0 \quad (4)$$

The solution of the above 4th order ordinary differential equation with constant coefficients [Eq. (4)] can normally be approximated as follows:

$$Y_m = A_m e^{r_1 y} + B_m y e^{r_2 y} + C_m e^{r_3 y} + D_m y e^{r_4 y} \quad (5)$$

But the ordinary differential equation (4) has the complementary function of repeated roots. Thus $r_1 = r_2 = \alpha$, $r_3 = r_4 = -\alpha$ and the general solution of Eq. (4) can be written as:

$$Y_m = (A_m + B_m y) e^{\alpha y} + (C_m + D_m y) e^{-\alpha y} \quad (6)$$

where A_m , B_m , C_m and D_m are arbitrary constants.

Now substituting the derivatives of ψ and Y_m in the expressions for displacement and stresses following expressions are found:

$$u_x(x, y) = \frac{\partial^2 \psi}{\partial x \partial y} = - \sum_{m=1}^{\infty} [A_m \alpha e^{\alpha y} + B_m (\alpha y + 1) e^{\alpha y} - C_m \alpha e^{-\alpha y} - D_m (\alpha y - 1) e^{-\alpha y}] p \sin \alpha x \quad (7a)$$

$$u_y(x, y) = - \frac{1}{(1+\mu)} \left[2 \frac{\partial^2 \psi}{\partial x^2} + (1-\mu) \frac{\partial^2 \psi}{\partial y^2} \right]$$

$$= \frac{-1}{(1+\mu)} \left[\sum_{m=1}^{\infty} \left\{ \begin{array}{l} -A_m (1+\mu) \alpha^2 e^{\alpha y} + \\ B_m (-\alpha y - \mu \alpha y - 2\mu + 2) \alpha e^{\alpha y} \\ -C_m (1+\mu) \alpha^2 e^{-\alpha y} + \\ D_m (-\alpha y - \mu \alpha y + 2\mu - 2) \alpha e^{-\alpha y} \end{array} \right\} \cos \alpha x + 6K(1-\mu)y \right] \quad (7b)$$

$$\sigma_{xx}(x, y) = \frac{E}{(1+\mu)^2} \left[\frac{\partial^3 \psi}{\partial x^2 \partial y} - \mu \frac{\partial^3 \psi}{\partial y^3} \right]$$

$$= \frac{-E}{(1+\mu)^2} \left[\sum_{m=1}^{\infty} \left\{ \begin{array}{l} A_m \alpha (1+\mu) e^{\alpha y} + B_m (\alpha y + \mu \alpha y + 3\mu + 1) e^{\alpha y} \\ -C_m \alpha (1+\mu) e^{-\alpha y} + D_m (-\alpha y - \mu \alpha y + 3\mu + 1) e^{-\alpha y} \end{array} \right\} \alpha^2 \cos \alpha x + 6\mu K \right] \quad (7c)$$

$$\sigma_{yy}(x, y) = \frac{-E}{(1+\mu)^2} \left[(2+\mu) \frac{\partial^3 \psi}{\partial x^2 \partial y} + \frac{\partial^3 \psi}{\partial y^3} \right]$$

$$= \frac{-E}{(1+\mu)^2} \left[\sum_{m=1}^{\infty} \left\{ \begin{array}{l} A_m \alpha (-1-\mu) e^{\alpha y} + B_m (-\alpha y - \mu \alpha y - \mu + 1) e^{\alpha y} + \\ C_m \alpha (1+\mu) e^{-\alpha y} + D_m (\alpha y + \mu \alpha y - \mu + 1) e^{-\alpha y} \end{array} \right\} \alpha^2 \cos \alpha x + 6K \right] \quad (7d)$$

$$\sigma_{xy}(x, y) = \frac{-E}{(1+\mu)^2} \left[\frac{\partial^3 \psi}{\partial x^3} - \mu \frac{\partial^3 \psi}{\partial x \partial y^2} \right]$$

$$= \frac{-E}{(1+\mu)^2} \left[\sum_{m=1}^{\infty} \left\{ \begin{array}{l} A_m (1+\mu) \alpha e^{\alpha y} + B_m (\alpha y + \mu \alpha y + 2\mu) e^{\alpha y} \\ +C_m (1+\mu) \alpha e^{-\alpha y} + D_m (\alpha y + \mu \alpha y - 2\mu) e^{-\alpha y} \end{array} \right\} \alpha^2 \sin \alpha x \right] \quad (7e)$$

Now, the reactions on the bottom boundary ($y = 0$) are acting over the two supports. It is considered that the supports are located at $x=0$ to $0.1L$ and $x=0.9L$ to L respectively. The total length for reaction is 20 percent of beam length. Now the compressive load exerted at the mid-span on the edge $y = D$ of the beam may be considered as acting over at least some length of the beam, for instance $x = 0.45L$ to $0.55L$. As a result the intensity of reaction is half of the load intensity. Therefore, the reactions over the beam at the supports can be taken as Fourier series in the following manner:

$$\sigma_{yy}(x, D) = \sigma_0 = I_0 + \sum_{m=1}^{\infty} I_m \cos \alpha x \quad \text{For } x = 0.45L \text{ to } 0.55L$$

Here

$$I_0 = \frac{1}{L} \left[\int_{9L/20}^{11L/20} \sigma_0 dx \right] = \frac{\sigma_0}{10} \quad (8a)$$

$$I_m = \frac{2}{L} \left[\int_{9L/20}^{11L/20} \sigma_0 \cos \alpha x dx \right]$$

$$= \frac{2\sigma_0}{\alpha L} \left\{ \sin \left(\frac{11\alpha L}{20} \right) - \sin \left(\frac{9\alpha L}{20} \right) \right\}$$

$$= \frac{2\sigma_0}{m\pi} \left\{ \sin \left(\frac{11m\pi}{20} \right) - \sin \left(\frac{9m\pi}{20} \right) \right\} \quad (8b)$$

The reaction load at the support on the edge $y = 0$ can also be given by a Fourier series as follows:

$$\sigma_{yy}(x, 0) = \sigma_0 / 2 = E_0 + \sum_{m=1}^{\infty} E_m \cos \alpha x \quad \text{For } x = 0 \text{ to } 0.1L \text{ and } 0.9L \text{ to } L$$

Here

$$E_0 = \frac{1}{L} \left[\int_0^{L/10} \sigma_0 / 2 dx + \int_{9L/10}^L \sigma_0 / 2 dx \right] = \frac{\sigma_0}{10} \quad (9a)$$

$$E_m = \frac{2}{L} \left[\int_0^{L/10} \frac{\sigma_0}{2} \cos \alpha x dx + \int_{9L/10}^L \frac{\sigma_0}{2} \cos \alpha x dx \right]$$

$$= \frac{\sigma_0}{\alpha L} \left\{ \sin\left(\frac{\alpha L}{10}\right) - 0 + \sin(\alpha L) - \sin\left(\frac{9\alpha L}{10}\right) \right\}$$

$$= \frac{\sigma_0}{m\pi} \left\{ \sin\left(\frac{m\pi}{10}\right) + \sin(m\pi) - \sin\left(\frac{9m\pi}{10}\right) \right\} \quad (9b)$$

The loading considerations of equations (8a) and (9a) are to satisfy the boundary conditions at the bottom and top boundaries of the beam. Using boundary condition $\sigma_{xy}(x, 0) = 0$ at the edge of $y = 0$, it is found that:

$$\frac{-E}{(1+\mu)^2} \left[\begin{aligned} &-(2+\mu) \sum_{m=1}^{\infty} \left\{ (A_m \alpha + B_m) + (-C_m \alpha + D_m) \right\} \alpha^2 \cos \alpha x \\ &+ \sum_{m=1}^{\infty} \left\{ (A_m \alpha^3 + 3B_m \alpha^2) + (-C_m \alpha^3 + 3D_m \alpha^2) \right\} \cos \alpha x + 6K \end{aligned} \right] = \sum_{m=1}^{\infty} E_m \cos \alpha x + E_o \quad (12)$$

Therefore,

$$\frac{E\alpha^2}{(1+\mu)^2} [A_m(1+\mu)\alpha + B_m(-1+\mu) - C_m(1+\mu)\alpha + D_m(-1+\mu)] = E_m \quad (13)$$

Using boundary condition $\sigma_{yy}(x, D) = \sigma_0$ at the edge of $y = D$

$$\begin{aligned} \frac{-E}{(1+\mu)^2} \left[\begin{aligned} &-(2+\mu) \sum_{m=1}^{\infty} \left\{ (A_m \alpha + B_m \alpha D + B_m) e^{\alpha D} + (-C_m \alpha - D_m \alpha D + D_m) e^{-\alpha D} \right\} \alpha^2 \cos \alpha x \\ &+ \sum_{m=1}^{\infty} \left\{ (A_m \alpha^3 + B_m \alpha^3 D + 3B_m \alpha^2) e^{\alpha D} + (-C_m \alpha^3 - D_m \alpha^3 D + 3D_m \alpha^2) e^{-\alpha D} \right\} \cos \alpha x + 6K \end{aligned} \right] = \sum_{m=1}^{\infty} I_m \cos \alpha x + I_o \quad \text{or,} \\ \frac{E\alpha^2}{(1+\mu)^2} \left[\begin{aligned} &A_m(1+\mu)\alpha e^{\alpha D} + B_m(\mu\alpha D + \alpha D + \mu - 1)e^{\alpha D} - \\ &C_m(1+\mu)\alpha e^{-\alpha D} - D_m(\mu\alpha D + \alpha D - \mu + 1)e^{-\alpha D} \end{aligned} \right] = I_m \quad (14) \end{aligned}$$

and using Eq. (9a) and Eq. (12) the arbitrary constant K can be obtained as follows:

$$\begin{aligned} \frac{-E}{(1+\mu)^2} 6K &= E_o = \frac{\sigma_0}{10} \\ \text{or, } K &= \frac{-\sigma_0(1+\mu)^2}{60E} \quad (15) \end{aligned}$$

The simultaneous equations (10), (11), (13) and (14) can be realized in a simplified matrix form for the solution of unknown terms like A_m , B_m , C_m and D_m as follows:

$$\begin{bmatrix} DD_1 & DD_2 & DD_3 & DD_4 \\ FF_1 & FF_2 & FF_3 & FF_4 \\ HH_1 & HH_2 & HH_3 & HH_4 \\ KK_1 & KK_2 & KK_3 & KK_4 \end{bmatrix} \begin{bmatrix} A_m \\ B_m \\ C_m \\ D_m \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ E_m \\ I_m \end{bmatrix} \quad (16)$$

$$\frac{-E\alpha^2}{(1+\mu)^2} [(1+\mu)\alpha A_m + 2\mu B_m + (1+\mu)\alpha C_m - 2\mu D_m] = 0 \quad (10)$$

Using boundary condition $\sigma_{xy}(x, D) = 0$ at the edge of $y = D$

$$\frac{-E\alpha^2}{(1+\mu)^2} \left[\begin{aligned} &A_m(1+\mu)\alpha e^{\alpha D} + B_m(\alpha D + \mu\alpha D + 2\mu) e^{\alpha D} + \\ &C_m(1+\mu)\alpha e^{-\alpha D} + D_m(\alpha D + \mu\alpha D - 2\mu) e^{-\alpha D} \end{aligned} \right] = 0 \quad (11)$$

Using boundary condition $\sigma_{yy}(x, 0) = \sigma_0 / 2$ at the edge of $y = 0$

where

$$DD_1 = Z_{11}(1+\mu)\alpha$$

$$DD_2 = 2\mu Z_{11}$$

$$DD_3 = Z_{11}(1+\mu)\alpha$$

$$DD_4 = -2\mu Z_{11}, \quad FF_1 = Z_{11}(1+\mu)\alpha e^{\alpha D}$$

$$FF_2 = Z_{11} \{ (1+\mu)\alpha D + 2\mu \} e^{\alpha D}$$

$$FF_3 = Z_{11}(1+\mu)\alpha e^{-\alpha D}$$

$$FF_4 = Z_{11} \{ (1+\mu)\alpha D - 2\mu \} e^{-\alpha D}$$

$$HH_1 = -Z_{11}(1+\mu)\alpha$$

$$HH_2 = -Z_{11}(-1+\mu)$$

$$HH_3 = Z_{11}(1 + \mu)\alpha$$

$$HH_4 = -Z_{11}(-1 + \mu)$$

$$KK_1 = -Z_{11}(1 + \mu)\alpha e^{\alpha D}$$

$$KK_2 = -Z_{11}\{(1 + \mu)\alpha D + \mu - 1\}e^{\alpha D}$$

$$KK_3 = Z_{11}(1 + \mu)\alpha e^{-\alpha D}$$

$$KK_4 = Z_{11}\{(1 + \mu)\alpha D - \mu + 1\}e^{-\alpha D}$$

$$Z_{11} = \frac{-E\alpha^2}{(1 + \mu)^2}$$

Therefore, stress and displacement components at various points of the beam can be obtained using equations of (7).

IV. RESULT ANALYSIS

The analytical solutions of displacement and stress components are obtained for various aspect ratios (L/D) of the beam. The material of the beam is mild steel whose modulus of elasticity is $E = 209 \times 10^9$ and poisson's ratio $\mu = 0.3$. The result of a guided isotropic beam having aspect ratio two and the uniform loading parameter $\sigma_0 = 40$ N/mm is presented in sequence of axial displacement (u_x), lateral displacement (u_y), bending stress (σ_{xx}), normal stress (σ_{yy}) and shearing stress (σ_{xy}).

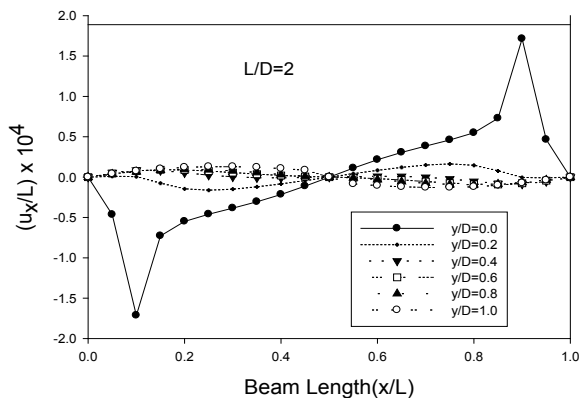


Fig. 3(a) : Axial Displacement along beam length

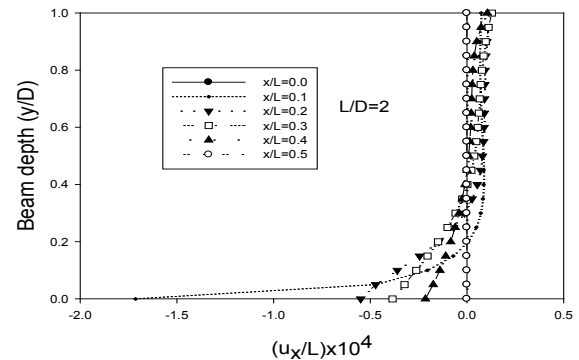


Fig. 3(b) : Axial Displacement along beam depth

Axial displacements (u_x) are found to be zero at the mid-section of span and at the lateral guided boundaries. Zero value of u_x at the guided ends verifies the boundary condition of those edges of the beam. Axial displacements distribution is found skew-symmetric about the mid-span of the beam. The values of u_x for sections $0 < x/L < 0.5$ are negative at the lower portion and positive at upper portion of the beam. The maximum magnitudes of $u_x/L = \pm 0.000158$ are observed on bottom fiber at the sections of $x/L = 0.1$ and $x/L = 0.9$ respectively.

Lateral displacements (u_y) are found to take positive value near the two guided lateral ends and negative in the region $0.25 < x/L < 0.75$ for $L/D=2$. The u_y results are in confirmation to the physical condition of the beam. The beam is being pushed up at the corners and forced down at the mid-span region. The normalized values of positive and negative maximum lateral displacements are $u_y/D = 0.000250$ and $u_y/D = -0.000426$ respectively for $L/D=2$. The maximum magnitude is observed on the topmost fiber at the mid-span.

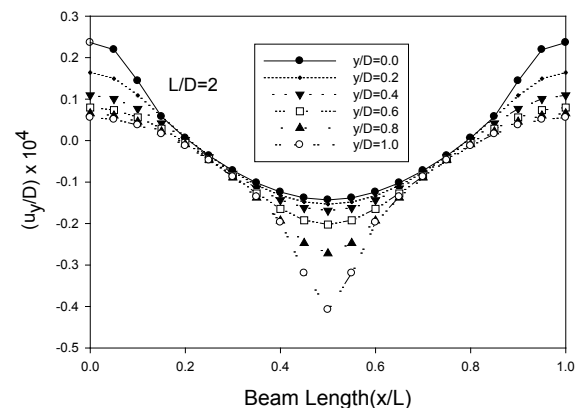


Fig. 4(a) : Lateral displacement along beam length

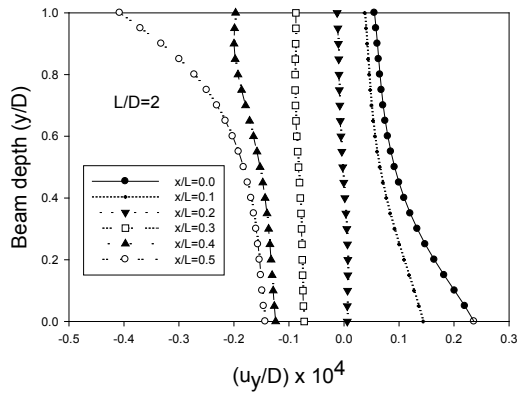


Fig. 4(b) : Lateral displacement along beam depth

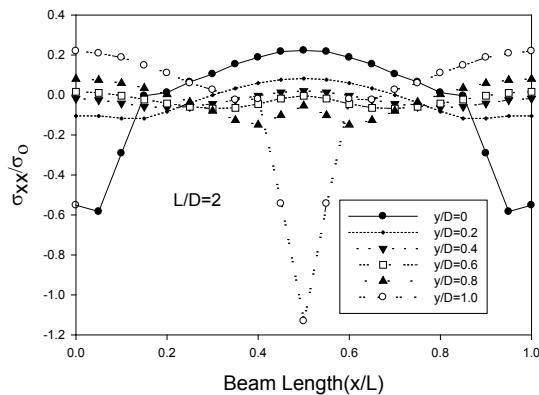


Fig. 5(a) : Bending stress along beam length

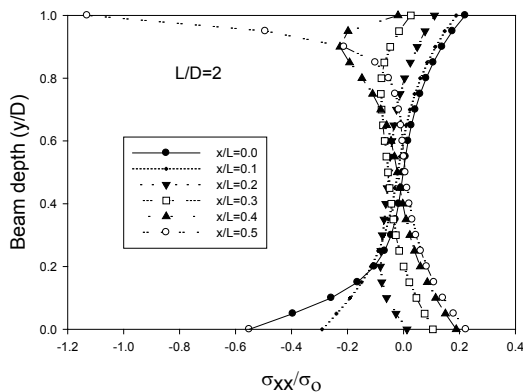


Fig. 5(b) : Bending stress along beam depth

Bending stress distribution is observed non-linear over the whole span. The stress (σ_x) maximizes at the mid-span at top fiber of the beam where the point type load is acting. The next locations of bending stress concentration are the two bottom corners and the bottom fiber at mid-span of the beam. The maximum magnitude of normalized bending stress on the top fiber at mid-span is a 1.19.

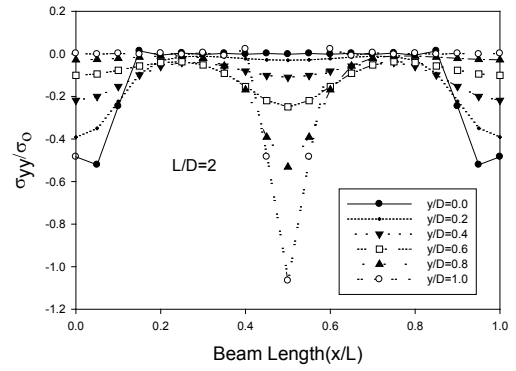


Fig. 6(a) : Lateral stress at beam length

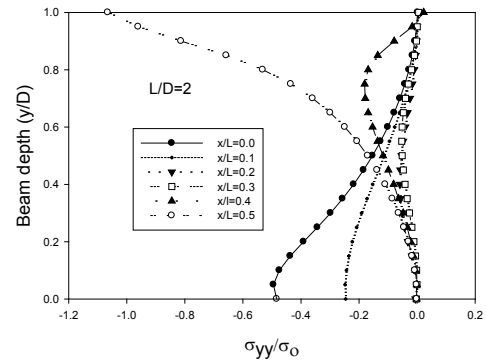


Fig. 6(b) : Lateral stress at beam depth

Lateral stress (σ_{yy}) concentrations are observed at the topmost fiber of mid-span section and bottom two corners. It is understandable that each reaction is half of the load along beam length and the normalized value of the lateral stress varies from zero to about unity at the topmost layer in the loaded region and it is almost half at the bottom layer of the support region along beam depth, which confirms the physical condition of the problem.

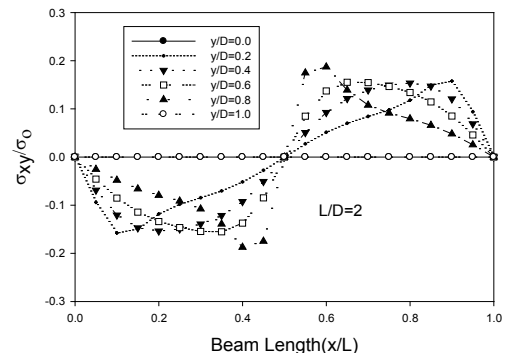


Fig. 7(a) : Shear stress along beam length

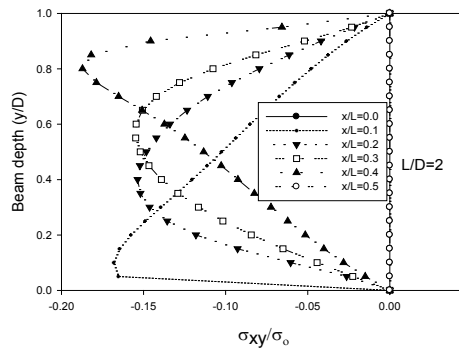


Fig. 7(b) : Shear stress along beam depth

All four edges and mid-span section of the guided simply supported beam are found free from

shearing stress. The distribution of shearing stress (σ_{xy}) for point loading is anti-symmetric in two sides about the mid-span of the beam. The maximum concentration of shearing stress is observed near the bottom corners at the supports and is at the top edge where the termination of loading takes place. The normalized maximum magnitude of shear stress is ± 0.2 for $L/D = 2$. Shear stress distribution at transverse section is nearly parabolic. Along beam depth maximum shear occurs at $x/L = 0.2$ and normalized value of this shear stress is -0.18 . Shear stress is zero at $x/L = 0.4$.

As the aspect ratio increases, the magnitude of both axial and lateral displacement increases. The sharp changes in the curve become gradually smoother for higher aspect ratios.

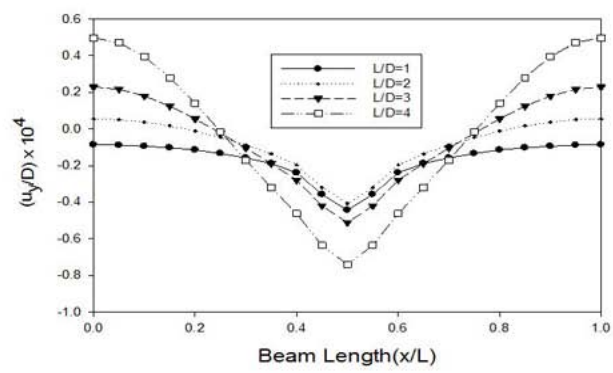
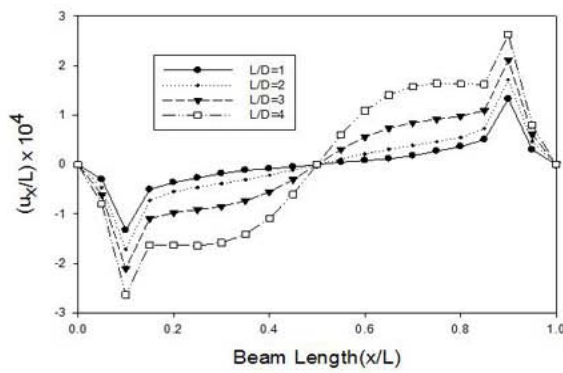


Fig. 8 : Comparison of axial and lateral displacement

V. VERIFICATION

Finite element analysis has also been carried out to verify the stress component of the beam. It can be observed that the stresses found out by the analytical solution, is in complete harmony to that of the finite element analysis. Hence the validity of the displacement potential formulation is justified.

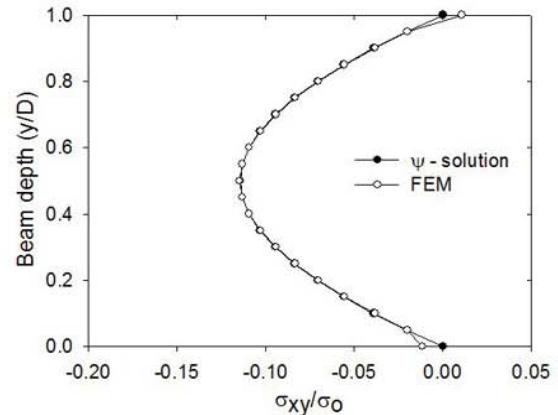
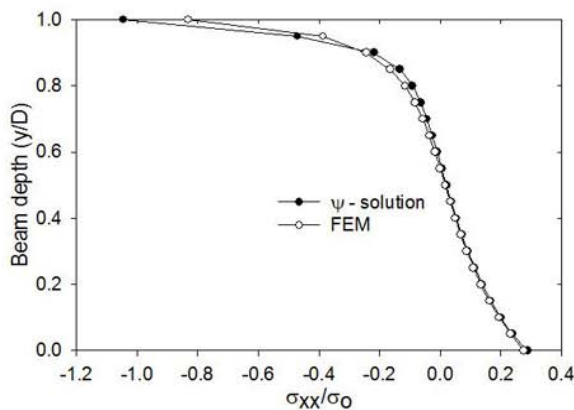


Fig. 9 : Verification of displacement potential formulation by finite element method

VI. CONCLUSION

Analytical solution using displacement potential approach for the elastic fields of a guided simply

supported beam of isotropic material under three point bending is explored satisfying all the physical conditions of the beam appropriately. The specialty of the guided ends is the mixed mode of boundary conditions.

Basically, the guided ends provide the freedom of lateral displacement but not the axial one. At this scenario the necessity of imposing boundary restraints is essential. But it is not practicable to use the classical Bernoulli-Euler beam theory for the solution of guided beam. Because, it cannot handle mixed mode of boundary conditions. Displacement potential formulation can handle mixed mode of boundary condition appropriately and we have found the solution of the beam.

It is observed from the solution of the beam that

- Axial displacement is maximum at bottom fiber i.e. $y/D=0.0$. As the aspect ratio increases axial displacement also increases.
- Lateral displacement is maximum at top fiber i.e. $y/D=1.0$. As the aspect ratio increases lateral displacement also increases. So if failure occurs it will occur at midsection of the beam i.e. where deflection is maximum.
- Maximum stress concentration occurs at midsection of the beam. So in terms of stress, midsection is more vulnerable to failure for a simply supported beam.

These findings will have applications in aircraft, spacecraft and vehicle structures for predicting appropriate stress distribution in them, thus allowing designers to design with greater safety.

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